NEW YORK UNIVERSITY

INSTITUTE . THE MATICAL SCIENCES

AFOSR TN-58-955
ASTIA DOCUMENT No. 2052348 averly Place, New York 3, N. Y.

IMM-NYU 250 OCTOBER 1958



NEW YORK UNIVERSITY
INSTITUTE OF
MATHEMATICAL SCIENCES

On Finite Deformations of an Elastic Isotropic Material

FRITZ JOHN

INSTITUTE COLLECTION OF CAL SCIENCES

IMM-250 John On definite formations of an elastic isotropic ...

PREPARED UNDER
CONTRACT NO. AF 49(638)-161
WITH THE
AIR FORCE OFFICE OF SCIENTIFIC RESEARCH
AIR RESEARCH AND DEVELOPMENT COMMAND



AFOSR TN-58-955
ASTIA Document No. 205 348

NEW YORK UNIVERSITY

Institute of Mathematical Sciences

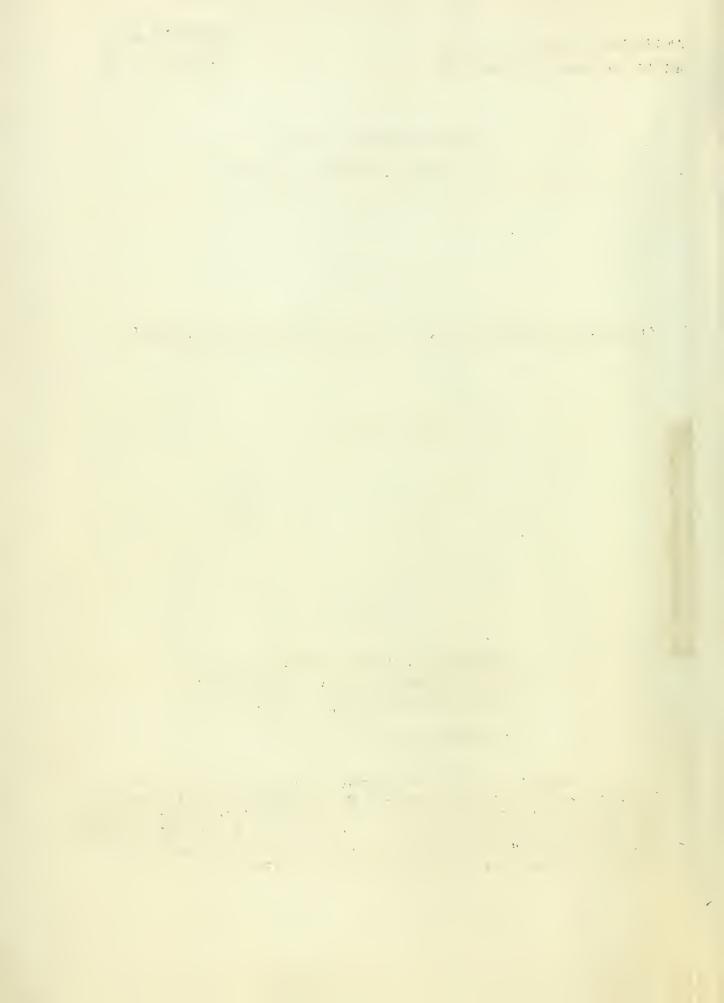
ON FINITE DEFORMATIONS OF AN ELASTIC ISOTROPIC MATERIAL

bу

Fritz John

Contract No. AF 49(638)-161
Mechanics Division, Air Force Office
of Scientific Research, ARDC,
Washington 25, D. C.

Qualified requestors may obtain copies of this report from the Armed Services Technical Information Agency, Arlington Hall Station, Arlington 12, Virginia. Department of Defense contractors must be established for ASTIA services, or have their "need-to-know" certified by the cognizant military agency of their project or contract.



## Introduction

This report contains some results on finite deformations of an ideal elastic material.

The first chapter deals with a purely geometric question, which is guite independent of the physical properties of materials. The question is how much a 1-1 transformation of a domain D can differ from a rigid motion, if the maximum relative extension

$$\frac{|ds' - ds|}{ds}$$

occurring is at most ε. The answer to this question gives information on the size of the strains that are bound to occur in a solid when the solid is subjected to a deformation that changes the distance of one pair of points by a certain amount. It is obvious that for a solid of diameter  $\bigwedge$  a deformation with maximum extension & cannot increase the distance of any two points by more than  $\varepsilon \wedge$ . It is not so obvious that mutual distances between two points could not decrease by larger amounts. What distance changes are compatible with extensions of maximum size & depends on the shape of the body. geometric analysis is carried through here only for the case of a square plate of side a and thickness h, where h < a. It is proved that there exists a constant M with the following property. For any transformation x' = f(x) of the plate (in vector notation) there exists a rigid motion  $x^{\dagger} = g(x)$  [g(x) is a linear transformation with constant coefficients] such that

(2) 
$$|f(x) - g(x)| \leq M \epsilon a^2/h .$$

The best value for M is not determined here. It depends doubtlessly on the range of  $\epsilon$ -values admitted. If, say, only values of  $\epsilon$  below 1/100 are considered, better estimates for M can be obtained, than if we permit for  $\epsilon$  all values between 0 and 1. For practical purposes it would seem worthwhile to obtain the best M for the limiting case  $\epsilon \longrightarrow 0$ .

egy for

randra de la companya de la company La companya de la co

1

Transfer to the second second

. 1

Formula (2) shows that a deformation of a plate differs from a rigid motion by a quantity of order  $\epsilon a$ , unless the thickness h of the plate is small in comparison to its width a. It is shown for thin plates the components of displacement tangential to the plate stay of order  $\epsilon a$  (at least for  $\epsilon << h^2/a^2$ ), while the normal components can be of order  $\epsilon a^2/h$ .

The proof of (2) is based on the realization that there exist subsets called "cores" of the domain D which is undergoing the transformation, in which the relative change in distance of any two points is at most  $\varepsilon$ . In fact, any convex subset of D whose distance from the boundary of D exceeds a certain amount  $\delta = \delta(\varepsilon, D)$  is such a core. Moreover,  $\delta$  tends to 0 for  $\varepsilon \longrightarrow 0$ . This implies that for convex D and small  $\varepsilon$  there are cores of D which fill all of D except for a thin boundary layer. Since in a core the mutual distance between any two points changes by a small amount for small  $\varepsilon$ , the transformation of a core is essentially rigid, which leads to the desired result. Similar estimates can doubtlessly be obtained for D which have shapes differing from square plates.

We have here measured the amount by which the transformation f differs from a rigid motion g by the maximum of f-g.
It is more customary to measure the difference by the difference
of the matrices of the first derivatives of f and g. Let p
denote the matrix formed from all first derivatives of the
components of f ("Jacobian matrix"). The matrix p can be
decomposed into an orthogonal matrix c giving the local rotation
and a symmetric matrix  $1+\eta$  determining the strain:

$$cp = 1 + \eta$$
.

For small elongations  $\eta$  is small. For  $\eta=0$  we have a rigid motion and c must be a constant orthogonal matrix. The question, then is how much c can differ from a constant matrix for small  $\eta$ .

It is shown by a counter example that small extensions do not guarantee that the local rotations at all points are approximately the same everywhere. However, this will be true in the mean. We can show that in the case of a square plate

. The state of the

, and the second of the second

 $g_{ij}^{*}(x_i) = g_{ij}^{*}(x_i)$ 

. 110

subjected to a transformation with maximum extension  $\epsilon$  there is a constant orthogonal matrix  $\gamma$  such that we have the estimate

$$\iiint (c - \gamma)(c - \gamma)^* dV = O(\epsilon a^{4}/h)$$

for the norm of  $c-\gamma$ . This means that on the average p differs little from a constant orthogonal matrix for small  $\epsilon$ . For h of the same order as a the norm is of order  $\epsilon a^3$ .

Chapter II gives an exposition of the classical equations for ideal elastic isotropic materials. The relations are equivalent to those given by F. D. Murnaghan [2], Novozhilov [1] and by Rivlin [6]. The emphasis is however on arriving at a formulation that permits formulation and solution of problems purely in terms of Lagrange coordinates. The strains and stresses which are the quantities usually in the foreground are replaced here by their more immediate Lagrangian analogues

$$p_{ik} = \frac{\partial x_i^!}{\partial x_k}$$
,  $q_{ik} = \frac{\partial U(p)}{\partial p_{ik}}$ 

where U is the strain energy function. The advantage of working with these quantities (not sufficiently stressed in the literature) is that the equilibrium equations and compatibility conditions are the simple linear relations

$$q_{ik,k} = 0$$
 ,  $p_{ik,r} = p_{ir,k}$  ,

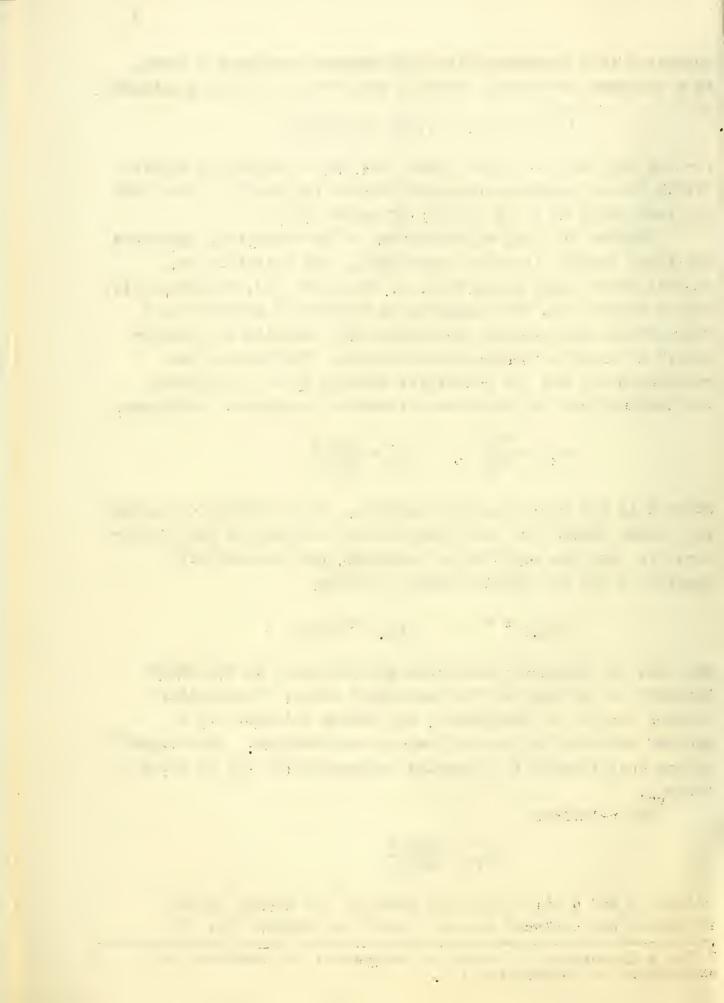
and that the boundary conditions are all taken on the known boundary of the body in the unstrained state. No curvature tensors have to be considered, and tensor calculus can be avoided entirely in favor of matrix computations. Non-linearity enters only through the formulae expressing the  $\mathbf{q}_{ik}$  in terms of the  $\mathbf{p}_{ns}$ .

The relations

$$q_{ik} = \frac{\partial U(p)}{\partial P_{ik}}$$

between p and q which take the place of the stress strain relations are analyzed in more detail in Chapter III. In

For a discussion of orders of magnitudes of rotations and extensions see Novozhilov [1].



particular, the expansion of q in terms of p near p = l is carried out up to terms of third order in p-l since information on these terms is needed for the later discussion of buckling of plates.

In Chapter IV we consider changes in potential energy resulting from changes in the transformation x' = f(x). The potential energy is stationary for the equilibrium state. It is shown that the equilibrium state actually corresponds to a minimum of potential energy if displacements are prescribed on the boundary, provided that for the deformations admitted the matrix p is restricted to a small finite neighborhood of the unit matrix. This yields a uniqueness theorem for the problem of determining the strained state of the body for given displacements on the boundary for an arbitrary choice of strain energy function.

Chapters V and VI derive the equations for bending of thin plates using Lagrange coordinates. In essence the method is similar to that used by Chien and Synge (see [3], [4]). The plate in the unstrained state is taken to be of constant thickness 2h with its faces given by  $x_3 = \pm h$ . It is assumed that the transformation corresponding to the strained state can be expanded into a formal power series in terms of  $x_3$  and h. We obtain from the equilibrium equations, compatibility conditions and boundary conditions on the faces recursion formulae for the coefficients of the power series which are functions of  $x_1$  and  $x_2$ . These recursion formulae essentially determine successively the coefficients and exclude a buckled solution unless on the limiting middle surface, i.e. for  $x_3 = h = 0$  the determinant

$$\left| \frac{\partial^2 \mathbf{U}(\mathbf{p})}{\partial \mathbf{p_{ik}} \partial \mathbf{p_{rs}}} \right|$$

vanishes. The determinant vanishes in particular when the stresses vanish to lowest order in h on the middle surface  $x_3 = 0$ . In that case the solution of the recursion formulae follows the pattern familiar from other asymptotic expansion

The second second

problems. At each stage of the expansion the recursion formulae determine the expansion coefficients only within an arbitrary function; differential equations for that arbitrary function are only found by considering the recursion formulae for the next higher orders and expressing that those are compatible with existence of a solution. In particular, the complete system of differential equations for the first non-vanishing quantities constitutes then what may be considered as the differential equations for thin plates.

In Chapter V this program is carried out for the case of two dimensional transformations, admitting large local rotations for h -> 0. The resulting differential equation for the inclination 9 of the limiting middle surface is similar to the equation of the elastica. In Chapter VI three dimensional transformations are considered, in which however the middle surface shall in the limit for h -> 0 coincide with the original unstrained surface  $x_3 = 0$  (instead of coinciding with a more general developable surface). The further assumption that the tangential stresses are of order h2 leads to the equations of v. Karman and Foppl. It turns out that the differential equations for the lowest order non-vanishing quantities only involve two material constants, the Lamé constants  $\lambda$  and  $\mu$ . In the derivation of the equations, higher order terms in the expansion of q in powers of p-1 have to be considered. These terms involve in addition to  $\lambda$  and  $\mu$  seven further constants, which however for reasons of symmetry do not enter the final result.

, A , .F \*1... , The state of the s . (), . . .  $\epsilon = \epsilon_{a,A_{a}}$ e deve . , , y,n  Geometric Analysis of Deformations with Small Strain

Let a transformation be described by the equations

(1.1) 
$$x_i' = f_i(x_1, x_2, x_3)$$
,  $i = 1, 2, 3$ 

or in vector notation by

$$(1.2)$$
  $x' = f(x)$ .

We associate with the transformation the matrix

(1.3) 
$$p = (p_{ik}) = \left(\frac{\partial x_i'}{\partial x_k}\right)$$

and assume that p has a positive determinant |p|. The corresponding local linear approximation to the transformation is then given by

$$(1.4) dx' = pdx .$$

The elongation of a line element is represented by the expression

(1.5) 
$$e = \frac{ds' - ds}{ds} = \sqrt{\frac{dx*p*pdx}{dx*dx}} - 1$$

where the asterisk denotes transposition.

The elongation in different directions is completely determined by the positive symmetric matrix

(1.6) 
$$g = p^*p$$
.

For dx varying in direction e is stationary when dx has the direction of an eigenvector of g. The stationary values of e are given by

$$(1.7) \sqrt{\lambda} - 1$$

where  $\lambda$  is an eigenvalue of g. Following K. O. Friedrichs we can introduce as strain matrix the matrix  $\eta$  determined by the conditions that  $1+\eta$  is positive symmetric and that

.5

for the second control of the second control · . . Development of the second of t 

. .

- .

to the second se 

1 1 

(1.8) 
$$(1+\eta)^2 = g = p^*p .$$

The principal elongations e are just the eigenvalues of  $\eta$  . If we introduce the matrix c by

(1.9) 
$$cp = 1 + \eta$$

we have

$$p^*p = (1+\eta)^2 = (1+\eta^*)(1+\eta) = p^*c^*ep$$

and hence

The matrix c will be orthogonal and of determinant +1, since p and l+p have positive determinant. Equation (1.9) corresponds to the decomposition of the infinitesimal transformation into a rotation c\* and a pure dilatation represented by l+p:

$$p = c^*(1 + \eta)$$
.

If the transformation is isometric we have

$$g = 1$$
,  $e = 0$ ,  $\eta = 0$ ,  $p = c^*$ .

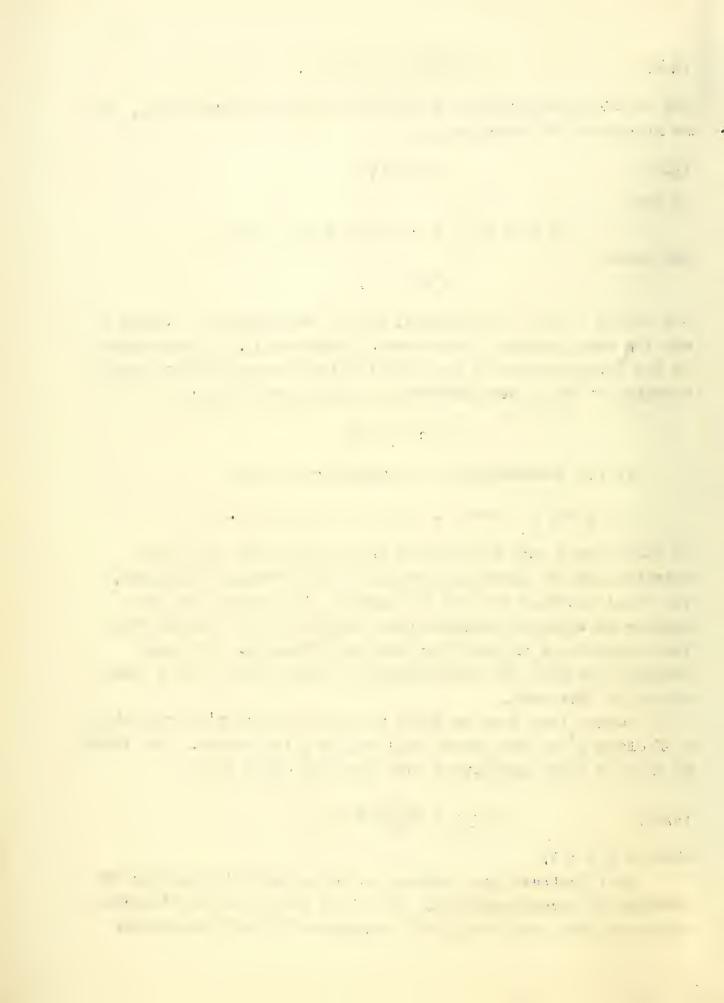
In that case f must represent a rigid motion and the local rotation must be given by a matrix c\* with constant elements. The local rotation is then the same at all points. In this chapter we shall investigate what happens if one assumes that the elongation e is small for all line elements. We shall estimate how much the transformation f can differ from a rigid motion in this case.

Assume then that we apply a transformation x' = f(x) with f of class  $C^1$  to the points of a region R in x-space. Let there be given a fixed quantity  $\varepsilon$  such that for all x in R

$$(1.10) -\varepsilon \le e = \frac{ds! - ds}{ds} \le +\varepsilon$$

where  $0 \le \epsilon < 1$ .

To illustrate the behavior to be expected we consider two examples of transformations. The first example to be discussed will show that even though the elongation is small everywhere



the local rotation does not have to be approximately the same in all points. We make use of cylindrical coordinates r, 0, z:

$$x_1 = r \cos \theta$$
,  $x_2 = r \sin \theta$ ,  $x_3 = z$ 

and consider a transformation of the form

(1.11) 
$$r' = r$$
,  $\theta' = \theta - F(r)$ ,  $z' = z$ 

with a suitable function F(r). The elements of length are given by

$$ds^{2} = dr^{2} + r^{2}d\theta^{2} + dz^{2}$$

$$ds^{2} = ds^{2} + r^{2}(d\theta^{2} - d\theta^{2})$$

Here

$$|r^{2}(d\theta^{2} - d\theta^{2})| = |rF'(-2rd\theta dr + rF'dr^{2})|$$

$$< |rF'|(1 + |rF'|)ds^{2} < \epsilon ds^{2}$$

if

$$|\mathbf{rF}'(\mathbf{r})| \leq \frac{\varepsilon}{2} .$$

It follows that the elongation e satisfies (1.10), if (1.12) holds.

The matrix p is here given by

$$p = \begin{pmatrix} \cos F + rF' \sin \theta' \cos \theta & \sin F + rF' \sin \theta' \sin \theta & 0 \\ -\sin F - rF' \cos \theta' \cos \theta & \cos F - rF' \cos \theta' \sin \theta & 0 \\ 0 & 0 & 1 \end{pmatrix}$$

Taking for c the orthogonal matrix

$$c = \begin{pmatrix} \cos \emptyset & -\sin \emptyset & 0 \\ +\sin \emptyset & \cos \emptyset & 0 \\ 0 & 0 & 1 \end{pmatrix}$$

the matrix cp will be symmetric, when

lugar 1 -:•  $f = \{x \in \mathcal{X} \mid x \in \mathcal{X}\}$ 250 100 . (\* 7.5) 

$$\emptyset = F(r) + arc tan (\frac{1}{2} rF'(r))$$
.

If we now choose for F(r) the function

$$F(r) = \varepsilon \log (\log \frac{1}{r}) ,$$

condition (1.12) will be satisfied for  $r < e^{-2}$ , while

$$\lim_{r \to 0} \emptyset = \infty ,$$

so that the rotation angle  $\emptyset$  takes every value in a neighborhood of the origin.

This shows that arbitrarily large variations in local rotation are compatible with small strains.

The next example indicates the magnitudes of the displacements in various directions to be expected in a transformation of a rectangular plate when the maximum elongation is of order  $\epsilon$ . Let the plate be given by

(1.13) 
$$0 \le x_1 \le a$$
,  $0 \le x_2 \le a$ ,  $0 \le x_3 \le h \le a$ .

We apply the transformation

$$x_1' = (R-x_3)\sin(x_1/R)$$
,  $x_2' = x_2$ ,  $x_3' = R - (R-x_3)\cos(x_1/R)$ 

which takes the plate into a portion of a cylindrical shell.

Here the parameter R shall have a value exceeding h and a. The strain matrix has the simple form

$$\eta = \begin{pmatrix}
-x_3/R & 0 & 0 \\
0 & 1 & 0 \\
0 & 0 & 1
\end{pmatrix}$$

in this case, so that the maximum elongation is  $\varepsilon = h/R$ . The maximum horizontal displacement occurs for  $x_1 = a$ ,  $x_3 = h$  and has the value

(1.12a) 
$$\max |x_1' - x_1| = a - (R-h)\sin(a/R) = a\varepsilon(1+0(a^2\varepsilon/h^2))$$
.

The maximum vertical displacement occurs for  $x_1 = a$ ,  $x_3 = 0$  and has the value

1-1-1-1-1 . ,  $1 - \psi f$ . . . n 3 <u>,</u> ≥.

(1.12b) 
$$\max |x_3^1 - x_3| = 2R \sin^2(a/2R) = 0(a^2 \varepsilon/h)$$
.

Since here by assumption  $a\varepsilon/h = a/R < 1$  the maximum total displacement of any point is of order  $a^2\varepsilon/h$ . In particular, if h is of the same order as a the maximum displacement is of order  $\varepsilon a$ . If h is small compared to a but  $a^2\varepsilon/h^2$  is still bounded the horizontal displacement is still of order  $\varepsilon a$ , whereas the vertical displacement goes up to the order  $(a/h)\varepsilon a$ . This latter situation arises when the total displacement of any point does not exceed the thickness h of the plate.

We find then in this example that for a rectangular plate with sides of the same order a and maximum elongation  $\epsilon$  the displacement is of order  $\epsilon a$ . The horizontal displacement is still of the same order as long as  $\sqrt{\epsilon}$  does not exceed the ratio of thickness h to width a. In any case the total displacement is of order  $\epsilon a^2/h$ , as long as  $\epsilon$  does not exceed h/a. We shall find essentially the same result for general transformations of the plate with maximum elongation  $\epsilon$  (after applying a suitable rigid motion to the transformed region).

Let then

$$x' = f(x)$$

be a transformation of a region R with boundary B, where

$$1 - \varepsilon \le \frac{\mathrm{d}s'}{\mathrm{d}s} \le 1 + \varepsilon$$

for  $x \in \mathbb{R}$ . If C is any curve in R and C' its image under f, we have the inequalities

$$(1-\epsilon)$$
  $\int_C ds \le \int_{C_1} ds' \le (1+\epsilon) \int_C ds$ 

for the lengths of the curves C and C!.

Let P, Q be any two points of R such that the line segment C with endpoints P, Q belongs to R. Let  $\overline{PQ}$  denote the distance of P and Q. Then

$$\overline{P'Q'} \leq \int_{C'} ds' \leq (1+\epsilon) \int_{C} ds = (1+\epsilon) \overline{PQ}$$
.

garden i de la companya de la compa

te de la companya de

A CASE

The second of th

If in addition the line segment  $\gamma'$  with endpoints P', Q' belongs to the image region R', then  $\gamma'$  is the image of an arc  $\gamma$  joining P and Q, and

$$\overline{P'Q'} = \int_{\gamma'} ds' \ge (1 - \epsilon) \int_{\gamma} ds \ge (1 - \epsilon) \overline{PQ} .$$

Hence

$$(14.3) \qquad (1-\epsilon)\overline{PQ} \leq \overline{P'Q'} \leq (1+\epsilon)\overline{PQ} ,$$

if P and Q are such that the segment PQ belongs to R, the segment P'Q' to R'.

Consider now the case of a line segment PQ lying in R which is such that the segment P'Q' does not lie in R! Let S be a point on PQ such that P'S' belongs to R'. The set of these points is closed; it also contains P but not Q. There is then a point So in this set furthest away from P. The segment P'S' belongs then to R' but cannot lie completely in the interior of R'. Thus P'S' contains a point T' of the boundary B' of R'. T' is the image of a point T on the boundary B of R. Let \( \gamma\) denote the original in R of the segment \( \gamma'\) joining P' and S'. Since \( \gamma\) passes through T we have (see Figure 1)

$$(1-\epsilon)(\overline{PT} + \overline{TS}_0) \le (1-\epsilon) \int_{\Upsilon} ds \le \int_{\Upsilon'} ds' = \overline{P'S_0'} \le (1+\epsilon)\overline{PS}_0$$
.

Then also

$$(1-\varepsilon)(\overline{PT} + \overline{TQ}) \leq (1-\varepsilon)(\overline{PT} + \overline{TS}_{o} + \overline{S}_{o}Q) \leq (1+\varepsilon)\overline{PS}_{o} + (1-\varepsilon)\overline{S}_{o}Q$$

$$\leq (1+\varepsilon)(\overline{PS}_{o} + \overline{S}_{o}Q) = (1+\varepsilon)\overline{PQ} .$$

Hence there must exist a point T on the boundary B such that

$$\overline{PT} + \overline{TQ} \le \frac{1+\varepsilon}{1-\varepsilon} \overline{PQ}$$
.

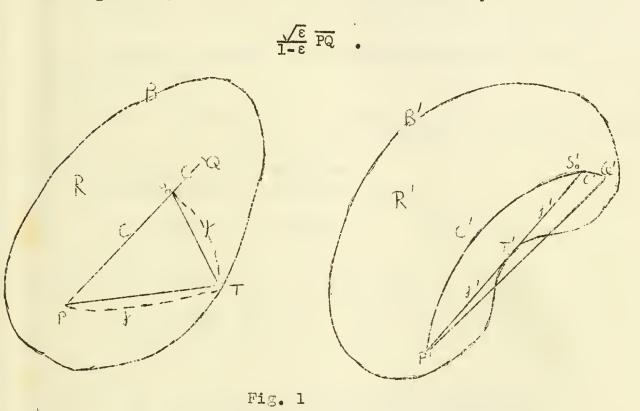
The points T satisfying this inequality fill a certain ellipse with foci P, Q and minor semi-axis

$$\frac{\sqrt{\varepsilon}}{1-\varepsilon} \overline{PQ}$$
.

. All of the second of the sec

• •

This is also the largest distance any point P of the ellipse can have from the segment  $\overline{PQ}$ . It follows that for a segment PQ in R the inequality (11) holds, if every point of the segment PQ has a distance from the boundary B which exceeds

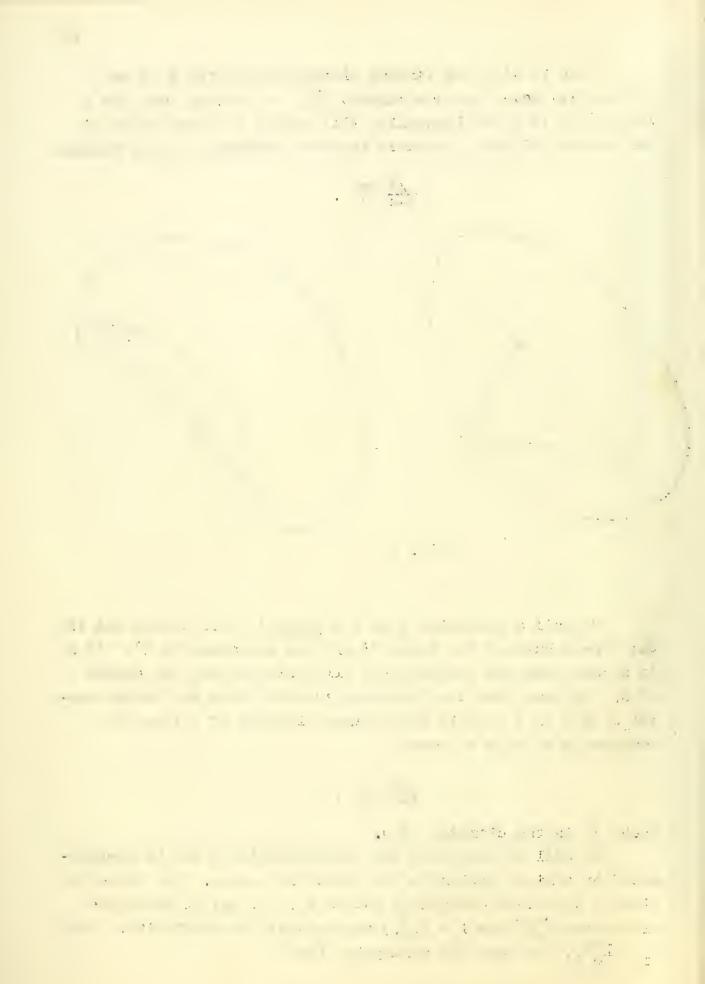


We call a subregion D of R a core, if D is convex and if the convex hull of the image D' of D is contained in R'. If D is a core then the inequalities (11) hold for any two points of D. We have from the preceding argument that any convex subset D of R is a core if the minimum distance of D from the boundary B of R is at least

$$\frac{\sqrt{\varepsilon}}{1-\varepsilon}$$
  $\triangle$  ,

where A is the diameter of D.

It will be seen that the transformation f can be approximated by a rigid motion in the points of a core. Let there be given a core that contains 4 points  $A_0$ ,  $A_1$ ,  $A_2$ ,  $A_3$  where the directions  $\overline{A_0}$  for i=1,2,3 are mutually perpendicular. Let  $a_i = \overline{A_0}$  We make the assumption that



$$(1.14)$$
  $a_1 = a_2 \ge a_3 > 0$ .

In a suitable rectangular coordinate system the A<sub>i</sub> will have coordinates

$$A_0 = (0,0,0), A_1 = (a_1,0,0), A_2 = (0,a_2,0), A_3 = (0,0,a_3).$$

By a suitable orthogonal transformation applied to R' we can bring about that

$$A_0^! = (0,0,0), \quad A_1^! = (a_{11},0,0), \quad A_2^! = (a_{21},a_{22},0), \quad A_3^! = (a_{31},a_{32},a_{33})$$

where all, a22, a33 are positive.

For any point P of the core we put

$$r_i = \overline{A_iP}$$
,  $r_i' = \overline{A_iP'}$  for  $i = 0,1,2,3$ .

Then for 
$$P = (x_1, x_2, x_3), P' = (x_1', x_2', x_3'), i = 1,2,3$$

$$2a_{i}x_{i} = r_{0}^{2} - r_{i}^{2} + a_{i}^{2}$$
,  $2a_{i}x_{i}' = r_{0}^{2} - r_{i}^{2} + a_{i}^{2}$ 

and hence

(1.15) 
$$x_{i}^{!} - x_{i} = \frac{(r_{0}^{!} - r_{0})(r_{0}^{!} + r_{0}) - (r_{i}^{!} - r_{i})(r_{i}^{!} + r_{i})}{2\alpha_{i}}$$

Here for i = 0, 1, 2, 3

(1.16) 
$$r_{\mathbf{i}}^{!} = \overline{A_{\mathbf{i}}P^{!}} = \overline{A_{\mathbf{i}}^{!}P^{!}} + O(\overline{A_{\mathbf{i}}A_{\mathbf{i}}^{!}}) = \overline{A_{\mathbf{i}}P} + O(\varepsilon \overline{A_{\mathbf{i}}P}) + O(\overline{A_{\mathbf{i}}A_{\mathbf{i}}^{!}})$$
$$= r_{\mathbf{i}} + O(\varepsilon r_{\mathbf{i}}) + O(\overline{A_{\mathbf{i}}A_{\mathbf{i}}^{!}}) .$$

In particular for i = 0

(1.17) 
$$r_{0}^{\dagger} = r_{0} + O(\epsilon r_{0}) .$$

If we take here for P the point A relation (1.16) becomes

$$a_{11} = \alpha_1 + O(\epsilon \alpha_1)$$

and hence

$$\overline{A_1 A_1'} = |a_{11} - a_1| = O(\epsilon a_1)$$
.

(2)

.

.

4

, 1

We restrict P to points of the core with  $r_0 \le 4a_1$ . Then also

$$r_i = \overline{A_i P} \le \overline{A_i A_o} + \overline{A_o P} = \alpha_i + r_o \le 5\alpha_1$$
.

We have then from (1.16) for i = 1

$$r_1' = r_1 + O(\epsilon r_1) + O(\epsilon \alpha_1) = r_1 + O(\epsilon \alpha_1) = O(\alpha_1)$$

It follows from (1.15) that

(1.18) 
$$x_1' - x_1 = 0(\varepsilon a_1)$$
.

In particular for  $P = A_2, A_3$ 

(1.19) 
$$a_{21} = O(\epsilon a_1)$$
,  $a_{31} = O(\epsilon a_1)$ .

Taking  $P = A_2$  we have from (1.17), (1.19)

$$a_{22} = \sqrt{a_{22}^2 + a_{21}^2} + o(a_{21}) = r_0' + o(\epsilon a_1)$$
  
=  $r_0 + o(\epsilon a_1) = a_2 + o(\epsilon a_1)$ 

and hence

$$A_2A_2^{\dagger} = \sqrt{a_{21}^2 + (a_{22} - a_2)^2} = O(\epsilon a_1)$$
.

Then generally for points P of the core with  $r_0 \le 4a_1$  by (1.16) and (1.15) (using  $a_2 = a_1$ )

$$r_2^{\dagger} = r_2 + O(\epsilon \alpha_1) = O(\alpha_1)$$

(1.20) 
$$x_2' - x_2 = 0(\epsilon \alpha_1)$$
.

In particular for  $P = A_3$  relation (1.20) becomes

$$a_{32} = O(\epsilon a_1)$$
,

and hence also

$$a_{33} = \sqrt{a_{33}^2 + a_{31}^2 + a_{32}^2} + 0\left(\sqrt{a_{31}^2 + a_{32}^2}\right) = r_0' + 0(\epsilon a_1)$$

$$= r_0 + 0(\epsilon a_1) = a_3 + 0(\epsilon a_1)$$

. The state of the

· · · )

the second second second second second

It follows that

$$\overline{A_3} \overline{A_3} = \sqrt{a_{31}^2 + a_{32}^2 + (a_{33} - a_3)^2} = O(\varepsilon a_1)$$

Then from (1.15), (1.16) for any P of the core with  $r_0 < 4a_1$ 

$$\mathbf{r}_3^{\dagger} = \mathbf{r}_3 + 0(\epsilon \alpha_1) = 0(\alpha_1)$$

(1.21) 
$$x_3' - x_3 = O(\epsilon \alpha_1^2/\alpha_3)$$
.

Take now the case where R is the rectangular plate

(1.21a) 
$$0 \le x_1 \le a$$
,  $0 \le x_2 \le a$ ,  $0 \le x_3 \le h$ 

with  $0 < h \le a$ . For given positive  $\delta$  the cell  $R_{\delta}$ 

(1.22) 
$$\delta \le x_1 \le a-\delta$$
,  $\delta \le x_2 \le a-\delta$ ,  $\delta \le x_3 \le h-\delta$ 

will have distance  $\delta$  from the boundary B of R, and its diameter will be at most  $\sqrt{3}$  a. It will be a core, if

$$\sqrt{3\varepsilon}$$
 a  $\leq \delta$ .

Assume that  $\varepsilon$  is so small that

(1.23) 
$$\epsilon \frac{a^2}{h^2} \le \frac{1}{48} (1 - \epsilon)^2$$

and take

$$\delta = \frac{\sqrt{3\varepsilon}}{1-\varepsilon} a .$$

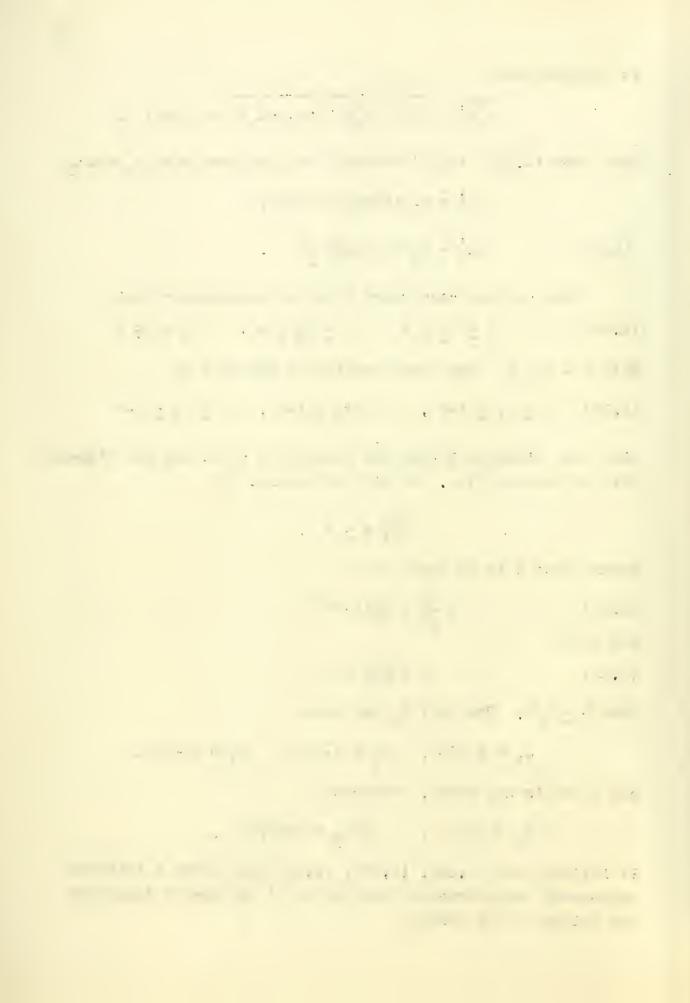
Then  $\delta \leq h/4$ . The cell  $R_{\delta}$  has sides

$$\alpha_1 = a - 2\delta$$
,  $\alpha_2 = a - 2\delta$ ,  $\alpha_3 = h - 2\delta$ ,

and hence is not empty. Moreover

$$a_1 = 0(a)$$
 ,  $a_1^2/a_3 = 0(a^2/h)$  .

It follows from (1.18), (1.20), (1.21) that after a suitable orthogonal transformation applied to  $R^{\dagger}$  we have at least for the points of the core  $R_{\delta}$ 



(1.25) 
$$x_1' - x_1 = 0(\epsilon a)$$
,  $x_2' - x_2 = 0(\epsilon a)$ ,  $x_3' - x_3 = 0(\epsilon a^2/h)$ .

These are exactly the orders of magnitude for horizontal and vertical displacements that occurred in the special example (1.12 a,b) when  $\epsilon a^2/h$  was bounded. They are proved here only for the core, which however in the limit for  $\epsilon \rightarrow 0$  will fill the whole region R.

A different kind of argument is needed to estimate the displacements outside the core  $R_{\delta}$  and also to treat the case where h is so small that (1.23) no longer holds. For this purpose we consider first the case of a core which contains a cube of side a and vertices  $A_{\delta}, A_{1}, \dots$ . If P is a point of the core with  $\overline{A_{\delta}P} = a$  we have by (1.18), (1.20), (1.21) after a suitable rigid motion applied to  $R^{\delta}$ 

$$\overline{PP^!} = O(\epsilon \alpha)$$
.

If 9 denotes the angle PA P we have

$$\neq PA_{O}P^{\dagger} = \Theta = O(PP^{\dagger}/A_{O}P) = O(\varepsilon)$$
.

If Q is the point of distance a from A on the ray from A opposite to P and if Q also belongs to the core, then

$$\neq QA_QQ^{\dagger} = O(\epsilon)$$

and hence

$$\not\equiv P^{\dagger}A_{0}Q^{\dagger} = \not\equiv P^{\dagger}A_{0}^{\dagger}Q^{\dagger} = \pi + O(\epsilon)$$
.

Thus if PQ is a line segment of length 2 a and A its midpoint we have

$$\not = P^{\dagger}A_{O}^{\dagger}A^{\dagger} = \pi + O(\epsilon) ,$$

provided there exists a core that contains P, Q and also contains a cube of side a and vertex A. This is true regardless of the special rigid motion applied to R which would not change the angle.

Let  $P_0, P_1, \dots, P_n$  be equidistant points on a line segment  $L = P_0 P_n$  of R, where

$$\overline{P_1P_{1+1}} = \alpha .$$

. Tay in the second of the sec

.

 Let there exist for each  $i=1,2,\ldots,n-1$  a core containing  $P_{i-1}$  and  $P_{i+1}$  and also containing a cube of side  $\alpha$  and vertex  $P_{i}$ . Let similarly  $P_{o}$ ,  $P_{1}$  belong to a cubical core of side  $\alpha$  and vertex  $P_{o}$ . The angle between two successive vectors  $P_{i}^{!}P_{i+1}^{!}$  is then  $O(\epsilon)$ . Hence the angle between any vector  $P_{i}^{!}P_{i+1}^{!}$  and  $P_{o}^{!}P_{1}^{!}$  is  $O(n\epsilon)$ . We assumed that there was a cubical core of side  $\alpha$  and vertex  $P_{o}$ . Let  $A_{1}$ ,  $A_{2}$ ,  $A_{3}$  be the vertices of the core adjacent to  $P_{o}$ . We take  $P_{o}$  as origin and the adjacent edges of the cube as coordinate axes. By a suitable rigid motion we bring about that  $P_{o}^{!}=P_{o}$ ,  $A_{1}^{!}$  lies on  $P_{o}A_{1}$ ,  $A_{2}^{!}$  lies in the plane  $P_{o}A_{1}A_{2}$ . Then the angle between  $P_{o}^{!}P_{1}^{!}$  and  $P_{o}^{!}P_{1}^{!}$ , that is the angle between  $P_{o}^{!}P_{1}^{!}$  and  $P_{o}^{!}P_{1}^{!}$  and  $P_{o}^{!}P_{1}^{!}$  and  $P_{o}^{!}P_{1}^{!}$  is  $P_{o}^{!}P_{1}^{!}$  and  $P_{o}^{!}P_{1}^{!}$  and  $P_{o}^{!}P_{1}^{!}$  is  $P_{o}^{!}P_{1}^{!}$ . Since also

$$\overline{P_{i}^{!}P_{i+1}^{!}} = \overline{P_{i}P_{i+1}} + O(\varepsilon\alpha) ,$$

we have for  $P_i = (x_{i1}, x_{i2}, x_{i3}), P_i' = (x_{i1}', x_{i2}', x_{i3}')$ 

$$x_{i+1 k}^{!} - x_{ik}^{!} = x_{i+1 k}^{!} - x_{ik}^{!} + 0(n\epsilon\alpha)$$
 for  $k = 1,2,3$ .

Hence

$$x_{nk}' = \sum_{i=0}^{n-1} (x_{i+1k}' - x_{ik}') = \sum_{i=0}^{n-1} (x_{i+1k} - x_{ik}) + 0(n^2 \epsilon \alpha)$$

$$= x_{nk}' + 0(n^2 \epsilon \alpha)$$

Consequently

$$\frac{\overline{P_n^{\dagger}P_n}}{P_n^{\dagger}P_n} = O(n^2 \epsilon \alpha) \quad .$$

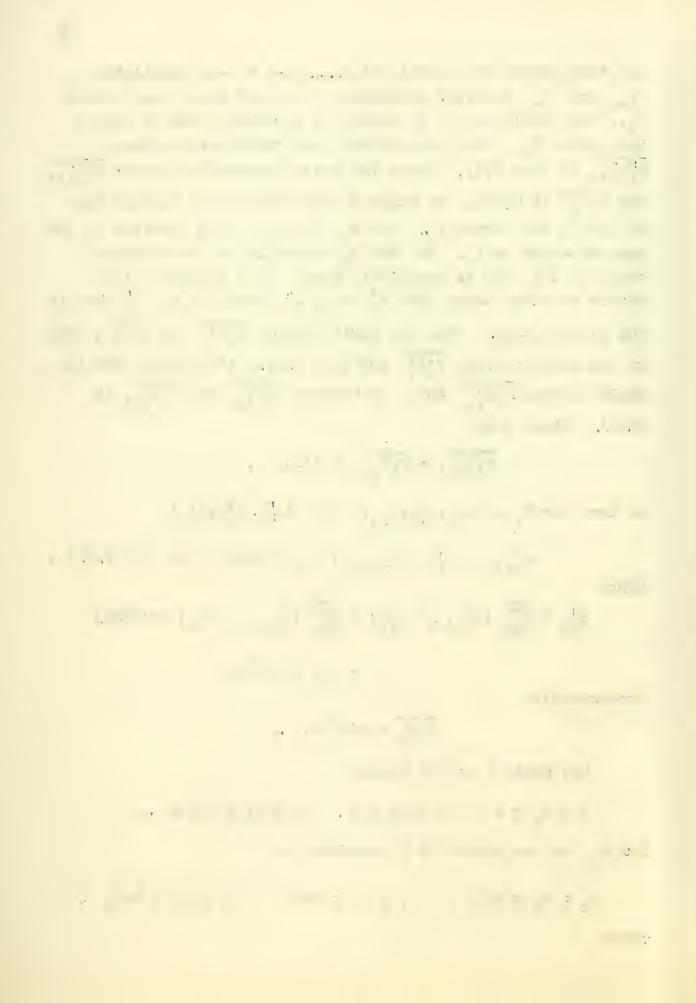
Let again R be the region

$$0 \le x_1 \le a$$
,  $0 \le x_2 \le a$ ,  $0 \le x_3 \le h \le a$ .

Let Roll be the subset of R described by

$$\delta_1 \le x_1 \le a - \delta_1$$
,  $\delta_1 \le x_2 \le a - \delta_1$ ,  $\delta_1 \le x_3 \le h - \delta_1$ ,

where



$$\delta_1 = \sqrt{\epsilon} h$$
.

Then  $\delta_1 \leq h/h$  if we assume that  $\epsilon < 1/16$ . Take for  $P_0$  the point  $(\delta_1, \delta_1, \delta_1)$  and for Q any point of  $R_{\delta_1}$ . We divide the line segment  $L = P_0Q$  into n equal parts  $\alpha$  by points  $P_1, P_2, \dots, P_{n-1}$  with  $P_n = Q$ . We take here for n some integer satisfying

$$20 \frac{a}{h} \le n \le 20 \frac{a}{h} + 1 \quad .$$

Then

$$\alpha = \frac{1}{n} \overline{P_0 Q} \le \frac{1}{n} \sqrt{3} a \le \frac{\sqrt{3}}{20} h \le \frac{1}{2} (h - 2\delta_1)$$
.

Every point  $P_i$  of L is the vertex of a cube in  $R_{\tilde{O}_l}$  with sides of length a which are parallel to the coordinate axes. The convex hull of this cube and of  $P_{i-l}$  and  $P_{i+l}$  is a core. For its diameter  $\triangle$  is at most

$$(1+\sqrt{3})\alpha \le \frac{\sqrt{3}}{20} (1+\sqrt{3})h$$

and its distance from B at least  $\delta_1$ , so that

$$\frac{\sqrt{\varepsilon}}{1-\varepsilon} \triangle \leq \delta_1 .$$

Consequently (after the proper orthogonal transformation applied to R<sup>†</sup>)

$$\overline{QQ^{\dagger}} = O(n^2 \epsilon \alpha) = O(n \epsilon \overline{PQ}) = O(n \epsilon \alpha) = O(\epsilon \alpha^2/h)$$

for any point Q of Rog.

It remains to extend the inequality

$$QQ' = O(\epsilon a^2/h)$$

to all points of R.

If Q is any point of R there exists a point P in  $R_{\delta}$  such that

$$\overline{PQ} < \delta$$
.

Then

$$Q'Q \leq Q'P' + P'P + \overline{PQ} \leq (1+\epsilon)\overline{PQ} + P'P + \overline{PQ} \leq (2+\epsilon)\delta + \overline{P'P}$$
.

. ~ = !-

- .

.

Company of the Compan

Y -- 1

. . . .

If here the inequality

$$P'P = O(\epsilon a^2/h)$$

has been established already for the points P of  $R_{\delta}$ , and if  $\delta = O(\epsilon a^2/h)$  it follows that

$$\overline{Q^!Q} = O(\epsilon a^2/h)$$

for the arbitrary point Q of R.

Taking here

$$\delta = \delta_1 = \sqrt{\epsilon} h$$

we find that (1.26) holds generally in R, if

$$\sqrt{\epsilon} h = O(\epsilon a^2/h)$$
.

This is the case when

$$\sqrt{\varepsilon} \frac{a^2}{h^2} \ge 1$$
.

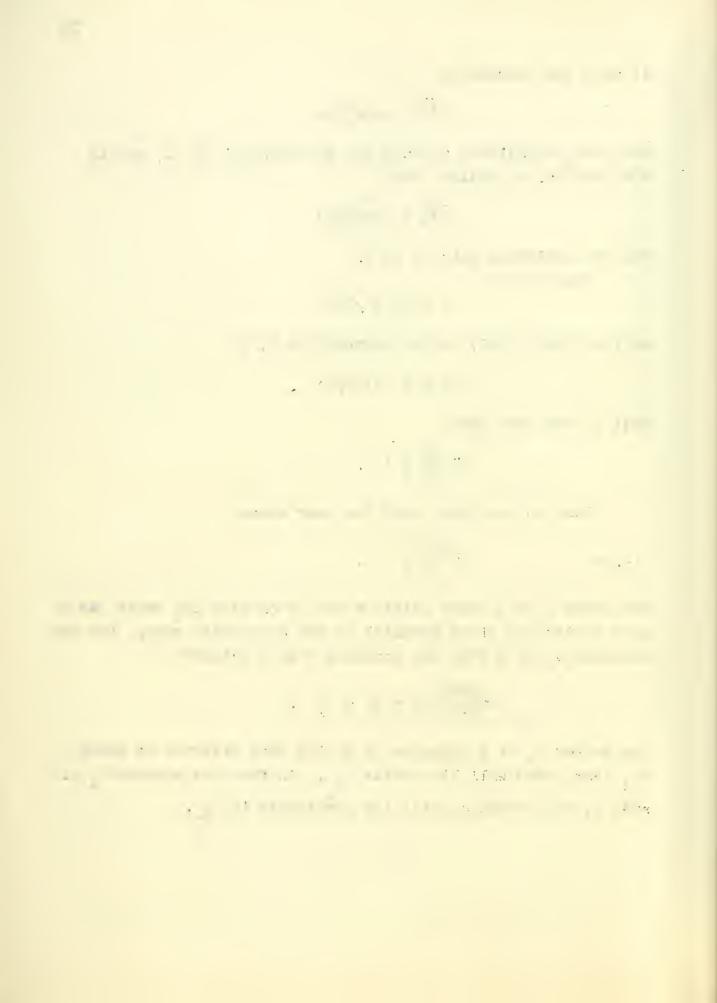
Take on the other hand the case where

$$(1.27) \sqrt{\varepsilon} \frac{a^2}{h^2} \le 1 .$$

For every Q in R there exists a cube C of side  $2\delta_1$  which has Q as a vertex and sides parallel to the coordinate axes. Let the distance  $\delta_2$  of Q from the boundary B of R satisfy

$$\frac{\sqrt{12 \ \epsilon}}{1-\epsilon} \delta_1 \leq \delta_2 \leq \delta_1$$
.

The vertex  $A_0$  of C opposite to Q will have distance at least  $2\delta_1$  from B and will lie inside  $R_{\delta_1}$ . In fact the subcube  $C_1$  of side  $\delta_1$  and vertex  $A_0$  will lie completely in  $R_{\delta_1}$ .



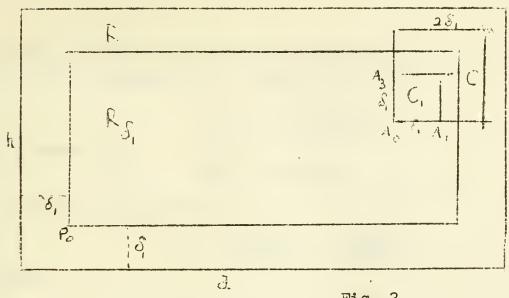


Fig. 2

Since C has diameter  $\triangle = \sqrt{12} \delta_1$  and distance  $\delta_2$  from B the cube C is a core since

$$\frac{\sqrt{\varepsilon}}{1-\varepsilon} \triangle = \frac{\sqrt{12\varepsilon}}{1-\varepsilon} \delta_1 \le \delta_2 \quad .$$

Let A1, A2, A3 be the vertices of the subcube C1 adjacent to A0. Put

$$r_{i} = \overline{A_{i}Q}$$
,  $r_{i}^{!} = \overline{A_{i}Q^{!}}$  (i = 0,2,3).

Then, since the  $A_i$  belong to  $R_{\delta_1}$ 

$$r_{i} = o(\delta_{1})$$

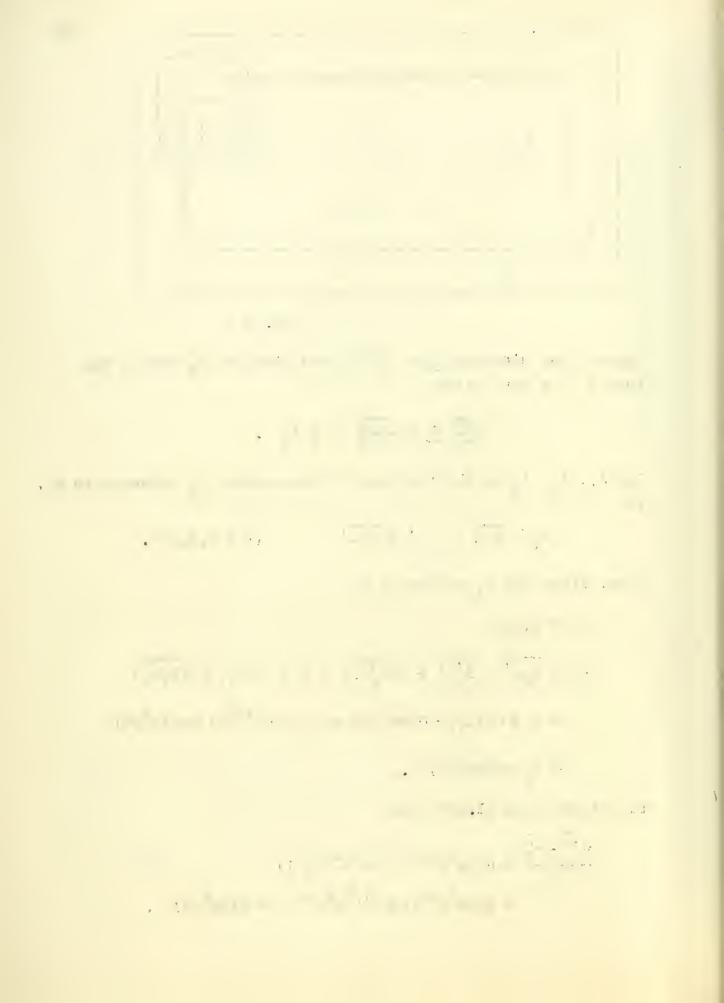
$$r_{i}' = \overline{A_{i}Q'} = \overline{A_{i}'Q'} + o(\overline{A_{i}'A_{i}}) = r_{i} + o(\epsilon r_{i}) + o(\overline{A_{i}'A_{i}})$$

$$= r_{i} + o(\epsilon \delta_{1}) + o(\epsilon a^{2}/h) = r_{i} + o(\epsilon^{3/2}h) + o(\epsilon a^{2}/h)$$

$$= r_{i} + o(\epsilon a^{2}/h) .$$

It follows from (1.27) that

$$\frac{\mathbf{r_i^2} - \mathbf{r_i^2}}{\delta_1} = 0(\epsilon a^2 h^{-1} (1 + \epsilon a^2 h^{-1} \delta_1^{-1}))$$
$$= 0(\epsilon a^2 h^{-1} (1 + \epsilon^{1/2} a^2 h^{-2})) = 0(\epsilon a^2 / h) .$$



Applying equation (1.15) with  $a_1 = a_2 = a_3 = \delta_1$  we find that for the coordinates  $x_i$  of Q and  $x_i^*$  of Q

$$x_i^{\dagger} - x_i = O(\epsilon a^2/h)$$

and hence that

$$\overline{Q'Q} = O(\epsilon a^2/h)$$

for any Q in  $R_{\delta_3}$ , where

$$\delta_3 = \frac{\sqrt{12 \, \epsilon}}{1 - \epsilon} \, \delta_1 = \frac{\sqrt{12 \, \epsilon}}{1 - \epsilon} \, h$$

Since

$$\delta_3 = O(\epsilon h) = O(\epsilon a^2/h)$$

it follows then that (1.26) holds for any Q in R (see page 19).

Summarizing we have the following theorem:

There exists a constant M with the following property: For any transformation  $x^i = f(x)$  with maximum elongation  $\epsilon$  applied to a square plate with sides a, a, h (where  $h \le a$ ) the maximum displacement of any point after applying a suitable orthogonal transformation is at most

$$Mea^2/h$$
 .

We can conclude that on the average the matrix p will differ little from the unit matrix and that the local rotation matrix c will also differ little from the identity.

First of all the symmetric matrix  $\eta$  can be brought into diagonal form by an orthogonal transformation. Since the eigenvalues of  $\eta$  are at most  $\varepsilon$  in absolute value and the orthogonal matrix has bounded elements, it follows that  $\eta = O(\varepsilon)$ . Then

$$p = c^*(1 + \eta) = 0(1 + \epsilon) = 0(1)$$
.

Consider now any parallel to the  $x_1$ -axis. On it ds =  $dx_1$  and

$$dx_{i}' = p_{i1}dx_{1} = p_{i1}ds$$
 ,  $\frac{ds'}{ds} = \sqrt{p_{11}^{2} + p_{21}^{2} + p_{31}^{2}}$  .

Since ds /ds lies between 1-E and 1+E we have

$$(1-\epsilon)^2 \le p_{11}^2 + p_{21}^2 + p_{31}^2 \le (1+\epsilon)^2$$
.

In particular,

$$|p_{11}| \leq 1+\varepsilon$$
.

Integrating over the portion of the parallel to the x<sub>1</sub>-axis lying inside R we have

$$\int dx_1' = \int dx_1 + O(\varepsilon(a^2/h)) ,$$

since the endpoints have moved at most by amounts  $O(\epsilon(a^2/h))$ . This relation can be written

$$\int p_{11} dx_1 = a + O(\epsilon(a^2/h))$$

or

$$\int (1+\varepsilon - p_{11}) dx_1 = O(\varepsilon(a^2/h) .$$

Integrating over all parallels to the x1-axis we find

$$\iiint_{\mathbb{R}} (1+\varepsilon - p_{11}) dx = O(\varepsilon a^3) \qquad (dx = dx_1 dx_2 dx_3) .$$

Here the integrand is non-negative. Since

$$|1 - p_{11}| \le |1 + \varepsilon - p_{11}| + |\varepsilon| = (1 + \varepsilon - p_{11}) + \varepsilon$$

it follows that

$$\iiint\limits_{R} |1 - p_{11}| dx = o(\epsilon a^3)$$

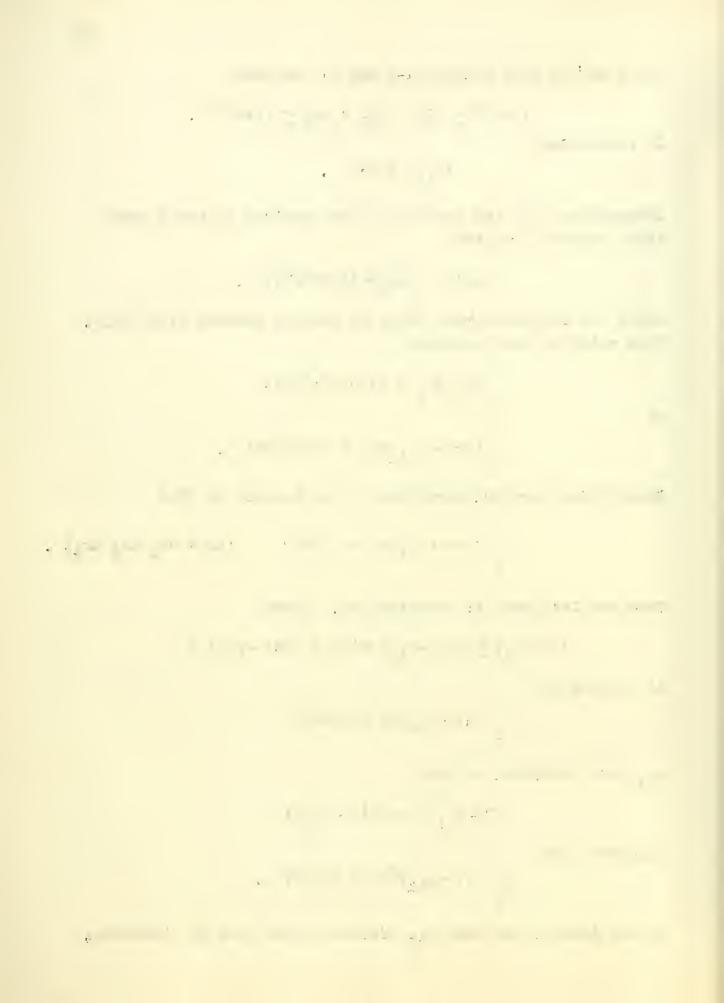
p11 being bounded, we have

$$(1-p_{11})^2 = 0(|1-p_{11}|)$$

and thus also

$$\iiint_{\mathbb{R}} (1-p_{11})^2 dx = o(\epsilon a^3) .$$

We see that in the mean pll differs little from 1. Moreover,



$$p_{21}^2 + p_{31}^2 \le (1+\epsilon - p_{11})(1+\epsilon + p_{11}) = 0(1+\epsilon - p_{11})$$
.

Hence

$$\iiint_{R} \left[ (p_{11} - 1)^2 + p_{21}^2 + p_{31}^2 \right] dx = O(\epsilon a^3) .$$

In the same manner one finds that

$$\iiint_{R} \left[ p_{12}^{2} + (p_{22} - 1)^{2} + p_{32}^{2} \right] dx = O(\epsilon a^{3}) ,$$

$$\iiint_{R} \left[ p_{13}^{2} + p_{23}^{2} + (p_{33} - 1)^{2} \right] dx = O(\epsilon(a^{4}/h)) .$$

Altogether

$$\iiint_{R} (p-1)(p^*-1)dx = O(\epsilon(a^{4}/h)) .$$

We measure the deviation of the rotation matrix c from unity by the matrix

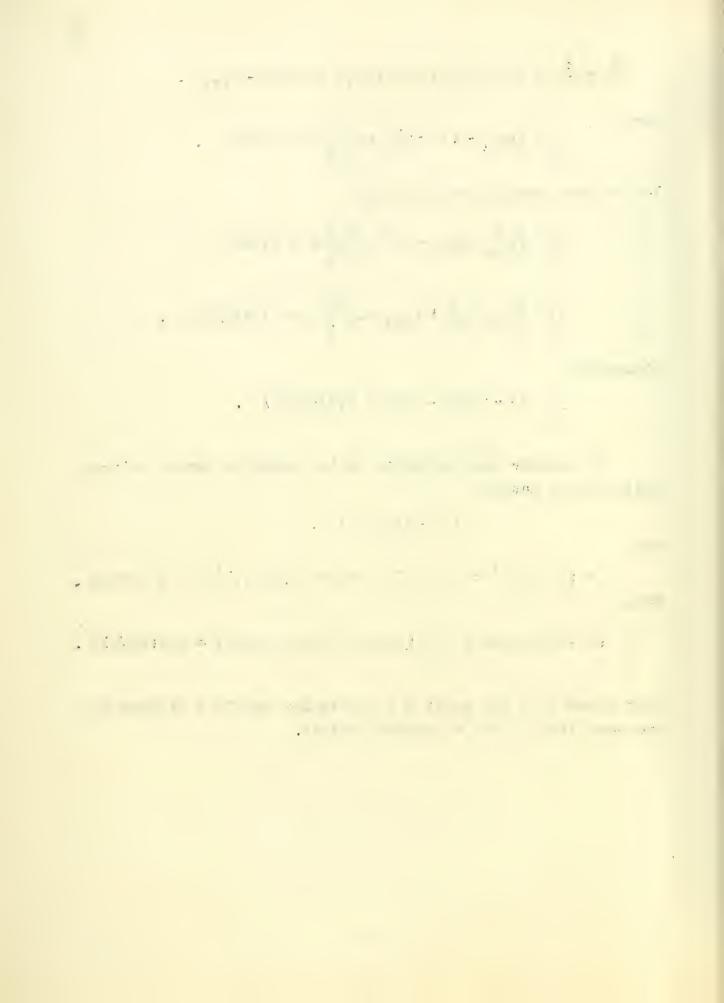
Now

$$c = (1 + \eta)p^{-1} = (1 + \eta)^{-1}p^* = p^* - \eta(1 + \eta)^{-1}p^* = p^* + O(\epsilon)$$

Hence

$$\iiint_{R} (c^*-1)(c-1)dx = \iiint_{R} (p-1)(p^*-1)dx + O(\epsilon a^3) = O(\epsilon(a^4/h)).$$

This shows that for small & the rotation matrix c differs in the mean little from a constant matrix.



## Chapter II

## Equations of Motion and Equilibrium in Lagrange Coordinates

A material is considered here in two states, "strained" and "unstrained". The Cartesian coordinates of a particle in the unstrained state ("Lagrange coordinates") shall be  $(x_1,x_2,x_3) = x$ . The Cartesian coordinates of the same particle in the strained state ("Euler coordinates") at the time t shall be  $(x_1^1,x_2^1,x_3^1) = x^1$ , where

$$x^{\dagger} = x^{\dagger}(x,t)$$
.

We introduce again the Jacobian matrix

(2.1) 
$$p = (p_{ik}) = \left(\frac{\partial x_i^{\dagger}}{\partial x_k}\right) .$$

Velocity and acceleration are respectively the vectors with components

$$\dot{\mathbf{x}}_{\mathbf{i}} = \frac{\partial \mathbf{x}_{\mathbf{i}}^{\dagger}}{\partial \mathbf{t}}$$
,  $\dot{\mathbf{x}}_{\mathbf{i}}^{\dagger} = \frac{\partial^{2} \mathbf{x}_{\mathbf{i}}^{\dagger}}{\partial \mathbf{t}^{2}}$ .

Let dV and  $\rho$  denote element of volume and density in the unstrained state, dV and  $\rho$  that in the strained state. Conservation of mass implies that for the element of mass

(2.2) 
$$dm = \rho dV = \rho^{\dagger} dV^{\dagger} ,$$

where

$$(2.3) dV' = |p|dV.$$

(Here |p, denotes the determinant of p, which is assumed to be positive.)

The stress in the strained state is described by a matrix  $\tau = (\tau_{ik})$ , such that the force acting on a surface element dS with unit normal  $\xi^{\dagger}$  in the strained state is given by

(2.4) 
$$\tau_{ik} \xi_k^{\dagger} dS^{\dagger} .$$

In addition to the stresses there may be present external forces represented by a vector  $F = (F_1, F_2, F_3)$ , such that  $FdV^{\dagger}$  represents the external force acting on a volume element  $dV^{\dagger}$  in the strained state.

7 . . , and the state of the state of the state of < . . . 0.0 : : 

Let R' be a region with boundary S' cut out from the material in the strained state. Let  $\xi_i^*$  be the components of the exterior unit normal of S'. Then Newton's law of motion gives

$$\iiint\limits_{\mathbf{R}^{\bullet}} \rho^{\bullet} \mathbf{x}_{1}^{\bullet} d\mathbf{V}^{\bullet} = \iiint\limits_{\mathbf{R}^{\bullet}} \mathbf{F}_{1} d\mathbf{V}^{\bullet} + \iint\limits_{\mathbf{S}^{\bullet}} \mathbf{\tau}_{1k} \boldsymbol{\xi}_{k}^{\bullet} d\mathbf{S}^{\bullet} .$$

Applying the divergence theorem to the surface integral and shrinking R<sup>1</sup> into a point we find the equations of motion

(2.5) 
$$\rho^{i} \overset{\text{e.i}}{x_{i}} = F_{i} + \frac{\partial \tau_{ik}}{\partial x_{k}^{i}}.$$

The work done by the surface stresses and exterior forces on R' in unit time is

(2.6) 
$$W = \iint_{S} \tau_{ik} \xi_{k} \dot{x}_{i} dS + \iiint_{R} F_{i} \dot{x}_{i} dV .$$

The kinetic energy of the mass in R is

(2.7) 
$$K = \frac{1}{2} \iiint_{R} \dot{x}_{i} \dot{x}_{i} \rho^{\dagger} dV$$

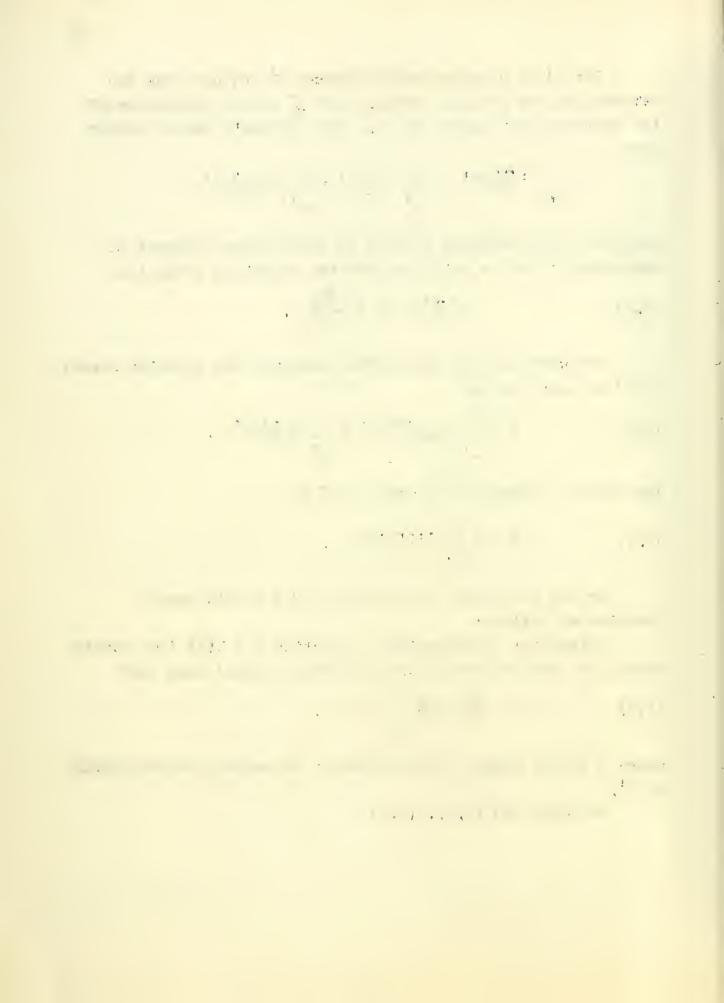
We now postulate the existence of a strain energy function as follows:

Principle: There exists a function U = U(p) (the strain energy per unit volume of the unstrained state) such that

$$(2.8) W = \frac{dK}{dt} + \frac{d}{dt} \iiint_{R} U dV ,$$

where R is the region with boundary S in x-space corresponding to R.

We have from (2.6), (2.5)



$$(2.9) \quad W = \iiint_{\mathbf{R}^{\dagger}} \left[ \frac{\partial}{\partial \mathbf{x}_{\mathbf{k}}^{\dagger}} \left( \tau_{\mathbf{i}\mathbf{k}} \dot{\mathbf{x}_{\mathbf{i}}^{\dagger}} \right) + F_{\mathbf{i}} \dot{\mathbf{x}_{\mathbf{i}}^{\dagger}} \right] dV'$$

$$= \iiint_{\mathbf{R}^{\dagger}} \left[ \left( \frac{\partial \tau_{\mathbf{i}\mathbf{k}}}{\partial \mathbf{x}_{\mathbf{k}}^{\dagger}} + F_{\mathbf{i}} \right) \dot{\mathbf{x}_{\mathbf{i}}^{\dagger}} + \tau_{\mathbf{i}\mathbf{k}} \frac{\partial \dot{\mathbf{x}_{\mathbf{i}}^{\dagger}}}{\partial \mathbf{x}_{\mathbf{k}}} \right] dV'$$

$$= \iiint_{\mathbf{R}^{\dagger}} \left[ \rho' \dot{\mathbf{x}_{\mathbf{i}}^{\dagger}} \dot{\mathbf{x}_{\mathbf{i}}^{\dagger}} + \tau_{\mathbf{i}\mathbf{k}} \frac{\partial \dot{\mathbf{x}_{\mathbf{i}}^{\dagger}}}{\partial \mathbf{x}_{\mathbf{r}}} \frac{\partial \mathbf{x}_{\mathbf{i}}}{\partial \mathbf{x}_{\mathbf{k}}^{\dagger}} \right] dV'$$

$$= \frac{dK}{dt} + \iiint_{\mathbf{R}^{\dagger}} v_{\mathbf{i}\mathbf{k}} \dot{\mathbf{p}}_{\mathbf{i}\mathbf{r}} \dot{\mathbf{p}}^{\mathbf{r}\mathbf{k}} dV'$$

where

$$g^{-1} = (x^{ik})$$

is the matrix reciprocal to p. On the other hand

(2.10) 
$$\frac{d}{dt} \iiint_{R} U(p) dV = \iiint_{R} \frac{\partial U}{\partial p_{ir}} \dot{p}_{ir} dV = \iiint_{R} \frac{1}{|p|} q_{ir} \dot{p}_{ir} dV',$$

where the qir are the functions of p defined by

(2.11) 
$$q_{ir}(p) = \frac{\partial U(p)}{\partial p_{ir}} .$$

By comparison of (2.8), (2.9), (2.10), we find for R'shrinking into a point that

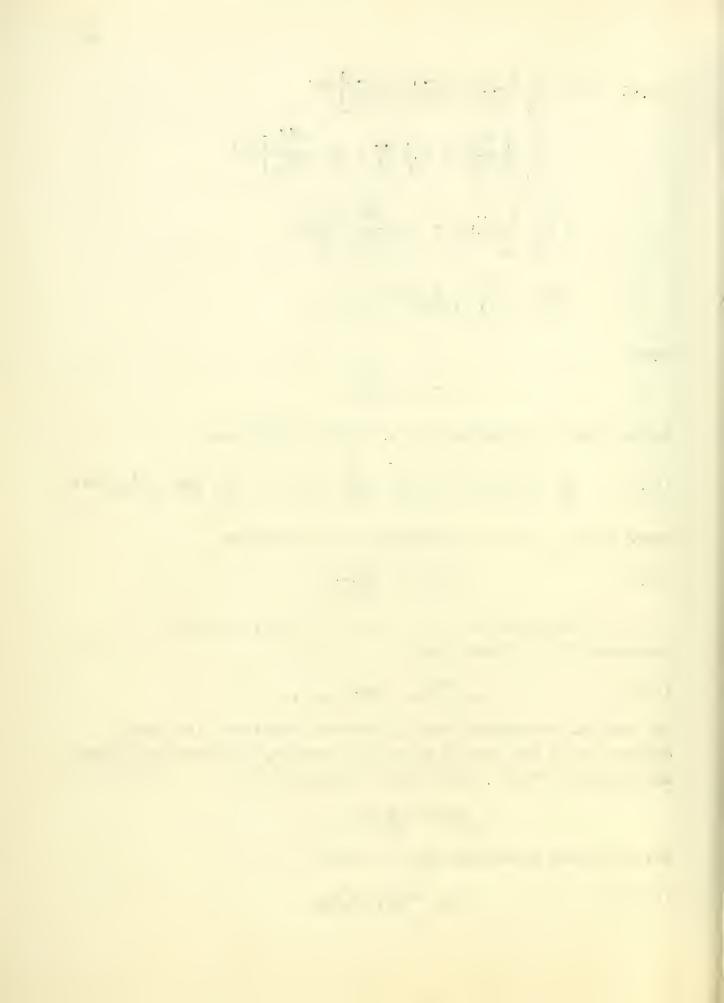
$$(2.12) \tau_{ik} p^{rk} \dot{p}_{ir} = \frac{1}{1p!} q_{ir} \dot{p}_{ir} .$$

We make the assumption that a particle, whatever its local deformation p and stress at a given moment, can have any given deformation rate p. Then (2.12) implies that

$$q_{ik}p^{rk} = \frac{1}{|p|}q_{ir}$$
.

Solving these equations for p we find

$$\tau_{ik} = \frac{1}{|p|} q_{ir} p_{kr}$$



or, in matrix notation,

Relations (2.14) take the place of the stress-strain relations of the classical linear theory, supplementing the equations of motion (2.5). We next derive the equations of motion and boundary conditions in Lagrange coordinates x.

We have from (2.10), (2.8), (2.6), (2.7)

$$\frac{d}{dt} \iiint_{R} U(p) dV = \iiint_{R} q_{ir} \dot{p}_{ir} dV = \iiint_{S} q_{ir} \dot{x}_{i}^{!} \xi_{r} dS - \iiint_{R} \frac{\partial q_{ir}}{\partial x_{r}} \dot{x}_{i}^{!} dV$$

$$= \iint_{S} q_{ir} \dot{x}_{i}^{!} \xi_{r} dS - \iiint_{R'} \frac{\partial q_{ir}}{\partial x_{r}} \dot{x}_{i}^{!} |p|^{-1} dV' = W - \frac{dK}{dt}$$

$$= \iint_{S'} \tau_{ik} \xi_{k}^{!} \dot{x}_{i}^{!} dS' + \iiint_{R'} [F_{i} - \rho' \dot{x}_{i}^{!}] \dot{x}_{i}^{!} dV' .$$

At any given moment t the  $\dot{x}_1^t$  are independent of the  $p_{ik}$  and  $q_{ik}$  and can be taken as arbitrary functions. Taking first for the  $\dot{x}_1^t$  arbitrary functions vanishing on S respectively S' we find by comparison of integrands that l

(2.15) 
$$\rho'x_{1}' = F_{1} + \frac{1}{|p|} \frac{\partial q_{1p}}{\partial x_{p}} .$$

Comparing next the remaining surface integrals for x! arbitrary on S we obtain the relations

(2.16) 
$$q_{ir}\xi_r dS = \tau_{ik}\xi_k' dS'.$$

Here  $dS^{\dagger}$  is the surface element in the strained state with unit normal  $\xi^{\dagger}$  that originates from a surface element dS with unit normal  $\xi$  in the unstrained state.

If U is known in its dependence on p and if the  $\mathbb{F}_i$  are known as functions of x or x' and t, we have in (2.15) a system of second order partial differential equations for the  $x_i^*$  as

These equations are given by Novozhilov [1] III (34), and Reissner [5] for the case of time-independent transformations.



functions of the  $\mathbf{x}_k$ . Stresses and strains in the ordinary sense do not enter these equations of motions. Their place is taken by the matrices q and p, which are connected by the relations

(2.17) 
$$q_{ik} = \frac{2U(p)}{\partial p_{ik}} .$$

Boundary conditions of the type of restrictions on the  $\mathcal{T}_{ik}$  on S: yield by (2.16) restrictions on the  $q_{ik}$  on S.

In the special case corresponding to elastic equilibrium and in the absence of external forces the x! are independent of t, and the equilibrium equations take the simple form

$$\frac{\partial q_{ik}}{\partial x_k} = 0 .$$

Making use of (2.17) we obtain from (2.18) a quasi-linear system of first order equations for the  $p_{ik}$ , which has to be supplemented by the obvious "compatibility conditions" for the  $p_{ik}$ 

$$\frac{\partial p_{ik}}{\partial x_n} = \frac{\partial p_{ir}}{\partial x_k} .$$

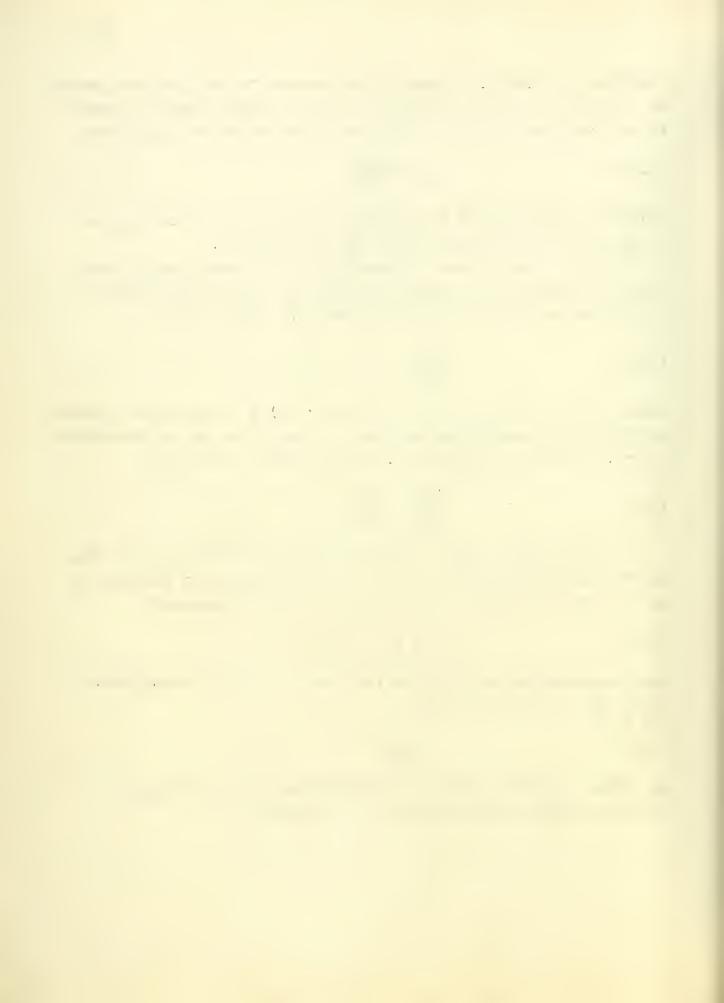
The two most common types of boundary conditions are that either the displacements or the tractions vanish on portions of S. In the first case we have  $x_1^* - x_2^* = 0$  or equivalently

$$(2.20) p_{ik} - \delta_{ik} = \sigma_i \xi_k$$

with suitable proportionality factors  $\sigma_i$ . In the second case  $\tau_{ik}\xi_k'$  dS' = 0 or by (2.16)

(2.21) 
$$q_{ik}\xi_k = 0$$
.

In either case the problem of determining the x! or pik is described completely in Lagrange coordinates.



## Chapter III

Invariants of the Strain Energy Function

We assume that the strain energy density U(p) is independent of the choice of the cartesian coordinate system used. This amounts to the requirement that

$$(3.1) U(p) = U(cpc*)$$

for any orthogonal matrix c. In addition we restrict ourselves to an isotropic medium. This has as a consequence that U(p) stays unchanged if a rigid motion is superimposed on the transformation x' = f(x) with Jacobian matrix p, i.e.

(3.2) 
$$U(p) = U(cp)$$

for orthogonal c. Altogether then

$$U(p_1) = U(p_2)$$

for two matrices p1, p2 that are connected by a relation

(3.3) 
$$p_2 = cp_1c$$

where c and C are orthogonal. The necessary and sufficient dondition for two matrices  $p_1$ ,  $p_2$  to be connected by a relation (3.3) with orthogonal c,C is that the matrices  $p_1^*p_1$  and  $p_2^*p_2$  have the same eigenvalues. Consequently our assumptions imply that U(p) is a symmetric function of the eigenvalues of  $p^*p = g$ .

Any symmetric function of the eigenvalues of a matrix g can be expressed in terms of any three independent symmetric functions of the eigenvalues. We choose here as basic symmetric functions the

(3.4) 
$$s_n = [(g-1)^n]$$
 for  $n = 1,2,3$ ,

where we denote generally by [a] the trace of the matrix a. Then

(3.5) 
$$U(p) = V(s_1, s_2, s_3) .$$



It is easy to establish the connection between the relations given here and those given customarily, in which the strain matrix is taken to be

$$\zeta = \frac{1}{2}(g - 1)$$
.

If  $\lambda_1, \lambda_2, \lambda_3$  denote the eigenvalues of  $\zeta$  and  $I_1, I_2, I_3$  the elementary symmetric functions of those eigenvalues, we have

(3.5a) 
$$\begin{cases} s_1 = \sum_{k} 2\lambda_k = 2I_1 \\ s_2 = \sum_{k} 4\lambda_k^2 = 4(I_1^2 - 2I_2) \\ s_3 = \sum_{k} 8\lambda_k^3 = 8(I_1^3 - 3I_1I_2 + 3I_3) \end{cases}$$

Then  $U=V(s_1,s_2,s_3)$  becomes a function of the  $I_k$ . One can write U in the form  $U=\psi(\zeta)$  where  $\psi(\zeta)=\psi(\zeta^*)$ . Then

$$\frac{\partial \psi(\zeta)}{\partial \zeta_{ik}} = \frac{\partial \psi(\zeta)}{\partial \zeta_{ki}}$$
 for  $\zeta = \zeta^*$ .

One obtains, using this symmetry property,

$$q_{ik} = \frac{\partial \Psi}{\partial \xi_{ks}} p_{is}$$
  $g_{nk} = \frac{\partial \Psi}{\partial F_{nk}}, p_{nk} = \frac{\partial Y}{\partial X_{ok}}$ 

and from  $(2.13)^2$ 

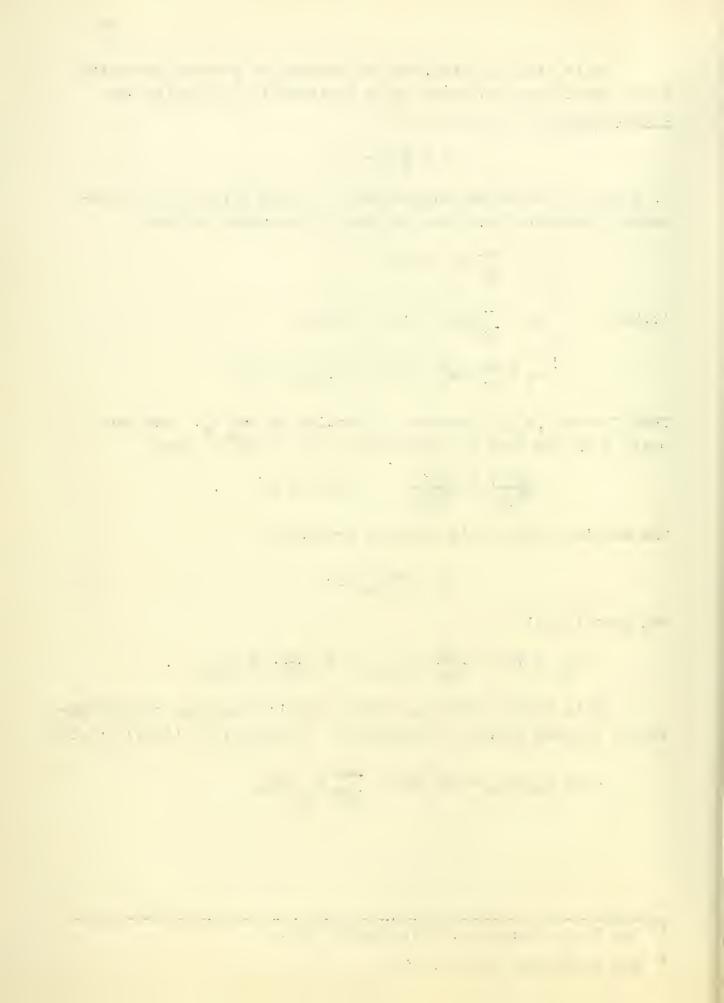
$$\tau_{\rm ik} = |{\bf p}|^{-1} \, \tfrac{\partial \psi}{\partial \zeta_{\rm rs}} \, {\bf p}_{\rm is} {\bf p}_{\rm kr} = \tfrac{\rho \, \prime}{\rho} \, \tfrac{\partial \psi}{\partial \zeta_{\rm rs}} \, {\bf p}_{\rm is} {\bf p}_{\rm kr} \quad . \label{eq:tik}$$

It is easy to find the stress strain relations corresponding to a given choice of function V. We have from (2.17), (3.5)

$$dU = q_{ik} dp_{ik} = [q^* dp] = \sum_{n=1}^{3} V_{s_n} ds_n$$
.

See F. D. Murnaghan, [2], Chapter III.

See Murnaghan, [2], p. 56.



By (3.4)

$$ds_{n} = n[(g-1)^{n-1} dg] = n[(g-1)^{n-1} d(p^{*}p)]$$

$$= n[(g-1)^{n-1} (dp^{*})p + (g-1)^{n-1}p^{*} (dp)]$$

$$= 2n[(g-1)^{n-1}p^{*} (dp)] .$$

Hence by comparison, for arbitrary dp,

$$q^* = \sum_{n=1}^{3} 2nV_{s_n} (g-1)^{n-1} p^*$$

or

(3.6) 
$$q = \sum_{n=1}^{3} 2nV_{s_n} p(p^*p-1)^{n-1} = \sum_{n=1}^{3} 2nV_{s_n} (pp^*-1)^{n-1} p .$$

It follows from (2.14) that the stress matrix & is given by

(3.7) 
$$\tau = |p|^{-1} \sum_{n=1}^{3} 2n V_{s_n} (pp^*-1)^{n-1} pp^*.$$

This formula makes evident that the stress matrix t is symmetric under our assumptions.

For small deformations we have

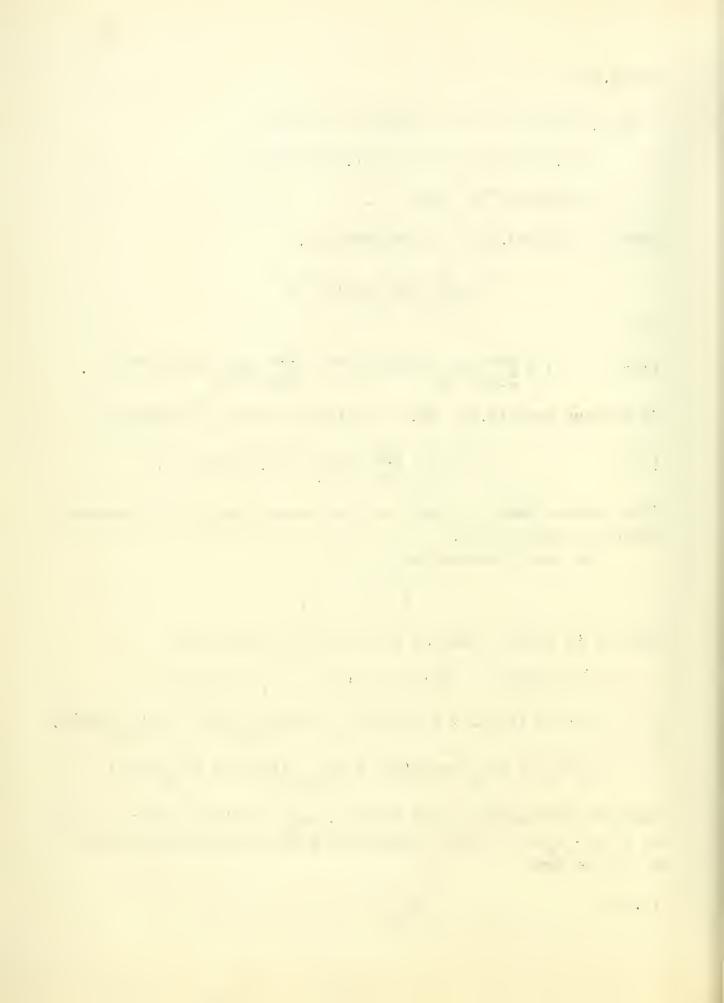
$$p = 1 + P ,$$

where P is small. Then up to terms of second order

$$\begin{aligned} &\text{pp} &\approx 1 + \text{P} + \text{P}^*, & |\text{p}| \approx 1 + |\text{P}|, & \text{s}_1 \approx |\text{P} + \text{P}^*| \\ &\text{V} \approx (1 - |\text{P}|)(1 + |\text{P}| + |\text{P}^*|)(2\text{V}_{\text{S}_1} + 2\text{V}_{\text{S}_1\text{S}_1}(\text{P} + \text{P}^*) + 4\text{V}_{\text{S}_2}(\text{P} + \text{P}^*)) \\ &\approx 2\text{V}_{\text{S}_1} + 2\text{V}_{\text{S}_1}(\text{P} + \text{P}^* - |\text{P}|) + 2\text{V}_{\text{S}_1\text{S}_1}(\text{P} + \text{P}^*) + 4\text{V}_{\text{S}_2}(\text{P} + \text{P}^*) \end{aligned} ,$$

where the derivatives of  $V = V(s_1, s_2, s_3)$  are taken for p = 1 or  $s_1 = s_2 = s_3 = 0$ . If we require that the stress vanishes for p = 0, we have

$$V_{s_1} = 0$$
.



Put for  $s_1 = s_2 = s_3 = 0$ 

(3.8b) 
$$4V_{s_1s_1} = \lambda$$
,  $4V_{s_2} = \mu$ ;

then for small deformations

(3.8c) 
$$\tau \approx \lambda[P] + \mu(P + P^*)$$
.

This is the classical linear approximation for the stress strain relations with the Lamé constants  $\lambda$  and  $\mu$ . The constants  $\lambda$  and  $\mu$  are necessarily positive by nature.

For later purposes it will be important to have the expansion of U(p) = U(1+P) in the neighborhood of P = 0 up to terms of fourth order in P.

According to (3.7) the stress matrix  $\mathcal{L}$  is expressible in terms of the symmetric matrix

$$\gamma = pp^*$$
.

We have

(3.8d) 
$$\tau = |\gamma|^{-1/2} \frac{3}{\sum_{n=1}^{\infty}} 2n V_{s_n} (\gamma - 1)^{n-1} \gamma$$

where

$$V = V(s_1, s_2, s_3)$$
 ,  $s_n = [(\gamma-1)^n]$  .

We can expand  $\tau$  into a Taylor series of powers of  $\gamma$ -1 and find from (3.8a,b)

$$\tau = (1 - \frac{1}{2}[\gamma - 1] + \dots)(2V_{s_1} + 2V_{s_1s_1}[\gamma - 1] + 4V_{s_2}(\gamma - 1) + \dots)$$

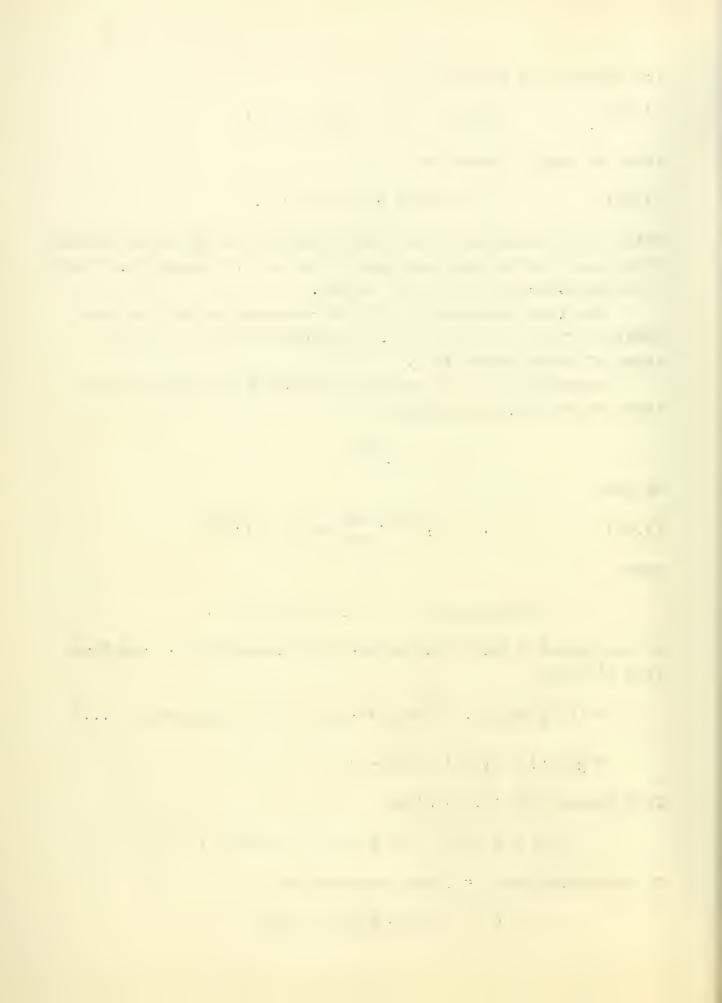
$$= \frac{\lambda}{2}[\gamma - 1] + \mu(\gamma - 1) + O((\gamma - 1)^2) .$$

This agrees with (3.8c) since

$$\gamma-1 = P + P^* + PP^* = P + P^* + O(P^2)$$
.

In particular for  $\gamma = 1$  and arbitrary  $d\gamma$ 

$$\tau = 0$$
 ,  $d\tau = \frac{\lambda}{2}[d\gamma] + \mu d\gamma$  .



If dt = 0 for  $\gamma = 1$  and some  $d\gamma$  then

$$0 = \frac{5}{\lambda}[d\gamma] + \mu d\gamma$$

and hence also

$$0 = [0] = (\frac{3}{2}\lambda + \mu)[d\gamma]$$
.

Since  $\lambda$  and  $\mu$  are positive this would imply that  $[d\gamma] = 0$  and thus also  $d\gamma = 0$ .

We can conclude that if  $\gamma$ -1 is sufficiently small then  $T \neq 0$  for  $\gamma$ -1  $\neq 0$ . Otherwise there would exist a sequence of matrices  $\gamma_n$  of the form  $\gamma_n = 1 + \epsilon_n \sigma_n$ , with scalars  $\epsilon_n$  tending to zero and matrices  $\sigma_n$  which are bounded and bounded away from zero such that the corresponding stress matrices  $\tau_n$  vanish. For a suitable subsequence the  $\sigma_n$  would converge towards a nonvanishing matrix d $\gamma$  for which dT = 0, in contradiction to what has been proved.

The matrix  $\gamma = pp^*$  has the same invariants as the matrix  $g = p^*p$ . Vanishing of  $\gamma$ -1 means that p is orthogonal; if the strains are small then  $\gamma$ -1 is small. We find then that for sufficiently small strains, i.e. for sufficiently small g-1, the stress matrix  $\tau$  can only vanish for orthogonal p. Since by (2.14)  $\tau$  and q vanish simultaneously, it follows then also that  $q \neq 0$  for sufficiently small strains unless p is orthogonal.

We have from Taylor's formula

$$U = V(s_1, s_2, s_3) = \frac{1}{i_1, i_2, i_3} \frac{1}{i_1! i_2! i_3!} V_{i_1 i_2 i_3} s_1^{i_1} s_2^{i_2} s_3^{i_3}$$

where

$$v_{i_1 i_2 i_3} = \left(\frac{\frac{i_1 + i_2 + i_3}{\delta_1 + i_2 + i_3} v}{(\delta_{s_1})^{i_1} (\delta_{s_2})^{i_2} (\delta_{s_3})^{i_3}}\right)_{s_1 = s_2 = s_3 = 0}.$$



We have for i = 1,2,3

$$s_{i} = [(g-1)^{i}] = [(P + P^{*} + P^{*}P)^{i}]$$

$$= \sum_{\alpha+\beta=i} \frac{i!}{\alpha!\beta!} [(P+P^{*})^{\alpha}(P^{*}P)^{\beta}] = \sum_{\alpha+\beta=i} s_{\alpha\beta}$$

where

(3.9) 
$$s_{\alpha\beta} = \frac{(\alpha+\beta)!}{\alpha!\beta!} \left[ (P + P^*)^{\alpha} (P^*P)^{\beta} \right].$$

Then

$$s_{1}^{i_{1}} s_{2}^{i_{2}} s_{3}^{i_{3}} = (s_{10} + s_{01})^{i_{1}} (s_{20} + s_{11} + s_{02})^{i_{2}} (s_{30} + s_{21} + s_{12} + s_{03})^{i_{3}}$$

$$= i_{1}! i_{2}! i_{3}! \sum_{i+k \leq 3} \frac{s_{ik}^{ik}}{a_{ik}!}$$

where the sum is extended over all sets of non-negative integers  $(a_{10}, a_{01}, a_{20}, a_{11}, a_{02}, a_{30}, a_{21}, a_{12}, a_{03})$  with

(3.10) 
$$\alpha_{10} + \alpha_{01} = i_1$$
,  $\alpha_{20} + \alpha_{11} + \alpha_{02} = i_2$ ,  $\alpha_{30} + \alpha_{21} + \alpha_{13} + \alpha_{03} = i_3$ .

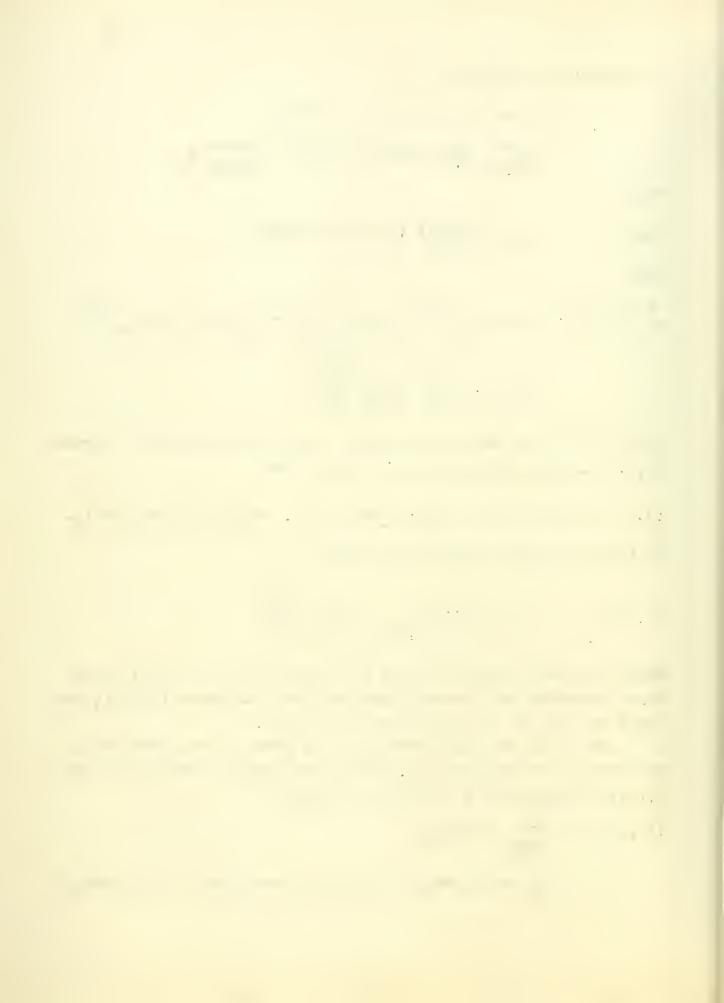
We finally arrive then at the formula

$$(3.11) U = \sum V_{i_1 i_2 i_3} \prod_{i+k \leq 3}^{\alpha_{ik}} \frac{s_{ik}^{\alpha_{ik}}}{\alpha_{ik}!}$$

where the sum is extended over all vectors  $(a_{10}, \ldots, a_{03})$  whose nine components are non-negative integers and where  $i_1, i_2, i_3$  are the functions of the  $a_{11}$  defined by (3.10).

By (3.9) the expression  $s_{\alpha\beta}$  is a form in the elements of the matrix P of degree  $\alpha+2\beta$ . Hence the general term in the sum (3.11) representing U is a form of degree

(3.12) 
$$N = \sum_{i+k \le 3} (i+2k)\alpha_{ik}$$
$$= \alpha_{10} + 2(\alpha_{01} + \alpha_{20}) + 3(\alpha_{11} + \alpha_{30}) + 4(\alpha_{02} + \alpha_{21}) + 5\alpha_{12} + 6\alpha_{03}.$$



Each set of non-negative integers satisfying this relation furnishes a homogeneous term of degree N in the expansion for U. Without attempting to determine in general the number of terms of degree N it is easy to find successively all solutions  $\alpha_{ik}$  of (3.12) for N = 0,1,2,3,4. One finds in this way

Without restriction of generality we can assume that  $V_{000} = 0$ . Moreover by (3.8a,b)

$$V_{loo} = V_{s_1} = 0$$
,  $4V_{200} = \lambda$ ,  $L_{l}V_{olo} = \mu$ .

The cubic terms contain 3 additional constants

$$A = \frac{1}{6} V_{300}$$
,  $B = V_{110}$ ,  $C = V_{001}$ 

and the fourth degree terms, 4 further constants

$$D = \frac{1}{24} V_{400}$$
,  $E = \frac{1}{2} V_{210}$ ,  $F = V_{101}$ ,  $G = \frac{1}{2} V_{020}$ .

We write

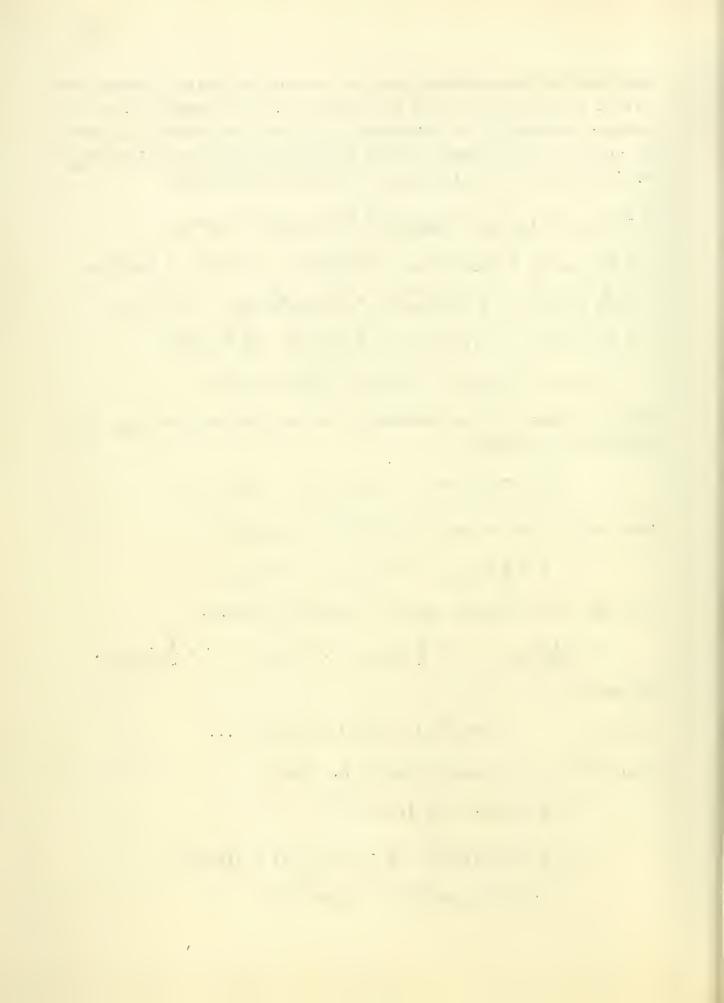
$$(3.1) U = U^{2}(P) + U^{3}(P) + U^{4}(P) + ...$$

where  $U^{k}(P)$  is a form of degree k. Here

$$u^2 = \frac{\lambda}{8} [P+P^*]^2 + \frac{\mu}{4} [(P+P^*)^2]$$

$$U^{3} = \frac{\lambda}{4} [P+P^{*}][P^{*}P] + \frac{\mu}{2} [(P+P^{*})P^{*}P] + A[P+P^{*}]^{3}$$

$$+ B[P+P^{*}][(P+P^{*})^{2}] + C[(P+P^{*})^{3}]$$



For any matrix  $\xi = (\xi_{ik})$  we use the symbol  $D_{\xi}$  for the differential operator defined by

$$D_{\xi} = \xi_{ik} \frac{\partial}{\partial p_{ik}} .$$

We have for independent matrices  $\xi, \eta, \zeta, \sigma, \ldots$  and for p = 1

$$D_{\xi}U = D_{\xi}U^{1} = U^{1}(\xi) , \qquad D_{\xi}D_{\eta}U = D_{\xi}D_{\eta}U^{2} = U^{2}(\xi,\eta) ,$$

$$D_{\xi}D_{\eta}D_{\xi}U = D_{\xi}D_{\eta}D_{\xi}U^{3} = U^{3}(\xi,\eta,\xi) ,$$

$$D_{\xi}D_{\eta}D_{\xi}D_{\sigma}U = D_{\xi}D_{\eta}D_{\xi}D_{\sigma}U^{4} = U^{4}(\xi,\eta,\xi,\sigma) .$$

Here for the form  $U^k(P)$  of degree k we have in  $U^k(\xi^1,\xi^2,\ldots,\xi^k)$  a polar form of  $U^k$ , which has the value  $k!U^k(P)$  for  $\xi^1=\xi^2=\ldots=\xi^k=P$ , and is symmetric in  $\xi^1,\xi^2,\ldots,\xi^k$ . It is simpler to write the expressions for these polar forms, which have the k-derivatives of U with respect to all the  $p_{ik}$  as coefficients, than to give expressions for the individual coefficients. We find immediately by polarisation from the expressions derived for the  $U^k(P)$  that for p=1

$$(3.15a) \quad D_{\xi}D_{\eta}U = U^{2}(\xi, \gamma) = \sum_{\xi, \eta} \left\{ \frac{\lambda}{8} [\xi + \xi^{*}] [\eta + \eta^{*}] + \frac{\mu}{4} [(\xi + \xi^{*}) (\eta + \eta^{*})] \right\}$$
$$= \lambda [\xi] [\eta] + \mu [\xi \eta + \xi \eta^{*}]$$

$$(3.15b) \quad D_{\xi}D_{\eta}D_{\zeta}U = U^{3}(\xi,\eta,\zeta)$$

$$= \sum_{\xi,\eta,\zeta} \left\{ \frac{\lambda}{4} [\xi + \xi^{*}] [\eta^{*}\zeta] + \frac{\mu}{2} [(\xi + \xi^{*})\eta^{*}\zeta] + A[\xi + \xi^{*}] [\eta + \eta^{*}] [\zeta + \zeta^{*}] + B[\xi + \xi^{*}] [(\eta + \eta^{*})(\zeta + \zeta^{*})] + C[(\xi + \xi^{*})(\eta + \eta^{*})(\zeta + \zeta^{*})] \right\}$$



$$\begin{array}{lll} (3.15c) & \mathsf{D}_{\xi} \mathsf{D}_{\eta} \mathsf{D}_{\zeta} \mathsf{D}_{\sigma} \mathsf{U} = \mathsf{U}^{\frac{1}{4}}(\xi,\eta,\zeta,\sigma) \\ & = \underbrace{\sum_{\xi,\eta,\zeta,\sigma}} \left\{ \frac{\lambda}{8} \left[ \xi^{*}\eta \right] [\zeta^{*}\sigma] + \frac{\mu}{4} \left[ \xi^{*}\eta \zeta^{*}\sigma \right] \right. \\ & + \left. 3 \mathsf{A} \left[ \xi + \xi^{*} \right] [\eta + \eta^{*}] [\zeta^{*}\sigma] + 2 \mathsf{B} \left[ \xi + \xi^{*} \right] \left[ (\eta + \eta^{*}) \zeta^{*}\sigma \right] \\ & + \mathsf{B} \left[ \xi^{*}\eta \right] \left[ (\zeta + \zeta^{*}) (\sigma + \sigma^{*}) \right] \\ & + \mathsf{D} \left[ \xi + \xi^{*} \right] \left[ \eta + \eta^{*} \right] \left[ (\zeta + \zeta^{*}) (\sigma + \sigma^{*}) \right] \\ & + \mathsf{E} \left[ \xi + \xi^{*} \right] \left[ (\eta + \eta^{*}) (\zeta + \zeta^{*}) (\sigma + \sigma^{*}) \right] \\ & + \mathsf{G} \left[ (\xi + \xi^{*}) (\eta + \eta^{*}) \right] \left[ (\zeta + \zeta^{*}) (\sigma + \sigma^{*}) \right] \\ & + 3 \mathsf{C} \left[ (\xi + \xi^{*}) (\eta + \eta^{*}) \zeta^{*}\sigma \right] \right\} \,. \end{array}$$

Here the sums on the right are to be extended over all permutations of  $\xi,\eta$  respectively of  $\xi,\eta,\zeta$  respectively of  $\xi,\eta,\zeta,\sigma$ .

We notice in particular that

$$\frac{\partial q_{ik}}{\partial p_{rs}} \Big|_{p=1} \xi_{ik} \eta_{rs} = \left( \frac{\partial^2 U(p)}{\partial p_{ik} \partial p_{rs}} \right)_{p=1} \xi_{ik} \eta_{rs} = D_{\xi} D_{\eta} U$$

$$= U^2(\xi, \eta) = \lambda[\xi][\eta] + \mu[\xi_{\eta} + \xi_{\eta}^*]$$

$$= \lambda \xi_{ii} \eta_{kk} + \mu \xi_{ik} (\eta_{ik} + \eta_{ki}) .$$

Having the expressions for the derivatives of U at p = 1 we can easily derive expressions for the derivatives of U(p) at any orthogonal p = c. We have indeed by (3.2)

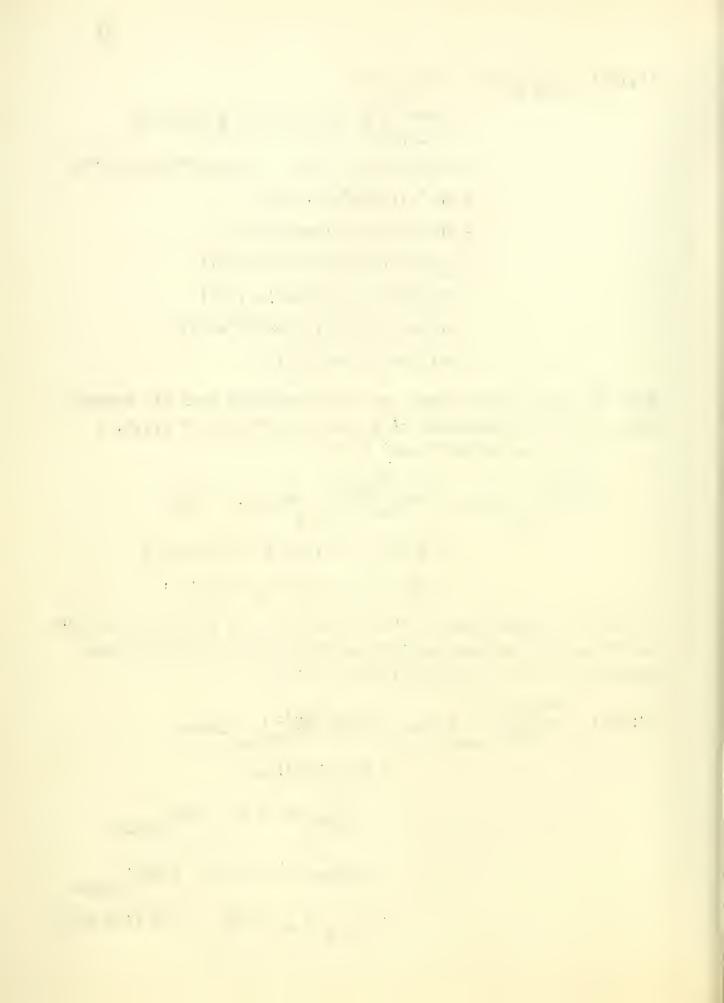
$$(3.17a) \left(\frac{\partial q_{ik}(p)}{\partial p_{rs}}\right)_{p=c} \xi_{ik} \eta_{rs} = \left(\frac{\partial^{2} U(p)}{\partial p_{ik} \partial p_{rs}}\right)_{p=c} \xi_{ik} \eta_{rs}$$

$$= \left(D_{\xi} D_{\eta} U(p)\right)_{p=c}$$

$$= \left(\frac{\partial^{2}}{\partial \alpha \partial \beta} U(c + \alpha \xi + \beta \eta)\right)_{\alpha=\beta=0}$$

$$= \left(\frac{\partial^{2}}{\partial \alpha \partial \beta} U(1 + c^{*}\alpha \xi + c^{*}\beta \eta)\right)_{\alpha=\beta=0}$$

$$= \left(D_{c}^{*}\xi D_{c}^{*}\eta U(p)\right)_{p=1} = U^{2}(c^{*}\xi, c^{*}\eta)$$



with  $U^2$  given by (3.15a). Similarly for c orthogonal (3.17b)  $[D_{\xi}D_{\eta}D_{\zeta}U(p)]_{p=c} = U^3(c^*\xi, c^*\eta, c^*\zeta) .$ 



## Chapter IV

The Second Variation of the Strain Energy. Uniqueness at Equilibrium for Prescribed Boundary Displacements.

We consider an equilibrium position  $x! = x^{\circ}(x)$  of the material maintained by certain body forces  $F_i^{\circ}$  and certain prescribed tractions or displacements on the boundary  $S_i^{\circ}$  of the region  $R_i^{\circ}$  filled by the material in the strained state. Let

(4.1) 
$$p_{ik}^{o} = \frac{\partial x_{i}^{o'}}{\partial x_{k}}, \qquad q_{ik}^{o} = \left(\frac{\partial U(p)}{\partial p_{ik}}\right)_{p=p}^{o}.$$

Then by (2.15), (2.16)

$$\frac{\partial q_{ir}^{\circ}}{\partial x_{r}} + |p^{\circ}|F_{i} = 0 \qquad \text{in } F$$

(4.3) 
$$q_{ir}^{o}\xi_{r}ds = t_{ik}^{o}\xi_{k}^{o'}ds_{o}^{i}$$
 for x on S,  $x_{o}^{o'}$  on  $s_{o}^{i}$ .

Let x' = x'(x) be an arbitrary transformation (not necessarily corresponding to an equilibrium state). We compute the difference of the total strain energies corresponding to the transformations x' and  $x^{0'}$ . We have, expanding U(p) by Taylor's theorem.

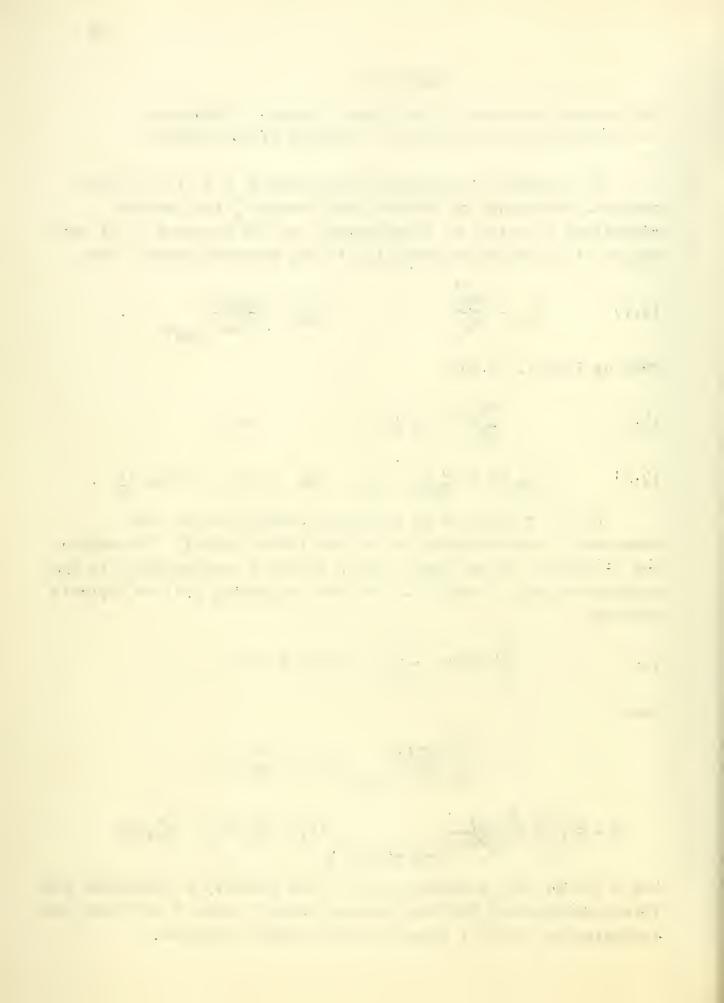
$$\iiint\limits_R U(p)dV - \iiint\limits_R U(p^\circ)dV = A + B ,$$

where

$$A = \iiint_{R} \left( \frac{\partial U(p)}{\partial p_{ik}} \right)_{p=p^{o}} (p_{ik} - p_{ik}^{o}) dV$$

$$B = \frac{1}{2} \iiint_{R} \left( \frac{\partial^{2} U(p)}{\partial p_{ik} \partial p_{rs}} \right)_{p=p^{o}+\theta(p-p^{o})} (p_{ik} - p_{ik}^{o}) (p_{rs} - p_{rs}^{o}) dV$$

and  $\theta = \theta(x)$  lies between 0 and 1. The quantity A represents the first variation of the total strain energy, while B (at least for infinitesimal  $x'-x^{0'}$ ) represents the second variation.



Using the equilibrium conditions (4.2), (4.3) we have

$$(4.5) \qquad A = \iiint_{R} q_{i,k}^{o} \frac{\partial (x_{i}^{!} - x_{i}^{o})}{\partial x_{k}} dV$$

$$= \iiint_{R} |p^{o}| F_{i}(x_{i}^{!} - x_{i}^{o'}) dV + \iint_{S} q_{ik}^{o} \xi_{k}(x_{i}^{!} - x_{i}^{o'}) dS$$

Assume that the forces acting on each volume element in the strained state are prescribed, i.e. that

$$F_i dV' = F_i |p| dV$$

is known. Let moreover on a portion of the boundary the stresses acting on a surface element

$$l_{ik}\xi_k^idS^i = q_{ik}\xi_kdS$$

be given, whereas on the remainder of the boundary the displacements  $x_i^*$  are prescribed. We can define for any transformation  $x^* = x^*(x)$  the potential energy by

$$P = \iiint_{R} U(p) dV - \iiint_{R} F_{i} |p|(x_{i} - x_{i}) dV - \iint_{S_{1}} q_{ik} \xi_{k}(x_{i} - x_{i}) dS$$

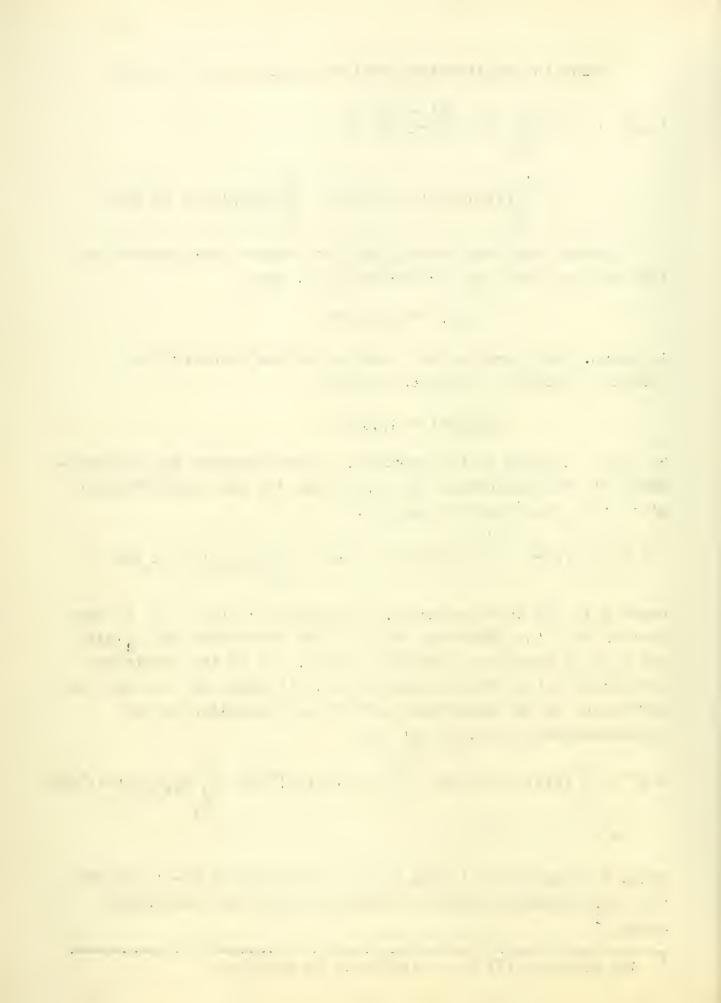
where p is the Jacobian matrix determined by x'(x).  $S_1$  is that portion of S for which the stresses are prescribed and  $F_1|p|dV$  and  $q_{ik}\xi_kdS$  have the prescribed values. If on the remaining portion of S the displacements x' and  $x'^0$  agree we have for the difference in the potential energies corresponding to the transformations x'(x) and  $x'^0(x)$ 

$$P-P^{o} = \iiint_{R} (U(p)-U(p^{o}))dx - \iiint_{R} F_{i}|p|(x_{i}^{i}-x_{i}^{i}^{o})dV - \iint_{S_{1}} q_{ik}\xi_{k}(x_{i}^{i}-x_{i}^{i}^{o})dS$$

$$= B .$$

Since B is quadratic in the first derivatives of x'-x' we have that the potential energy is stationary for the equilibrium state.

See Reissner [5] for a different formulation.



The second variation B is of the form

$$(4.6) B = \iiint_{B} Q dV$$

with

(4.7) 
$$Q = \frac{1}{2} \left( \frac{\partial^2 U(p)}{\partial p_{ik} \partial p_{rs}} \right)_{p=\overline{p}} (p_{ik} - p_{ik}^{\circ}) (p_{rs} - p_{rs}^{\circ})$$

$$\overline{p} = p^{\circ} + \Theta(p - p^{\circ}) .$$

If B is non-negative for all displacements  $x'-x^{\circ}'$  of a certain type, which do no work, then the total strain energy will be a minimum in the equilibrium state for all those admitted x'. This would clearly be the case if the quadratic form Q in the  $p_{ik}^{\circ}-p_{ik}^{\circ}$  is positive definite or semi-definite. Now in the notation (3.14)

(4.9) 
$$Q = \frac{1}{2} (D_{\xi} D_{\eta} U(p))_{\xi = \underline{\eta} = p - p}^{\circ}$$

For the special value  $\overline{p} = 1$  we have from (3.15a)

(4.10) 
$$2Q = \lambda[\xi]^2 + \mu[\xi\xi + \xi\xi^*] = \lambda\xi_{ii}\xi_{kk} + \mu(\xi_{ik}\xi_{ik} + \xi_{ki}\xi_{ki})$$
  
$$= \lambda[\xi]^2 + \frac{\mu}{2} [(\xi + \xi^*)(\xi^* + \xi)] \ge 0 ,$$

where  $\xi = p-p^0$ . This form in  $\xi$  is positive semi-definite. It vanishes whenever  $\xi$  is a skew symmetric matrix. There is no reason to assume that Q will still be semi-definite when  $\overline{p}$  is just close to 1 instead of being equal to 1. In spite of the expected indefinite character of Q there are cases where it is possible to establish that at least B is non-negative. This type of situation has been studied in recent years by  $G^{Q}$  arding, Aronszajn and others for general linear elliptic systems.

We observe that the form

$$(4.11) \quad \lambda \xi_{ii} \xi_{kk} + \mu (\xi_{ik} \xi_{ik} + \xi_{ik} \xi_{ki}) - \mu (\xi_{ik} \xi_{ki} - \xi_{ii} \xi_{kk})$$

$$= (\lambda + \mu) (\xi_{ii})^{2} + \mu \xi_{ik} \xi_{ik}$$



The second variation B is of the form

$$(4.6) B = \iiint_{B} Q dV$$

with

(4.7) 
$$Q = \frac{1}{2} \left( \frac{\partial^2 U(p)}{\partial p_{ik} \partial p_{rs}} \right)_{p=\overline{p}} (p_{ik} - p_{ik}^{o}) (p_{rs} - p_{rs}^{o})$$

$$\overline{p} = p^{\circ} + e(p-p^{\circ}) .$$

If B is non-negative for all displacements  $x'-x^{\circ}'$  of a certain type, which do no work, then the total strain energy will be a minimum in the equilibrium state for all those admitted x'. This would clearly be the case if the quadratic form Q in the  $p_{ik}-p_{ik}^{\circ}$  is positive definite or semi-definite. Now in the notation (3.14)

(4.9) 
$$Q = \frac{1}{2} \left( D_{\xi} D_{\eta} U(p) \right)_{\xi = \underline{\eta} = p - p}^{\circ} .$$

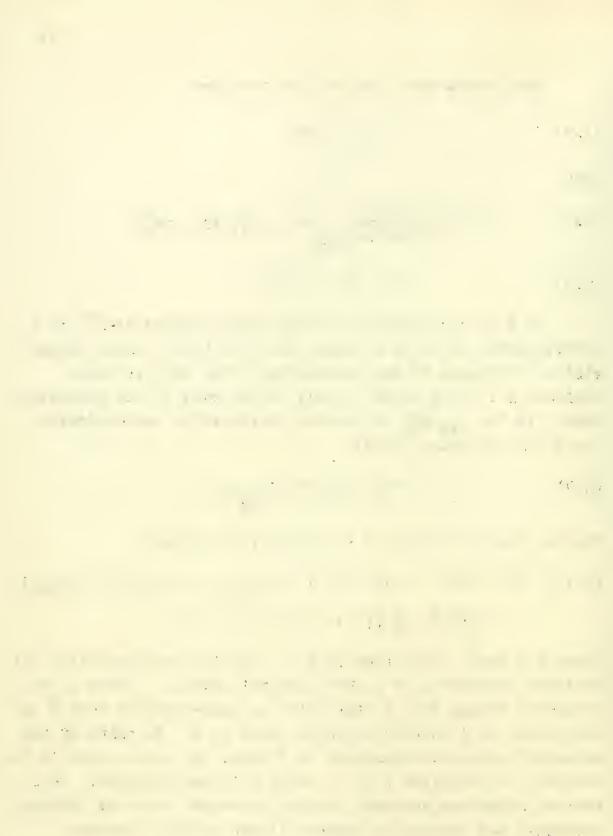
For the special value  $\overline{p} = 1$  we have from (3.15a)

(4.10) 
$$2Q = \lambda[\xi]^2 + \mu[\xi\xi + \xi\xi^*] = \lambda\xi_{ii}\xi_{kk} + \mu(\xi_{ik}\xi_{ik} + \xi_{ki}\xi_{ki})$$
  
$$= \lambda[\xi]^2 + \frac{\mu}{2} \{(\xi + \xi^*)(\xi^* + \xi)\} \ge 0 ,$$

where  $\xi = p-p^{\circ}$ . This form in  $\xi$  is positive semi-definite. It vanishes whenever  $\xi$  is a skew symmetric matrix. There is no reason to assume that  $\mathbb Q$  will still be semi-definite when  $\overline p$  is just close to 1 instead of being equal to 1. In spite of the expected indefinite character of  $\mathbb Q$  there are cases where it is possible to establish that at least B is non-negative. This type of situation has been studied in recent years by Garding, Aronszajn and others for general linear elliptic systems.

We observe that the form

$$(4.11) \quad \lambda \xi_{ii} \xi_{kk} + \mu (\xi_{ik} \xi_{ik} + \xi_{ik} \xi_{ki}) - \mu (\xi_{ik} \xi_{ki} - \xi_{ii} \xi_{kk})$$
$$= (\lambda + \mu) (\xi_{ii})^2 + \mu \xi_{ik} \xi_{ik}$$



is positive definite in  $\mathcal{E}$ , since  $\lambda$  and  $\mu$  are positive constants. The same holds then for all quadratic forms with neighboring coefficients. Assuming U=U(p) to have continuous second derivatives with respect to the  $p_{ik}$  there exists then an  $\varepsilon$  only depending on the choice of strain energy function U such that

$$\frac{1}{2} D_{\varepsilon} D_{\varepsilon} U(\overline{p}) - \frac{\mu}{2} ([\varepsilon^{2}] - [\varepsilon]^{2})$$

$$= \frac{1}{2} D_{\varepsilon} D_{\varepsilon} U(\overline{p}) - \frac{\mu}{2} (\varepsilon_{ik} \varepsilon_{ki} - \varepsilon_{ii} \varepsilon_{kk}) > 0$$
for
$$\varepsilon \neq 0, (\overline{p} - 1)(\overline{p} - 1) < \varepsilon.$$

Hence

$$(4.12) \qquad Q > \frac{\mu}{2} \left( (p_{ik} - p_{ik}^{\circ}) (p_{ki} - p_{ki}^{\circ}) - (p_{ii} - p_{ii}^{\circ}) (p_{kk} - p_{kk}^{\circ}) \right)$$

$$= \frac{\mu}{2} \frac{\partial}{\partial x_{i}} \left( (p_{ik} - p_{ik}^{\circ}) (x_{k}^{i} - x_{k}^{\circ i}) - (p_{kk} - p_{kk}^{\circ}) (x_{i}^{i} - x_{i}^{\circ i}) \right)$$

for  $p \neq p^{\circ}$  and

(4.13) 
$$[(p^*-1)(p-1)] < \varepsilon$$
,  $[(p^{\circ*}-1)(p^{\circ}-1)] < \varepsilon$ .

It follows that under the assumptions (4.13)

$$B > \frac{\mu}{2} \int_{S} \left( (p_{ik} - p_{ik}^{o}) (x_{k}^{i} - x_{k}^{o'}) - (p_{kk} - p_{kk}^{o}) (x_{i}^{i} - x_{i}^{o'}) \right) \varepsilon_{i} dS$$

unless

$$p_{ik} = p_{ik}^0$$
 in R.

If the displacements x' and  $x^{\circ}$  are equal on the boundary and if p and  $p^{\circ}$  lie in the  $\epsilon$ -neighborhood of the identity described by (4.13), we find that B>0 unless  $p=p^{\circ}$  in R. The identity of p and  $p^{\circ}$  implies that the  $x_1'-x_1^{\circ}$  are constant and hence vanish, since they vanish on the boundary S. This yields the theorem:

In the absence of external volume forces any change from the equilibrium state that leaves the boundary points fixed leads to an increase in total strain energy, as long as only states in an  $\varepsilon$ -neighborhood of the identity are considered. Here the  $\varepsilon$ -neighborhood of a transformation with Jacobian matrix  $p^O$  is defined by

$$[(p^{-}p^{0})] < \epsilon$$

Since then for given boundary displacements any state differing from a given equilibrium state (in an  $\varepsilon$ -neighborhood of the identity) corresponds to a larger total strain energy, we arrive at a uniqueness theorem:

For given displacements at the boundary there can exist at most one equilibrium state in the  $\epsilon$ -neighborhood of the identity. Here  $\epsilon$  depends exclusively on the choice of strain energy function.

It would be desirable to have a uniqueness theorem applying to transformations x' not restricted to the c-neighborhood of the identity. It is clear that the validity of such a theorem would depend on the behavior of U(p) for matrices p removed from the identity.

## Chapter $\overline{\underline{V}}$

Two Dimensional Bending of a Thin Plate

In this chapter we only consider plane deformations of the form

(5.1) 
$$x_1' = x_1'(x_1, x_2), x_2' = x_2'(x_1, x_2), x_3' = x_3.$$

Then the Jacobian matrix p is of the form

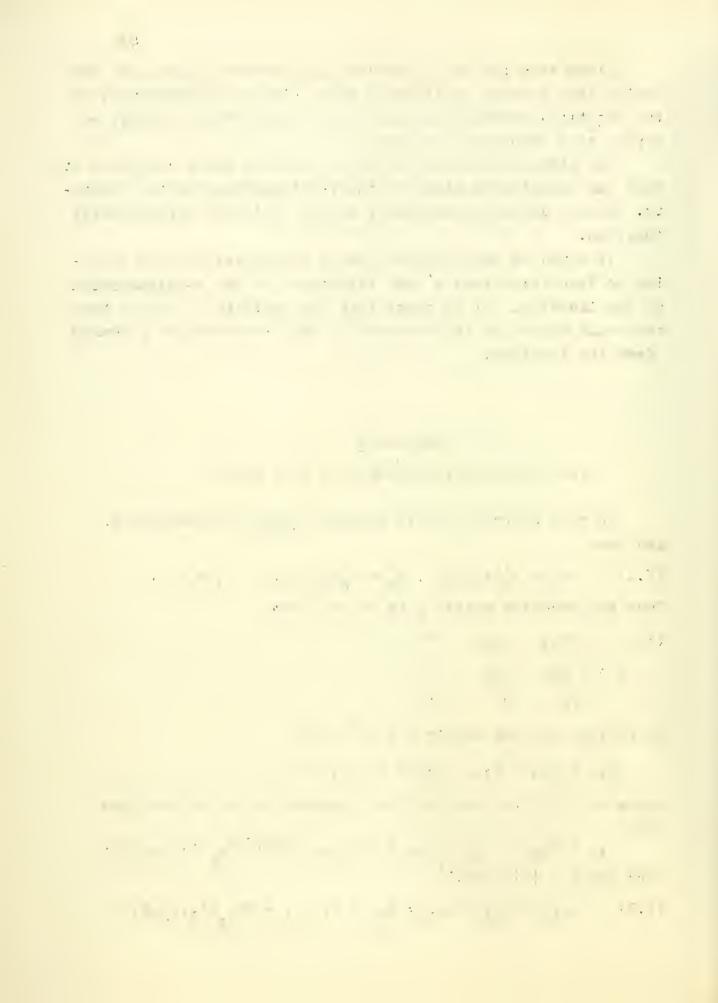
It follows for the matrix  $\gamma = pp^{n}$  that

$$\gamma_{13} = \gamma_{23} = \gamma_{31} = \gamma_{32} = 0, \gamma_{33} = 1.$$

Since by (3.8) the matrix  $\gamma$  is a function of  $\gamma$  we have then also

also 
$$\tau_{13} = \tau_{23} = \tau_{31} = \tau_{32} = 0$$
,  $\tau_{33} = 2|p|^{-1}v_{s_1}(s_1, s_2, s_3)$ .  
Thus for  $q = |p|\tau(p^*)^{-1}$ 

$$q_{13} = q_{23} = q_{31} = q_{32} = 0, \ q_{33} = 2V_{s_1}(s_1, s_2, s_3).$$



For a matrix p of the special form (5.2) the s<sub>j</sub> are determined completely by  $p_{11}$ ,  $p_{12}$ ,  $p_{21}$ ,  $p_{22}$ . Once those quantities are known we can find all of the  $\gamma_{ik}$  and  $q_{ik}$  explicitly.

Using (5.3) and the fact that all  $p_{ik}$ ,  $q_{ik}$  are functions of  $x_1$ ,  $x_2$  only, the equilibrium equations (2.18) in the absence of volume forces reduce to the two conditions

(5.4) 
$$\frac{\partial q_{i1}}{\partial x_1} + \frac{\partial q_{i2}}{\partial x_2} = 0 \quad \text{for } i = 1,2$$

In addition we have the compatibility conditions

(5.5) 
$$\frac{\partial p_{i1}}{\partial x_2} - \frac{\partial p_{i2}}{\partial x_1} = 0 \text{ for } i = 1,2.$$

The matrices p and q are connected by the relations

$$q_{ik} = \frac{\partial U(p)}{\partial P_{ik}}$$

In forming the  $q_{ik}$  with i,k = 1,2 the matrix p can already be assumed to be of the special form (5.2).

We consider now the case of a medium which in the unstrained state fills the region with rectangular cross section

$$(5.7) 0 < x_1 < a , -h < x_2 < +h .$$

The boundary requirements on the "horizontal" sides  $x_2 = \frac{t}{2}h$  shall be the absence of tractions, or by (2.21)

$$(5.8)$$
  $q_{12} = q_{22} = 0$  for  $x_2 = \frac{1}{2}h$ .

Boundary conditions on the vertical sides  $x_1 = 0$ , a would have to be imposed as well but will not be specified at present.

One trivial family of solutions is obtained by taking for  $x_1, x_2$  linear functions of  $x_1, x_2$ . The matrices p and q will then have constant elements, so that (5.4), (5.5) are satisfied automatically. The constant matrix p has only to be chosen in accordance with the boundary condition (5.8) that  $q_{12}(p) = q_{22}(p) = 0$ .

We are interested in other types of solutions corresponding to a buckled state of the medium for small thickness h.

We assume that there exists a solution of (5.4), (5.5), (5.6), (5.8) which depends analytically on  $x_2$  and h. The matrix p can then be expanded into a power series

(5.9) 
$$p = \sum_{\alpha,\beta=0}^{\infty} \frac{1}{\alpha!\beta!} p^{\alpha\beta} x_2^{\alpha} h^{\beta}$$

where

$$p^{\alpha\beta} = p^{\alpha\beta}(x_1) = \left(\frac{3^{\alpha+\beta}p(x_1,x_2,h)}{3x_2^{\alpha}3h^{\beta}}\right)_{x_2=h=0}.$$

Because p is of the form (5.2)

(5.10a) 
$$p_{i3}^{\alpha\beta} = p_{3k}^{\alpha\beta} = 0$$
 for  $\alpha + \beta > 0$ ;  $i,k = 1,2,3$ .

Similarly

(5.11) 
$$q = q(x_1, x_2, h) = \sum_{\alpha, \beta=0}^{\infty} \frac{1}{\alpha! \beta!} q^{\alpha\beta} x_2^{\alpha} h^{\beta}$$

where the matrices  $q^{\alpha\beta}$  depend on  $x_1$  alone. Here

(511a) 
$$q_{i3}^{\alpha\beta} = q_{3k}^{\alpha\beta} = 0 \text{ for } i,k = 1,2.$$

The differential equations (5.4), (5.5) become

(5.12) 
$$\frac{d q_{i1}^{\alpha\beta}}{dx_{1}} + q_{i2}^{\alpha+1}{}^{\beta} = 0, \frac{d p_{i2}^{\alpha\beta}}{dx_{1}} - p_{i1}^{\alpha+1}{}^{\beta} = 0 \text{ for } i = 1,2.$$

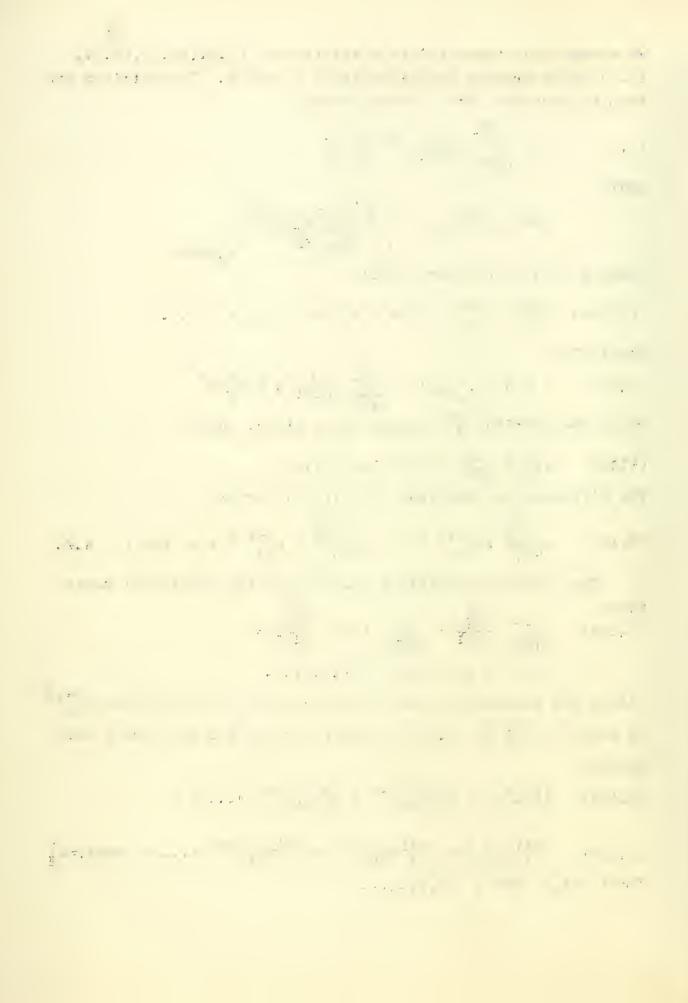
The boundary conditions (5.8) yield the additional equa-

tions
$$(5.12a) \qquad \sum_{\alpha+\beta=\gamma} \frac{q_{12}^{\alpha\beta}}{\alpha!\beta!} = \sum_{\alpha+\beta=\gamma} (-1)^{\alpha} \frac{q_{12}^{\alpha\beta}}{\alpha!\beta!} = 0$$
for  $i = 1, 2$ , and  $\gamma = 0, 1, 2, ...$ 

Adding and subtracting these two relations and expressing  $q_{12}^{\alpha+1}$  in terms of  $q_{11}^{\alpha\beta}$  by (5.12) we arrive at the two equivalent conditions

$$(5.13a) \quad {\binom{\gamma}{0}} q_{11}^{0} + {\binom{\gamma}{2}} q_{12}^{2} + {\binom{\gamma}{4}} q_{12}^{4} + \dots = 0$$

(5.13b) 
$$\binom{\gamma+1}{1}q_{11}^{0} + \binom{\gamma+1}{3}q_{11}^{2} + \binom{\gamma+1}{5}q_{11}^{4} + \dots = \text{const.} = C_{1}^{\gamma}$$
  
for  $i = 1, 2$  and  $\gamma = 0, 1, 2, \dots$ 



Equations (5.12),(5.13a,b) are supplemented by the recursion formulae obtained from (5.6) that connect the  $q^{\alpha\beta}$  with the  $p^{\alpha\beta}$ . We have first of all

(5.14a) 
$$q_{ik}^{00} = \left(\frac{\partial U}{\partial p_{ik}}\right)_{p = p^{00}}$$

(5.14b) 
$$q^{10} = \left(\frac{3q}{\partial x_2}\right)$$

$$x_2 = h = 0$$

$$= \left(\frac{3q}{\partial p_{rs}} \frac{\partial^p rs}{\partial x_2}\right)_{x_2 = h = 0} = \left(\frac{3q}{\partial p_{rs}}\right)_{p=p} = 0$$

$$= (D_p^{10} q)_{p=p} = 0$$

Similarly

(5.14c) 
$$q^{ol} = (D_{pol} q)_{p=p}^{oo}$$

(5.14d) 
$$q^{20} = (D_{plo} D_{plo} q + D_{p2o} q)_{p=p00}$$

(5.14e) 
$$q^{11} = (D_{p10} D_{p01} q + D_{p11} q)_{p=p00}$$

(5.14f) 
$$q^{\circ 2} = (D_{p} \circ 1 \quad D_{p} \circ 1 \quad q + D_{p} \circ 2 \quad q) \quad p = p^{\circ 0}$$

etc.

For the quantities of lowest order we have from (5.13,a,b) for  $\gamma = 0$ 

(5.15) 
$$q_{i2}^{00} = 0$$
,  $q_{i1}^{00} = const. = C_i^0$  for  $i = 1,2$ 

The p<sup>oo</sup> would then have to be determined from relations (5.14a). For known p<sup>oo</sup><sub>ik</sub>, q<sup>oo</sup><sub>ik</sub> the values of the q<sup>lo</sup><sub>i2</sub> and p<sup>lo</sup><sub>i1</sub> follow from (5.12) for  $\alpha=\beta=0$ . The remaining components p<sup>lo</sup><sub>i2</sub> would follow from the relations (5.14b) from the known q<sup>lo</sup><sub>i2</sub>. From (5.13a,b) for  $\gamma=1$  we have

(5.16) 
$$q_{12}^{ol} = 0$$
,  $q_{11}^{ol} = const. =  $\frac{1}{2} C_1^1$  for  $i = 1, 2$ .$ 

- - -÷ ... 1.5 The second of th Constitution of the Consti 

For known  $p^{00}$ ,  $q^{01}$  we have in (5.14c) a system of linear equations for the  $p^{01}_{ik}$ . If this system has a unique solution we can proceed in the same manner and get successively all  $p^{\alpha\beta}$ ,  $q^{\alpha\beta}$  if  $p^{00}$  has been determined and the constants  $C_i^{\gamma}$  are known.

The values of the constants  $C_1^{\gamma}$  are determined if suitable boundary conditions are prescribed on the vertical sides  $x_1=0,a$ . These constants are indeed determined uniquely by the resultant stress force applied to the vertical faces. We have from (5.13b)

For  $x_1$  = a we have by (2.16) for the components of the resultant of the stresses acting on the face  $x_1$  = a (per unit of length of  $x_3$ )

(5.18) 
$$\int_{ik} x_k' ds' = \frac{h}{h} q_{i1}(a, x_2) dx_2$$

If this resultant stress is given for each h the constants  $C_{i}^{\gamma}$  are determined.

The procedure outlined determines for known  $p^{00}$  and  $C_1^{\gamma}$  uniquely all  $p^{\alpha\beta}$ ,  $q^{\alpha\beta}$  and hence also p and q, <u>provided</u> equation (5.14c) can be solved for  $p^{01}$ , equation (5.14f) for  $p^{02}$ , etc. This is the case as long as

$$(D_{\xi} q)_{p=p^{00}} = \left(\frac{\partial q}{\partial p_{rs}}\right)_{p=p^{00}} \xi_{rs}$$

cannot vanish for  $\xi \neq 0$ . The interesting cases corresponding to buckling occur, when this condition is violated. These are the cases where the determinant of the

$$\frac{\partial q_{ik}}{\partial p_{ik}} = \frac{\partial^2 u}{\partial p_{ik} \partial p_{re}}$$

vanishes for  $p = p^{\infty}$ .

÷

. - 4

This exceptional situation always arises, when

$$(5.19)$$
  $q^{00} = 0.$ 

i.e. when the stresses vanish to lowest order in h. Condition (5.19) has been shown in Chapter 3 to be equivalent to orthogonality of the matrix  $p^{00}$ .

Let now

(5.20) 
$$p^{\circ \circ} = c$$
 where  $cc^* = 1$ .

By (3.17a), (3.15a)

(5.21) 
$$\left(\frac{\partial q_{1k}}{\partial p_{rs}}\right)_{p=c} \varepsilon_{1k} \eta_{rs} = \left[\left(D_{\xi}q\right)_{p=c} \eta^*\right] = \left(D_{\xi}D_{\eta}U(p)\right)_{p=c}$$

$$= \left(\frac{\partial q_{rs}}{\partial p_{1k}}\right)_{p=c} \varepsilon_{1k} \eta_{rs}$$

$$= U^2(c^*\xi, c^*\eta) = \lambda[c^*\xi][c^*\eta] + \mu[c^*\xi c^*\eta + \xi^*\eta].$$

The relation

$$(5.22)$$
  $(D_{\xi}q)_{p=c} = 0$ 

for a certain E is equivalent with

$$U^2(c^*\xi,c^*\eta) = 0$$
 for all %.

For any matrix a the expression

$$[a_i] = a_{ki} \eta_{ik}$$

vanishes identically in  $\eta$  if and only if a = 0. Hence (5.22) is equivalent to

$$\lambda[c^*\xi]c^* + \mu(c^*\xi c^* + \xi^*) = 0$$

or to

(5.23) 
$$\lambda[c^*\xi] + \mu(c^*\xi + \xi^*c) = 0.$$

Taking the trace of relation (5.23) we find

$$(3\lambda + 2\mu)[c^*\xi] = 0$$

and hence (for positive  $\lambda, \mu$ )

$$[c^* \varepsilon] = 0$$

and then

$$(c^*\xi) + (c^*\xi)^* = 0.$$

4 4 2

3°

Thus (5.23) is equivalent to skew symmetry of  $c^*\xi$ , i.e. to the existence of a matrix  $\Omega$  such that

(5.24) 
$$\xi = 0 \Omega.$$
  $\Omega^* = -\Omega$ 

Whenever (5.24) holds then for any

(5.24a) 
$$[(D_{\xi}q)_{p=c}\eta^*] = 0.$$
 Assume then that 
$$q^{00} = 0$$

which is consistent with (5.15). The matrix  $p^{00}$  is then orthogonal. Since  $p^{00}$  is of the form (5.2) also we must have an angle  $\theta = \theta(x_1)$  such that

angle 
$$e = e(x_1)$$
 such that.

/cos  $e$  -sin  $e$  0

(5.26)  $e$   $e$  cos  $e$  0

0 0 1

We make the further important assumption that  $\theta(x_1)$  is not constant, thus excluding trivial solutions.

Along the line  $x_2 = const. = 0$  we have  $tan \theta = \frac{dx_2}{dx_1}$  so that

 $\theta$  gives the inclination of the "middle surface" in the strained state in the limit for h = 0. It is our object to derive a differential equation for  $\theta$ .

The  $q_{ik}^{ol}$  for i,k=1,2 have constant values given by (5.16). By (5.14c),(5.21)

(5.27) 
$$q_{ik}^{ol} \xi_{ik} = \left(\frac{\partial q_{ik}}{\partial p_{rs}}\right) \xi_{ik} p_{rs}^{ol} = U^2(c^*p^{ol}, c^*\xi) = 0$$

when  $\xi$  is of the form (5.24). Take for  $\Omega$  the matrix

$$(5.27a) \Omega = \begin{pmatrix} 0 & -1 & 0 \\ 1 & 0 & 0 \\ 0 & 0 & 0 \end{pmatrix}$$
so that

so that
$$(5.28) E = c \Omega = \begin{pmatrix} -\sin \theta & -\cos \theta & 0 \\ \cos \theta & -\sin \theta & 0 \\ 0 & 0 & 0 \end{pmatrix} = \frac{dc}{d\theta}$$

Then by (5.27), (5.16)

$$0 = q_{ik}^{ol} \xi_{ik} = \frac{1}{2} C_i^l \xi_{il} = \frac{1}{2} (-C_l^l \sin \theta + C_2^l \cos \theta)$$

for all x1. Since 0 is not constant it follows that

$$c_1^1 = c_2^1 = 0$$

and hence (see (5.11a))

(5.30) 
$$q_{ik}^{ol} = 0$$
 except possibly for  $i=k=3$ .

There is no need to determine pol completely. We have from (5.30), (5.10a), (5.14c), (5.21)

$$0 = q_{ik}^{ol} p_{ik}^{ol} = (D_{pol}^{ol} p_{pol}^{ol} U(p))_{p=c} = U^{2}(c^{pol}, c^{pol}).$$

By (4.10) the form in  $\xi$ 

$$U^{2}(\xi,\xi) = \lambda[\xi]^{2} + \frac{\mu}{2}[(\xi + \xi^{*})(\xi^{*} + \xi)]$$

is semi-definite and vanishes only for skew symmetric E. Hence

$$(5.31)$$
  $(c^*p^{ol}) + (c^*p^{ol})^* = 0.$ 

For q<sup>02</sup> we have from (5.14f) and any E

$$[q^{\circ 2}\xi^*] = [(D_{p \circ 1}D_{p \circ 1}q + D_{p \circ 2}q)_{p=c}\xi^*].$$

For & given by (5.28) it follows from (5.24a) that

$$[(D_{po2}q)_{p=c}\xi^*] = [(D_{\xi}q)_{p=c}p^{o2^*}] = 0.$$

Consequently, using (3.17b),

(5.32) 
$$q_{ik}^{o2} \xi_{ik} = [q^{o2}\xi^*] = [(D_{pol}^{D}D_{pol}^{Q})\xi^*]$$

$$= (D_{pol}^{D}D_{pol}^{D}\xi^{U(p)}) = U^3(c^*p^{ol},c^*p^{ol},c^*\xi)$$

$$= U^3(c^*p^{ol},c^*p^{ol},\Omega).$$

The general expression for  $U^3$  has been computed in (3.15b). Since here all three arguments  $c^+p^{ol}$ ,  $c^+p^{ol}$ ,  $\Omega$  are skew symmetric and each term in  $U^3$  contains at least one factor depending on the symmetric part of one of the arguments, we clearly have

. . . . . . . . . . . . . . . .

(5.33) 
$$q_{ik}^{o2} \xi_{ik} = U^3(c^*p^{o1}, c^*p^{o1}, \Omega) = 0.$$

By (5.13a,b) with  $\gamma=2$ 

(533a) 
$$q_{i2}^{02} = -q_{i2}^{20}$$
,  $q_{i1}^{02} = -\frac{1}{3}(q_{i1}^{20} - C_i^2)$  for i=1,2

Hence (5.33) can be given the form

We can write this relation in the form

$$(5.34) c_i^2 \xi_{i1} = 2q_{i2}^{20} \xi_{i2} + q_{ik}^{20} \xi_{ik}.$$

Here by (5.12) for  $\alpha=1,\beta=0$ 

$$q_{i2}^{20} \xi_{i2} = -\frac{d q_{i1}^{10}}{dx_1} \xi_{i2}$$

whereas by (5.14d) (in analogy to (5.32))

$$q_{ik}^{2o} \xi_{ik} = (D_{plo}D_{plo}D_{\xi}U(p))_{p=c}$$
  
=  $U^{3}(c^{*plo}, c^{*plo}, L).$ 

Hence our relation (5.34) goes over into a relation only involving the matrices plo, qlo:

(5.35) 
$$C_i^2 \xi_{i1} = -2 \frac{dq_{i1}^{10}}{dx_1} \xi_{i2} + U^3(c^*p^{10}, c^*p^{10}, \Omega).$$

In the expression (3.15b) for U<sup>3</sup> the terms containing the higher elastic constants A,B,C make no contribution here, since C is skew symmetric. Hence

$$U^{3}(c^{*}p^{10}, c^{*}p^{10}, C) = 2\lambda[c^{*}p^{10}][(\tilde{c}^{*}c^{*}p^{10}] + 2\mu[c^{*}p^{10}^{*}c)^{2}][(\tilde{c}^{*}c^{*}p^{10}] + 2\mu[c^{*}p^{10}^{*}c)^{2}][(\tilde{c}^{*}p^{10}^{*}c)^{2}] = 2\lambda[c^{*}p^{10}][(\tilde{c}^{*}p^{10}^{*}c)^{2}] + 2\mu[c^{*}p^{10}^{*}c)^{2}[(\tilde{c}^{*}p^{10}^{*}c)^{2}] + 2\mu[c^{*}p^{10}^{*}c)^{2}[(\tilde{c}^{*}p^{10}^{*}c)^{2}] = 2U^{2}(c^{*}p^{10}, \xi^{*}p^{10}) = 2(p^{10}p^{10}^{*}\xi^{c}) = 2(p^{10}p^{10}^{*}\xi^{c})$$

14.1 1 11 11 E 1. 

For the matrices  $\xi c$ ,  $\Omega$  given by (5.28),(5.26), (5.27a) we have

Hence

$$U^{3}(c^{*}p^{10}, c^{*}p^{10}, 0) = 2[q^{10}p^{10} \cap 1]$$
$$= 2(q^{10}_{1k} p^{10}_{2k} - q^{10}_{2k} p^{10}_{1k})$$

The differential equations (5.12) yield for  $\alpha=\beta=0$ 

$$q_{12}^{10} = -\frac{dq_{11}^{00}}{dx_1} = 0$$
,  $p_{11}^{10} = \frac{dp_{12}^{00}}{dx_1} = \frac{dc_{12}}{dx_1}$ 

Hence

$$U^{3}(c^{*}p^{1o}, c^{*}p^{1o}, \Omega) = 2(q_{11}^{1o} \frac{d^{c}22}{dx_{1}} - q_{21}^{1o} \frac{d^{c}12}{dx_{1}})$$

$$= 2(q_{11}^{1o} \frac{d^{c}11}{dx_{1}} + q_{21}^{o1} \frac{d^{c}21}{dx_{1}}) = 2 q_{11}^{1o} \frac{d^{c}11}{dx_{1}}.$$

On the other hand

$$-2 \frac{dq_{i1}^{10}}{dx_{1}} \xi_{i2} = 2 \frac{dq_{i1}^{10}}{dx_{1}} c_{i1}.$$

Consequently relation (5.35) takes the form

(5.36) 
$$c_i^2 \xi_{i1} = 2 \frac{d}{dx_1} (q_{i1}^{10} c_{i1})$$

It remains to evaluate the term on the right hand side of (5.36) in terms of expressions involving 0. We have for any matrix 7

$$[q^{10}]^* = U^2(c^*p^{10}, c^*)$$

$$= \lambda[c^*p^{10}][c^*\gamma] + \mu[(c^*p^{10} + p^{10}c^*)c^*\gamma].$$

Since  $\eta$  is arbitrary this implies that

$$q^{10} = \lambda [e^*p^{10}] e + \mu (p^{10} + ep^{10}).$$

Multiplying this relation from the left with c we obtain

$$c^*q^{10} = \lambda[c^*p^{10}] + \mu(c^*p^{10} + p^{10}c)$$

or component wise

$$c_{ri} q_{rk}^{lo} = \lambda (c^*p^{lo}) \delta_{ik} + \mu (c_{ri} p_{rk}^{lo} + p_{ri}^{lo} c_{rk}).$$

territoria.

and the second of the second o

11.1

For i=k=l and i=k=2 we obtain respectively

$$c_{rl} q_{rl}^{lo} = \lambda [c_{p}^{*}l^{0}] + 2\mu c_{rl} p_{rl}^{lo}$$

$$0 = \lambda [c_{p}^{*}l^{0}] + 2\mu c_{r2} p_{r2}^{lo}$$

$$= \lambda [c_{p}^{*}l^{0}] + 2\mu c_{rs} p_{rs}^{lo} - 2\mu c_{rl} p_{rl}^{lo} - 2\mu c_{r3} p_{r3}^{lo}$$

$$= (\lambda + 2\mu)[c_{p}^{*}l^{0}] - 2\mu c_{rl} p_{rl}^{lo}.$$

Eliminating [c\*plo] between the two equations we find

$$\begin{split} \mathbf{c_{rl}} & \ \mathbf{q_{rl}^{lo}} = \frac{\mu\mu(\lambda+\mu)}{\lambda+2\mu} \ \mathbf{c_{rl}} & \ \mathbf{p_{rl}^{lo}} \\ & = \frac{\mu\mu(\lambda+\mu)}{\lambda+2\mu} \ \mathbf{c_{rl}} & \frac{\mathrm{d} \ \mathbf{p_{r2}^{oo}}}{\mathrm{d} x_{l}} \\ & = \frac{\mu\mu(\lambda+\mu)}{\lambda+2\mu} \ \mathbf{c_{rl}} & \ \xi_{r2} & \frac{\mathrm{d} \theta}{\mathrm{d} x_{l}} = -\frac{\mu\mu(\mu+\lambda)}{\lambda+2\mu} & \frac{\mathrm{d} \theta}{\mathrm{d} x_{l}} \ . \end{split}$$

Substituting this expression into (5.28) we obtain for  $\theta = \theta(x_1)$  the differential equation

(5.37) 
$$\frac{8\mu(\mu+\lambda)}{\lambda+2\mu} \frac{d^2\theta}{dx_1^2} - c_1^2 \sin\theta + c_2^2 \cos\theta = 0.$$

The coefficient

(5.38) 
$$D = \frac{8\mu(\lambda + \mu)}{3(\lambda + 2\mu)} h^3$$

is the "flexural rigidity" of the plate. (See Love [1],p.464). The constants  $C_i^2$  were defined by (5.33a)

$$C_i^2 = q_{i1}^{20} + 3q_{i1}^{02}$$
 for i=1,2,

The terminal force (per unit of length of  $x_3$ ) acting on the face  $x_1 = a$  is by (5.18),(5.17)

$$\gamma_{ik} \xi_k' ds' = 2 \sum_{\gamma=0}^{\infty} \frac{1}{(\gamma+1)!} C_i^{\gamma} h^{\gamma+1}$$

$$x_1 = a$$

This is also the total force acting on any cross section  $x_1$  = const. in the strained state. Since in the case considered

· · ·

Here the  $C_{i}^{l}$  and  $C_{i}^{o}$  vanish, we have for the components of the terminal force

$$\frac{1}{3} c_1^2 h^3 + O(h^4), \quad (i = 1,2).$$

The component of this force vertical to the middle surface in the strained state (the Shear force in a cross section  $x_1 = const$ ) is then given, up to terms of higher order in h, by

$$T = \frac{1}{3}(-C_1^2 \sin \theta + C_2^2 \cos \theta) h^3 + O(h^{\frac{1}{4}}).$$

Hence (5.37) yields up to terms of higher order.

$$\frac{d^2\theta}{dx_1^2} = -\frac{T}{D}$$

Since  $p^{00}$  is orthogonal, we have also up to terms of higher order in h for  $x_2 = 0$ 

$$ds^{2} = dx_{1}^{2} + dx_{2}^{2} = (p_{11}^{2} + p_{21}^{2}) dx_{1}^{2}$$
$$= dx_{1}^{2}(1 + o(h))$$

Hence up to terms of higher order

$$D \frac{d^2\theta}{ds} + T = 0.$$

By a suitable rigid motion we can always bring about that the terminal force acting on the end  $x_1$  = a is parallel to the  $x_1$ -axis and has components (-R,0), where R is positive in case of compression. Then

$$\frac{1}{3} c_1^2 h^3 = -R, c_2^2 = 0$$

so that the differential equation for 8 takes the familiar form

$$D \frac{d^2\theta}{dx^2} + R \sin \theta = 0.$$

with the state of the state of

. \_

and the state of t

s - 1 - 1 - 1

## Chapter VI

Buckling of thin plates.

We consider a solid, which in the unstrained state is restricted to a portion of the parallel slab

$$- h < x_3 < + h.$$

The faces in the strained state corresponding to  $x_3 = \pm h$  shall be free of tractions, i.e. we assume the boundary conditions

(6.1) 
$$q_{13} = 0$$
 for  $x_3 = \frac{1}{2}$  h.

Let the solid be in equilibrium under the influence of certain lateral forces with external volume forces absent. To derive differential equation for the "middle surface" for  $h \rightarrow 0$ , we proceed exactly as in the preceding chapter. We assume that there is a family of solutions depending analytically on  $x_3$  and h, for which p and q are given by power series expansions

(6.2) 
$$p = \sum_{\alpha,\beta=0}^{\frac{7}{6}} \frac{x_3^{\alpha} h^{\beta}}{\frac{3!}{\beta!}} p^{\alpha\beta}(x_1,x_2)$$

(6.3) 
$$q = \sum_{\alpha, \beta=0}^{\infty} \frac{x_3^{\alpha} h^{\beta}}{\alpha! \beta!} q^{\alpha\beta}(x_1, x_2)$$

The differential equations of equilibrium (2.18) become

(6.4) 
$$q_{i1,1}^{\alpha\beta} + q_{i2,2}^{\alpha\beta} + q_{i3}^{\alpha+1}^{\beta} = 0$$
 for i=1,2,3

(6.5) 
$$p_{i2,1}^{\alpha\beta} = p_{i1,2}^{\alpha\beta}$$
,  $p_{i3,1}^{\alpha\beta} = p_{i1}^{\alpha+1}{}^{\beta}$ ,  $p_{i3,2}^{\alpha\beta} = p_{i2}^{\alpha+1}{}^{\beta}$ 

for i=1,2,3. In addition we have from the boundary conditions (6.1) for  $\gamma = 0,1,2,...$ 

$$\sum_{\alpha+\beta=\gamma}^{} \frac{q_{13}^{\alpha\beta}}{\alpha!\beta!} = 0 , \sum_{\alpha+\beta=\gamma}^{} (-1)^{\alpha} \frac{q_{13}^{\alpha\beta}}{\alpha!\beta!} = 0$$

in analogy to (5.12a). The result of adding these two relations can be written

- 1

.

Euro e 

•),

The property of the property o

(6.6) 
$${\binom{\gamma}{0}}q_{13}^{0}{}^{\gamma} + {\binom{\gamma}{2}}q_{13}^{2}{}^{\gamma-2} + {\binom{\gamma}{4}}q_{13}^{4}{}^{\gamma-4} + \dots 0.$$

If we subtract the two relations and use (6.4) we obtain

$$\binom{\gamma+1}{1}$$
 $(q_{i1,1}^{\circ} + q_{i2,2}^{\circ}) + \binom{\gamma+1}{3}$  $(q_{i1,1}^{2} + q_{i2,2}^{2}) + \dots = 0.$ 

To lowest order we have from (6.6), (6.5) the nine equations

(6.8) 
$$q_{i3}^{00} = 0$$
,  $q_{i1,1}^{00} + q_{i2,2}^{00} = 0$ ,  $p_{i2,1}^{00} - p_{i1,2}^{00} = 0$ 

for i=1,2,3

where the matrices q oo and p oo are connected by

(6.9) 
$$q^{\circ \circ} = q(p^{\circ \circ}).$$

We choose again among the solutions of the system (6.8), (6.9) the ones corresponding to vanishing stresses

$$(6.10)$$
  $q^{00} = 0.$ 

Then, as we know, poo must be an orthogonal matrix. We can easily find the most general  $p^{00} = p^{00}(x_1, x_2)$  which is orthogonal and also satisfies the last group of relations in (6.8): The equations

$$p_{12.1}^{00} - p_{i1.2}^{00} = 0, i=1,2,3$$

are equivalent to the existence of three functions

$$w_i(x_1,x_2) = x_i^{100}$$
 such that

$$p_{ik}^{00} = w_{i,k}$$
 for i=1,2,3 and k=1,2.

Orthogonality of poo implies that

$$p_{ji}^{00} p_{jk}^{00} = \delta_{ik}$$
 for i,k=1,2,3.

For i,k = 1,2 this yields
$$w_{j,k} p_{j3}^{00} = 0 \quad \text{for k=1,2}$$

This means that the vectors  $(w_1, w_2, w_3) = (x_1^{100}, x_2^{100}, x_3^{100})$ 

for varying x, x, describe a developable surface referred to the isometric parameters x1,x2. Since, moreover

$$w_{j,k} p_{j3}^{00} = 0$$
 for k=1,2

we see that the quantities  $p_{i,3}^{00}$  are the components of the unit vector normal to this developable surface.

We shall only consider here the case where this limiting middle surface is the plane  $x_3' = 0$ , and where in fact the whole \* \*\* ! \_((\_{1} e\_{1} + ()\_{1} + ()\_{2} + ()\_{3} + ()\_{4}

, 4

 $\epsilon$ 

The state of

f(x) = f(x) + f(x) = f(x) =

transformation is the identity to lowest order, i.e. we take

(6.11) 
$$p_{ik}^{00} = \delta_{ik}$$

This is consistent with all relations (6.8).

We proceed to the determination of pol and qol.

We have from (6.5),(6.6),(6.7)

(6.12) 
$$q_{i3}^{ol} = 0$$
,  $q_{i1.1}^{ol} + q_{i2.2}^{ol} = 0$ ,  $p_{i2.1}^{ol} - p_{i1.2}^{ol} = 0$ 

for i=1,2,3.

Moreover, by (5.14c)

$$q^{ol} = (D_{pol} q)_{p=1}$$

and hence for an arbitrary matrix &

$$q_{ik}^{ol} \, \epsilon_{ik} = [q^{ol} \, \xi^*] = (D_{pol} \, D_{\xi} \, U(p))_{p=1} = U^2(p^{ol}, \xi)$$

$$= \lambda[p^{ol}][\xi] + \mu[p^{ol}(\xi + \xi^*)].$$

Since this expression does not change if  $\xi$  is replaced by  $\xi^*$  the matrix  $q^{ol}$  must be symmetric:

(6.14) 
$$q_{ik}^{ol} = q_{ki}^{ol}$$
 for i,k = 1,2,3

We then have from (6.12)

(6.15) 
$$q_{i3}^{ol} = q_{3i}^{ol} = 0 \text{ for } i=1,2,3$$

(6.16) 
$$q_{11,1}^{\text{ol}} + q_{12,2}^{\text{ol}} = 0, q_{21,1}^{\text{ol}} + q_{22,2}^{\text{ol}} = 0, q_{12}^{\text{ol}} = q_{21}^{\text{ol}}$$

The general solution of equations (6.16) is of the form

(6.17) 
$$q_{11}^{o1} = \phi_{,22}$$
,  $q_{12}^{o1} = q_{21}^{o1} = -\phi_{,12}$ ,  $q_{22}^{o1} = \phi_{,11}$ 

with a suitable function  $\phi(x_1,x_2)$ . (Airy function).

We can show that  $\phi$  has to satisfy the bi-harmonic equation. For this purpose we write identity (6.13) component wise

(6.18) 
$$q_{ik}^{ol} = \lambda[p^{ol}] \delta_{ik} + \mu(p_{ik}^{ol} + p_{ki}^{ol}).$$
We observe that by (6.17)

(6.19)  $\triangle^2 \phi = q_{11,22}^{\text{ol}} - q_{12,12}^{\text{ol}} - q_{21,21}^{\text{ol}} + q_{22,11}^{\text{ol}}$ where  $\triangle$  denotes the Laplace operator

$$\triangle = \frac{3x_1^2}{3x_1^2} + \frac{3x_2^2}{3x_2^2}$$

.

We have, quite generally, if F and  $\zeta$  are matrices connected by

(6.20) 
$$\zeta = (D_{\xi} q(p))_{p=1}$$

or

$$\zeta_{ik} = \lambda \xi_{rr} \delta_{ik} + \mu(\xi_{ik} + \xi_{ki})$$

that

(6.21) 
$$\zeta_{ik} = \zeta_{ki}$$
 for all i, k = 1,2,3

and that

(6.22) 
$$\zeta_{11,22} - \zeta_{12,12} - \zeta_{21,21} + \zeta_{22,11} - \frac{\lambda}{3\lambda + 2\mu} \triangle (\zeta_{11} + \zeta_{22} + \zeta_{33})$$
  
=  $2\mu(\xi_{11,2} - \xi_{12,1}) \cdot 2 - 2\mu(\xi_{21,2} - \xi_{22,1})_1$ 

Applying this identity for  $\xi = p^{ol}$ ,  $\zeta = q^{ol}$ , we find from (6.17), (6.15),(6.12) that

$$(1 - \frac{\lambda}{3\lambda + 2\mu}) \triangle^2 \phi = 0$$

and hence, since  $\lambda$  and  $\mu$  are positive,

$$(6.23) \qquad \triangle^2 \phi = 0.$$

In addition, we have from (6.15), (6.18), (6.12) that

$$p_{31,2}^{o1} = p_{32,1}^{o1}$$
,  $p_{13}^{o1} + p_{31}^{o1} = 0$ ,  $p_{23}^{o1} + p_{32}^{o1} = 0$ .

Hence there exists a function  $v = v(x_1, x_2)$  such that

(6.24) 
$$p_{31}^{ol} = p_{13}^{ol} = v_{.1}, p_{32}^{ol} = -p_{23}^{ol} = v_{.2}$$

The scalar functions  $\phi$  and v essentially describe the matrices  $p^{ol}$ ,  $q^{ol}$ . We have already the bi-harmonic equation for  $\phi$ . A differential equation for v follows from consideration of the terms of next order.

We have from (5.14b)

$$q^{lo} = (D_{plo} q(p))_{p=1}$$

or

(6.25a) 
$$q_{ik}^{lo} = \lambda p^{lo} \delta_{ik} + \mu(p_{ik}^{lo} + p_{ki}^{lo})$$
 for i,k = 1,2,3.

Moreover, from (6.4),(6.5) for  $\alpha=\beta=0$ , using  $q^{00}=0$ ,  $p^{00}=1$ :

(6.25b) 
$$q_{i3}^{lo} = p_{i1}^{lo} = p_{i2}^{lo} = 0 \text{ for } i = 1,2,3.$$

It follows immediately from (6.25a,b) that all elements

.

r ...

2 -

.

7

4

;

It follows immediately from (6.25a,b) that all elements plo , qlo vanish

$$(6.26) plo = qlo = 0.$$

Then by (5.14d)

$$q^{20} = (D_{p^{20}} q)_{p=1}$$

Since again from (6.4), (6.5) for  $\alpha=1$ ,  $\beta=0$  using (6.26)

$$q_{i3}^{20} = p_{i1}^{20} = p_{i2}^{20} = 0$$
 for  $i=1,2,3$ 

it follows in the same way that

$$(6.27) q^{20} = p^{20} = 0.$$

Generally the assumptions q oo = 0, p oo = 1 imply that

(6.27a) 
$$q^{\alpha o} = p^{\alpha o} = 0$$
 for  $\alpha=1,2,3...$ 

Equations (6.6),(6.7) for  $\gamma=2$  then yield

(6.28) 
$$q_{i3}^{02} = 0$$
,  $q_{i1,1}^{02} + q_{i2,2}^{02} = 0$  for  $i=1,2,3$ .

Moreover, by (5.14f)

(6.29) 
$$q^{o2} = (D_{pol} D_{pol} q + D_{po2} q)_{p=1}$$

Put

(6.30) 
$$Q^{\circ 2} = (D_{p^{\circ 1}} D_{p^{\circ 1}} q)_{p=1}$$

so that

(6.30a) 
$$q_{ik}^{o2} = \lambda[p^{o2}] \delta_{ik} + \mu(p_{ik}^{o2} + p_{ki}^{o2}) + Q_{ik}^{o2}$$
.

By (6.28)

$$(q_{13}^{02} - q_{31}^{02})_{,1} + (q_{23}^{02} - q_{32}^{02})_{,2} = 0$$

and hence, since qo2 - Qo2 is symmetric,

$$(6.31) \qquad (\mathfrak{Q}_{13}^{02} - \mathfrak{Q}_{31}^{02})_{.1} + (\mathfrak{Q}_{23}^{02} - \mathfrak{Q}_{32}^{02})_{.2} = 0$$

By (6.30)

$$(Q_{ik}^{o2} - Q_{ki}^{o2})\xi_{ik} = [Q^{o2}(\xi^* - \xi)]$$

$$= (D_{pol}^{D} pol^{D} \xi^{U(p)} - D_{pol}^{D} pol^{D} \xi^{*U(p)})_{p=1}$$

$$= U^{3}(p^{o1}, p^{o1}, \xi^{o1}, \xi^{o1}$$

e v

The state of the state of

1

,

40 mg

Since  $\xi - \xi^*$  is skew symmetric, the terms with A,B,C, in the expression (3.15b) for  $U^3(p^{ol},p^{ol},\xi-\xi^*)$  make no contribution, and we find

$$(Q_{ik}^{o2} - Q_{ki}^{o2})\xi_{ik} = 2\lambda[p^{o1}][p^{o1}(\xi^* - \xi)] + 2\mu[p^{o1}p^{o1}(\xi^* - \xi)]$$
 identically in  $\xi$ . In particular

$$Q_{3i}^{o2} - Q_{i3}^{o2} = 2\lambda[p^{o1}](p_{3i}^{o1} - p_{i3}^{o1}) + 2\mu(p_{3i}^{o1} p_{1i}^{o1} - p_{ij}^{o1} p_{j3}^{o1})$$

For i=1,2 we obtain then, using (6.24),(6.13)

$$Q_{3i}^{02} - Q_{i3}^{02} = 4\lambda[p^{01}]v_{,i} + 2\mu(v_{,1}(p_{1i}^{01} + p_{i1}^{01}))$$

+ 
$$v_{,2}(p_{2i}^{ol} + p_{i2}^{ol}))$$
 +  $4\mu v_{,i}p_{33}^{ol}$   
=  $2(q_{1i}^{ol}v_{,1} + q_{2i}^{ol}v_{,2} + q_{33}^{ol}v_{,i})$ 

Hence by (6.15),(6.17)

$$Q_{31}^{02} - Q_{13}^{02} = 2(\phi_{,22} v_{,1} - \phi_{,12} v_{,2})$$

$$Q_{32}^{02} - Q_{23}^{02} = 2(-\phi_{,21} v_{,1} + \phi_{,11} v_{,2}).$$

Substituting these expressions into (6.31) we obtain the desired differential equation for v:

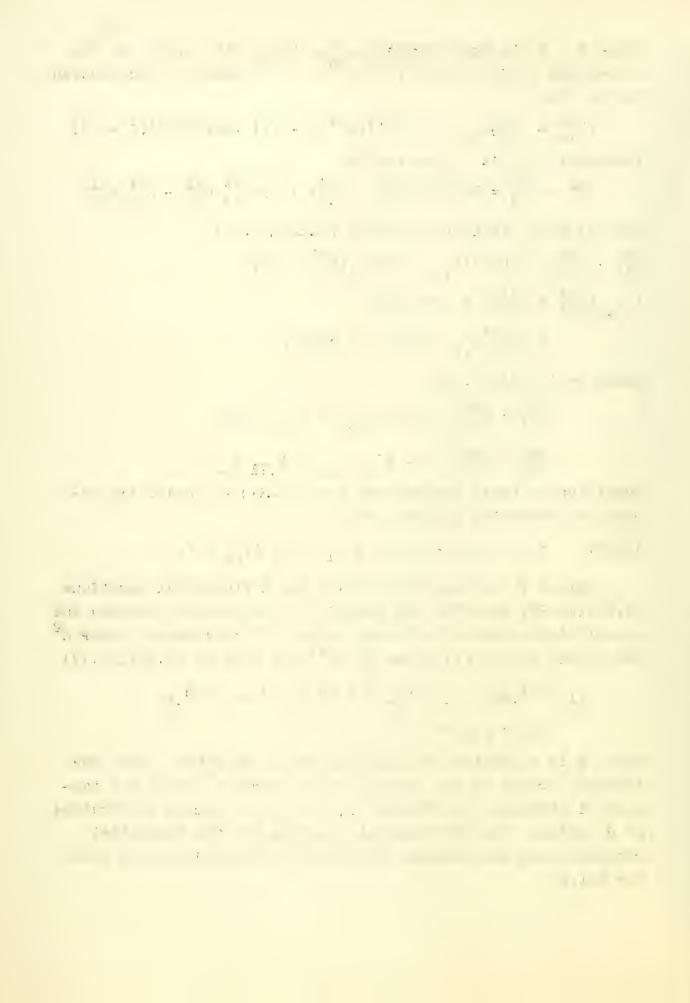
(6.32) 
$$v_{,11} \phi_{,22} - 2v_{,12} \phi_{,11} + v_{,22} \phi_{,11} = 0.$$

Unless & vanishes identically the differential equations (6.23),(6.32) describe the behavior of tangential stresses and normal displacements to lowest order. Up to terms of order he the stress matrix z is given by hq<sup>ol</sup> and thus by (6.15),(6.17)

$$\tau_{11} \approx h\phi_{,22}, \ \tau_{12} = \tau_{21} \approx -h\phi_{,12}, \ \tau_{22} \approx h\phi_{,11}$$

$$\tau_{13} = \tau_{31} \approx 0$$

where  $\phi$  is a solution of the bi-harmonic equation. Thus the stresses normal to the surface are of order  $h^2$  while the tangential stresses are of order h, unless the second derivatives of  $\phi$  vanish. The differential equation for the tangential stresses does not involve the vertical deflection, since here for i=1,2



$$\frac{\partial x_3^i}{\partial x_1} = p_{3i} = hp_{3i}^{01} + O(h^2) = hv_{,i} + O(h^2)$$

the vertical deflection is given by

(6.33) 
$$x_3' = hv(x_1, x_2) + O(h^2)$$
,

where v satisfies equation (6.32). Thus up to terms of higher order

$$\mathcal{C}_{11} \frac{\partial^2 x_3^{'}}{\partial x_1^{2}} + 2\mathcal{C}_{12} \frac{\partial^2 x_3^{'}}{\partial x_1 \partial x_2} + \mathcal{C}_{22} \frac{\partial^2 x_3^{'}}{\partial x_2^{2}} = 0.$$

We next consider the case where all stress components  $\mathbf{q}_{ik}$  are of the order  $\mathbf{h}^2$  i.e. where

$$(6.34)$$
  $q^{01} = 0.$ 

In this case the differential equation (6.32) collapses and higher order terms in the expansion have to be considered. It follows from (6.18) that

$$p_{11}^{ol} = p_{22}^{ol} = p_{33}^{ol} = p_{12}^{ol} + p_{21}^{ol} = 0.$$

Then also

$$p_{12,1}^{\text{ol}} = -p_{21,1}^{\text{ol}} = p_{11,2}^{\text{ol}} = 0,$$
 $p_{12,2}^{\text{ol}} = p_{21,2}^{\text{ol}} = -p_{22,1}^{\text{ol}} = 0.$ 

Hence

$$p_{12}^{ol} = -p_{21}^{ol} = const.$$

By a suitable rigid motion in the  $x_1x_2$ -plane one can bring about that this constant vanishes. That leaves only the components

p<sub>31</sub>, p<sub>32</sub>, p<sub>13</sub>, p<sub>23</sub>

in  $p^{ol}$  which are described by the single function  $v(x_1,x_2)$  through equations (6.24), where by (6.33)  $hv(x_1,x_2)$  is essentially the normal displacement of the plate. We notice that now  $p^{ol}$  is skew symmetric:

$$(6.35) p^{ol} = -p^{ol}^*.$$

genus to the second of the sec

. . . .

- A-

Then by (6.30), (3.15b)

$$Q_{1k}^{02} = [Q^{02}\xi^*] = (D_{pol} D_{pol} D_{\xi} U(p))$$

$$= U^3(p^{0l}, p^{0l}, \xi)$$

$$= \lambda[\xi][p^{0l}p^{0l}] + \mu[(\xi + \xi^*) p^{0l}p^{0l}]$$
(6.36)

Substituting the specific values of the pol in terms of v, we find

$$Q_{ik}^{02} = 2\lambda(v_{,1}^2 + v_{,2}^2) \delta_{ik} + 2\mu v_{,i} v_{,k}$$
 for i=k=1,2

(6.37) 
$$Q_{i3}^{02} = Q_{3i}^{02} = 0$$
 for  $i=1,2$   $Q_{33}^{02} = (2\lambda + 2\mu)(v_{,1}^2 + v_{,2}^2)$ .

Equations (6.37) show that  $0^{02}$  is symmetric. It follows from (6.30a) that also the matrix  $q^{02}$  is symmetric. We have then from (6.28)

$$q_{11,1}^{02} + q_{12,2}^{02} = 0$$
,  $q_{21,1}^{02} + q_{22,2}^{02} = 0$ ,  $q_{12}^{02} = q_{21}^{02}$ 

It follows from these equations that there exists an Airy function  $\psi = \psi(x_1, x_2)$  such that

(6.38) 
$$q_{11}^{02} = \psi_{,22} = , q_{12}^{02} = q_{21}^{02} = \psi_{,12}, q_{22}^{02} = \psi_{,11}.$$

From (6.28) and the symmetry of  $q^{02}$  we have for the remaining components of  $q^{02}$ 

(6.39) 
$$q_{13}^{02} = q_{31}^{02} = 0$$
 for i=1,2,3.

We obtain a differential equation for  $\psi$  by applying identity (6.22) to

$$\xi = p^{02}, \ \zeta = (D_{p^{02}} \ q(p))_{p=1} = q^{02} - Q^{02}.$$

The right-hand side in (6.22) vanishes because of the compatibility conditions

$$(6.40)$$
  $p_{12.1}^{02} = p_{11.2}^{02}$ 

for  $p^{02}$ ; (see (6.5)). Substituting for the  $q_{ik}^{02}$  and  $Q_{ik}^{02}$  their values from (6.38),(6.39),(6.37) we obtain

(6.41) 
$$\triangle^2 \psi = \frac{2\mu(3\lambda+2\mu)}{\lambda+\mu} (v_{,12} v_{,12} - v_{,11} v_{,22})$$

....

A STATE OF THE STA

A second equation between  $\psi$  and v is derived from the fact that the relations connecting  $q^{03}$  and  $p^{03}$  have to be compatible. We find from (6.6), (6.7) for  $\gamma=3$ 

$$q_{i1}^{\circ 3} + 3q_{i1}^{21} = 0$$
,  $q_{i1,1}^{\circ 3} + q_{i2,2}^{\circ 3} + q_{i1,1}^{21} + q_{i2,2}^{21} = 0$  for  $i=1,2,3$ .

A combination of these equations is

$$(q_{31}^{\circ 3} - q_{13}^{\circ 3})_{,1} + (q_{32}^{\circ 3} - q_{23}^{\circ 3})_{,2}$$
  
=  $(3q_{13}^{21} - q_{31}^{21})_{,1} + (3q_{23}^{21} - q_{32}^{21})_{,2}$ 

This relation will yield the desired differential equation. We find as in (5.14a,b,c,d,e,f)

(6.43a) 
$$q^{\circ 3} = (D_{p} \circ 1 \quad D_{p} \circ 1 \quad D_{p} \circ 1 \quad Q + 3D_{p} \circ 1 \quad D_{p} \circ 2 \quad Q + D_{p} \circ 3 \quad Q)_{p=1}$$

(6.43b) 
$$q^{21} = (D_{p} lo D_{p} lo D_{p} lo Q + D_{p} lo D_{p} lo Q + D_{p} lo D_{p} lo D_{p} lo Q + D_{p} lo D_{p} lo Q + D_{p} lo$$

or for any matrix E

(6.44a) 
$$[q^{03} \varepsilon^*] = U^{\downarrow}(p^{01}, p^{01}, p^{01}, \xi) + 3U^{3}(p^{01}, p^{02}, \xi) + U^{2}(p^{03}, \xi)$$

(6.44b) 
$$[q^{21}\epsilon^*] = U^{4}(p^{10}, p^{10}, p^{01}, \epsilon) + U^{3}(p^{01}, p^{20}, \epsilon) + 2U^{3}(p^{10}, p^{11}, \epsilon) + U^{2}(p^{21}, \epsilon).$$

Since only combinations  $q_{ik}^{03} - q_{ki}^{03}$  enter (6.42) we only need to determine

$$[(q^{\circ 3} - q^{\circ 3})^{\sharp}] = [q^{\circ 3}(\xi^{\sharp} - \xi)]$$

$$= U^{4}(p^{\circ 1}, p^{\circ 1}, p^{\circ 1}, \xi - \xi^{\sharp}) + 3U^{3}(p^{\circ 1}, p^{\circ 2}, \xi - \xi^{\sharp})$$

$$+ U^{2}(p^{\circ 3}, \xi - \xi^{\sharp}).$$

Using the expressions (3.15a,b,c) for  $U^2$ ,  $U^3$ ,  $U^4$  we see that only terms with coefficients  $\lambda$  or  $\mu$  make contributions, since both  $p^{ol}$  and  $\xi^{-\epsilon}$  are skew symmetric. We obtain

( · · · · | · · · | 7 o o the state of the s

- 13

1 . . .

. .

 $rac{1}{2} e^{i \omega t} = rac{1}{2} e^{i \omega t}$  ,  $rac{1}{2} e^{i \omega t} = rac{1}{2} e^{i \omega t}$  ,  $rac{1}{2} e^{i \omega t} = rac{1}{2} e^{i \omega t}$ 

\_\_\_\_\_

s. · · ·

(6.45) 
$$[(q^{o3} - q^{o3})^*] \xi^*]$$

$$= -6\lambda[p^{o1}p^{o1}][p^{o1}\xi^*] - 12\mu[p^{o1}p^{o1}p^{o1}\xi^*] + 6\lambda[p^{o2}][p^{o1}\xi^*]$$

$$+ 3\mu[(p^{o2} + p^{o2})^*] p^{o1}\xi^* + p^{o1}(p^{o2} + p^{o2})^*] \xi^*].$$

Here pol is the matrix given by

(6.45b) 
$$p^{ol} = \begin{pmatrix} 0 & 0 & -v_{,1} \\ 0 & 0 & -v_{,2} \\ v_{,1} & v_{,2} & 0 \end{pmatrix}$$

Taking for & the special matrix

$$\xi = \begin{pmatrix} 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \\ 1 & 0 & 0 \end{pmatrix}$$

relation (6.45) reduces to

(6.46) 
$$q_{31}^{\circ 3} - q_{13}^{\circ 3} = + 12(\lambda + \mu)(v_{,1}^{2} + v_{,2}^{2})v_{,1} + 6(\lambda + \mu)[p^{\circ 2}]v_{,1}$$
  
-  $6\mu p_{22}^{\circ 2} v_{,1}^{2} + 3\mu(p_{12}^{\circ 2} + p_{21}^{\circ 2})v_{,2}$ 

The  $p_{ik}^{o2}$  are connected with the known  $q_{ik}^{o2}$  by relations (6.30a) which yield

$$q^{02} = (3\lambda + 2\mu) p^{02} + Q^{02}$$

$$q^{02}_{22} = \lambda p^{02} + 2\mu p^{02}_{22} + Q^{02}_{22}$$

$$q^{02}_{12} = \mu(p^{02}_{12} + p^{02}_{21}) + Q^{02}_{12}$$

Hence, using the values of the  $q_{ik}^{o2}$ ,  $Q_{ik}^{o2}$  given by (6.37), (6.38), (6.39)

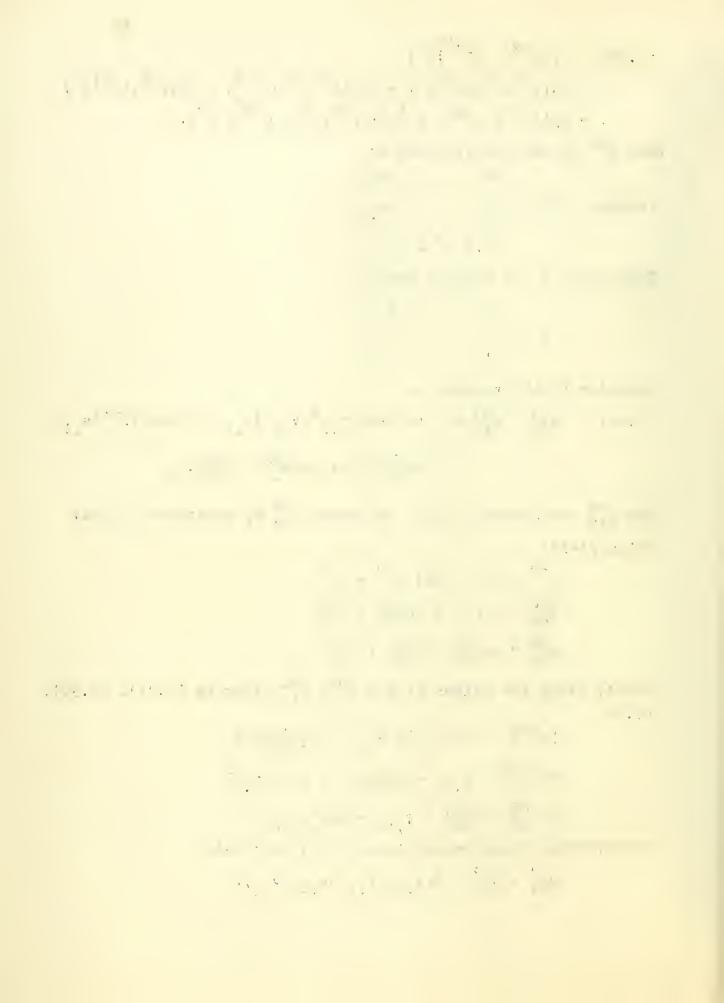
$$[p^{\circ 2}] = -2(v_{,1}^{2} + v_{,2}^{2}) + \frac{1}{3\lambda + 2\mu} \triangle \psi$$

$$2\mu p_{22}^{\circ 2} = \psi_{,11} - \frac{\lambda}{3\lambda + 2\mu} \triangle \psi - 2\mu v_{,2}^{2}$$

$$\mu(p_{12}^{\circ 2} + p_{21}^{\circ 2}) = \psi_{,12} - 2\mu v_{,1}^{2}v_{,2}^{2}$$

Substituting these values into (6.46) we obtain

$$q_{31}^{\circ 3} - q_{13}^{\circ 3} = 3(\psi_{,22} \, v_{,1} \, -\psi_{,12} \, v_{,2}).$$



Similarly, it follows that

$$q_{32}^{03} - q_{23}^{03} = 3(\psi_{,11} v_{,2} - \psi_{,12} v_{,1}).$$

This enables us finally to express the left hand side of (6.42) in terms of  $\psi$  and v:

$$(6.47) \qquad (q_{31}^{\circ 3} - q_{13}^{\circ 3})_{,1} + (q_{32}^{\circ 3} - q_{23}^{\circ 3})_{,2}$$

$$= 3(\psi_{,22} v_{,11} - 2\psi_{,12} v_{,12} + \psi_{,11} v_{,22})$$

It remains to carry out a similar reduction for the right hand side of (6.42). The matrix  $q^{21}$  is determined by (6.44b). Since by (6.25), (6.27) p<sup>10</sup> and p<sup>20</sup> vanish the terms with U<sup>4</sup> and U<sup>3</sup> in the expression (6.44b) make no contribution and we are left with

$$[q^{21}\xi^*] = U^2(p^{21},\xi)$$

or

$$q_{ik}^{2l} = \lambda[p^{2l}]\delta_{ik} + \mu(p_{ik}^{2l} + p_{ki}^{2l}).$$

Thus  $q^{21}$  is symmetric. Moreover by (6.4) for  $\alpha=\beta=1$ 

$$q_{i3}^{21} = -q_{i1,1}^{11} - q_{i2,2}^{11}$$

Hence

$$(6.48) 3q_{13}^{21} - q_{31}^{21} = 2q_{13}^{21} = -2(q_{11,1}^{11} + q_{12,2}^{11}).$$

By (5.14e) for  $p^{10} = 0$ 

(6.49) 
$$q^{11} = (D_{p11} \ q)_{p=1} = \lambda[p^{11}] + \mu(p^{11} + p^{11})$$

Here from (6.4) for  $\alpha=0$ ,  $\beta=1$ 

(6.50) 
$$q_{i3}^{11} = q_{i1,1}^{01} - q_{i2,2}^{01} = 0$$
 for  $i=1,2,3$ ,

since q<sup>ol</sup> = 0. In particular for i=3

(6.51) 
$$0 = q_{33}^{11} = \lambda[p^{11}] + 2\mu p_{33}^{11}.$$

On the other hand by (6.5), (6.45a)

(6.52) 
$$p_{ik}^{11} = p_{i3,k}^{01} = -v_{,ik}$$
 for i,k=1,2  
 $[p^{11}] = -\Delta v + p_{33}^{11}$ 

1 - 1 - 1 - 1 - 1 - 1

er l

and thus by (6.51)

(6.53) 
$$[p^{11}] = \frac{-2\mu}{\lambda + 2\mu} \Delta v$$
.

Combining (6.49) with (6.52), (6.53) we get

$$q_{ik}^{11} = -\frac{2\lambda\mu}{\lambda+2\mu} \delta_{ik} \triangle v - 2\mu v_{,ik}$$
 for i,k=1,2.

Substituting these values for the  $q_{ik}^{11}$  into (6.48) we find that

$$3q_{13}^{21} - q_{3i}^{21} = \frac{+8\mu(\lambda + \mu)}{\lambda + 2\mu} \triangle v_{i}$$
 for i=1,2

and hence that

(6.54) 
$$(3q_{13}^{21} - q_{31}^{21})_{.1} + (3q_{23}^{21} - q_{32}^{21})_{.2} = \frac{8\mu(\lambda + \mu)}{\lambda + 2\mu} \triangle^2 v.$$

Combining (6.54),(6.47),(6.42) we obtain the desired second differential equation connecting v and V:

(6.55) 
$$\triangle^2 v = \frac{3(\lambda + 2\mu)}{8\mu(\lambda + \mu)} (\psi_{,22} v_{,11} - 2\psi_{,12} v_{,12} + \psi_{,11} v_{,22})$$

Up to terms of higher order

$$w = hv(x_1, x_2)$$

represent the tangential components of the stress in the middle surface. These quantities represent at the same time the average tangential stresses throughout the thickness of the plate, since  $q^{20} = 0$  and the average of  $q^{11}x_3h$  vanishes. Equation (6.5) can be written in the form

(6.56) 
$$\triangle^2 w = \frac{2h}{N} (\zeta_{11} w_{,11} + 2\zeta_{12} w_{,12} + \zeta_{22} w_{,22})$$

where

$$N = \frac{8\mu(\lambda + \mu)}{3(\lambda + 2\mu)} h^3$$

is the plate stiffness constant: We introduce the Airy stress function

$$\chi = \frac{h^2}{2} \psi$$

so that

$$x_{11} = x_{.22}$$
,  $x_{12} = x_{21} = -x_{.12}$ ,  $x_{22} = x_{.11}$ .

•

wa b

5 - 2 - 1

the first of the second

the second secon

term to the second of the seco

 $I_{I}$ 

Then equation (6.56) takes the form

$$\triangle^2 w = \frac{2h}{N} (x,_{22} w,_{11} -2x,_{12} w,_{12} + x,_{11} w,_{22})$$

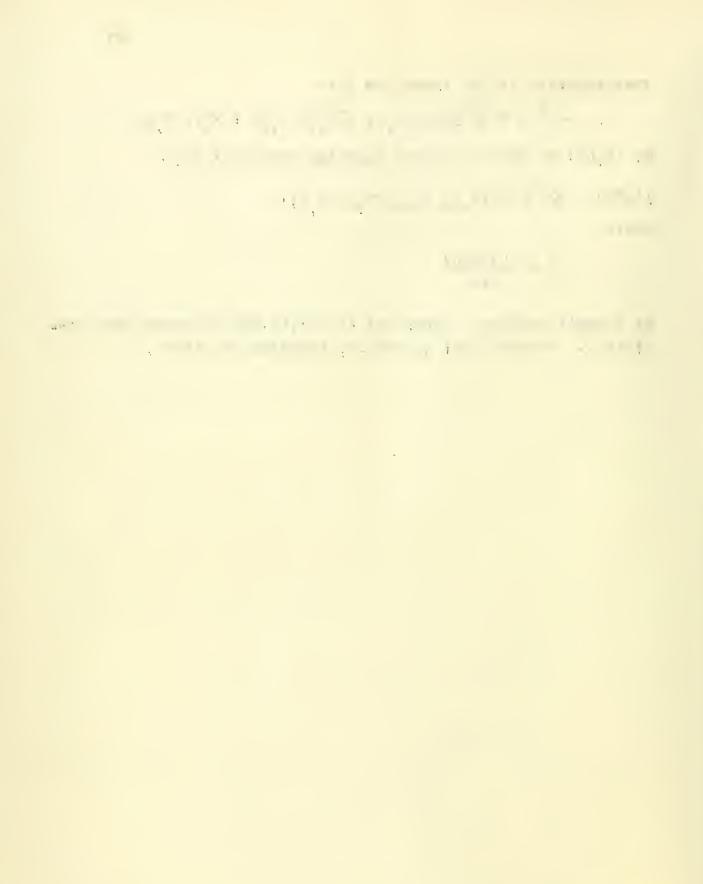
By (6.41) we have as second equation between X and w

(6.58) 
$$\triangle^2 \chi = E(w_{,12} w_{,21} - w_{,22} w_{,11})$$

where

$$E = \frac{\mu(3\lambda + 2\mu)}{\lambda + \mu}$$

is Young's modulus, Equations (6.57),(6.58) represent the familiar v. Karman-Föppl system for buckling of plates.



## References

- [1] V.V. Novozhilov: Foundations of the nonlinear theory of elasticity. (Graylock press, 1939).
- [2] F.D. Murnaghan: Finite deformation of an elastic solid.
  (Wiley Sons, 1951).
- [3] W.Z. Chien: The intrinsic theory of thin shells and plates. Quart. Appl. Math. 1(1943) pp.297-327, II, III 1bid. v.2, 1944, pp.43-59, 120-135.
- [4] J.L. Synge and W.Z. Chien: The intrinsic theory of elastic shells and plates. Applied Mathematics, Th.v. Karman anniversary volume (1941) pp.103-120.
- [5] E. Reissner: On a variational theorem for finite elastic deformations. J. Math. Physics 32(1953) pp.129-135.
- [6] R.S. Rivlin: <u>Lectures on solid mechanics</u>. Seminar in Applied Mathematics, Boulder, Colorado, 1957 (in preparation).

## DISTRIBUTION OF REPORTS CONTRACT NO. AF 49(638)-161

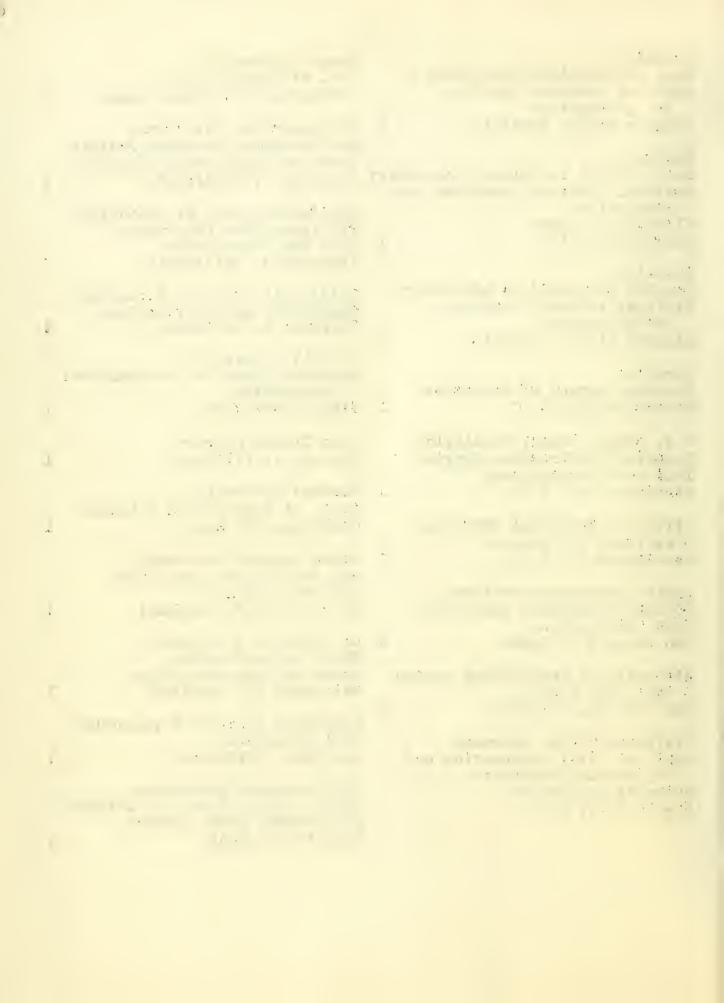
Commander Armed Services Technical Information Agency ATTN: TIPDR Arlington Hall Station		Commander Air Force Fligth Center Edwards Air Force Base California
	10	Commander Air Force Special Weapons Center
Commander Air Force Ballistic Missile Division		Kirtland Air Force Base New Mexico
Air Research + Development Command P. O. Box 262		Commander Air Research + Development Command ATTN: Director of Research
Inglewood, California	1	P. 0. Box 1395 Baltimore, Maryland 1
Commander Air Force Office of Scientific Research		Commander Hq. Air Force Missiles Development Center
Air Research + Development Command		Holloman Air Force Base New Mexico
ATTN: Mechanics Division Washington 25, D. C.	2	Commander
Director of Advanced Studies Air Force Office of Scientific Research		Air Force Missile Test Center Patrick Air Force Base Florida
Box 2035 Pasadena 2, California	1	Commander 5 Wright Development Center
Commander Air Force Armament Center ATTN: ACOP		ATTN: Aircraft Laboratory (1) Aeronautical Research Laboratory (1)
Eglin Air Force Base, Florida	1	Director of Weapons Systems Operations (1)
Commander Arnold Engineering Development		WCOSI-3 (1) Materials Laboratory (1)
Center ATTN: Library P. O. Box 162		Wright-Patterson Air Force Base, Ohio
Tullahoma, Tennessee	2	Commanding Officer Maxwell Air Force Base
Commander Air Force Cambridge Research Center		ATTN: Air University Library Maxwell Air Force Base, Alabama
L. G. Hanscom Field ATTN: CRQST-2		
Bedford, Mass.	1	
Office of Technical Services		
Department of Commerce Washington 25. D. C.	7	

110 110 110 10 1. the state of the 1... 11 71 ... . 2 ( ) ( ) . . 11, 11, .. - " = Indiana \* 1/2 1,17,10 The second secon 

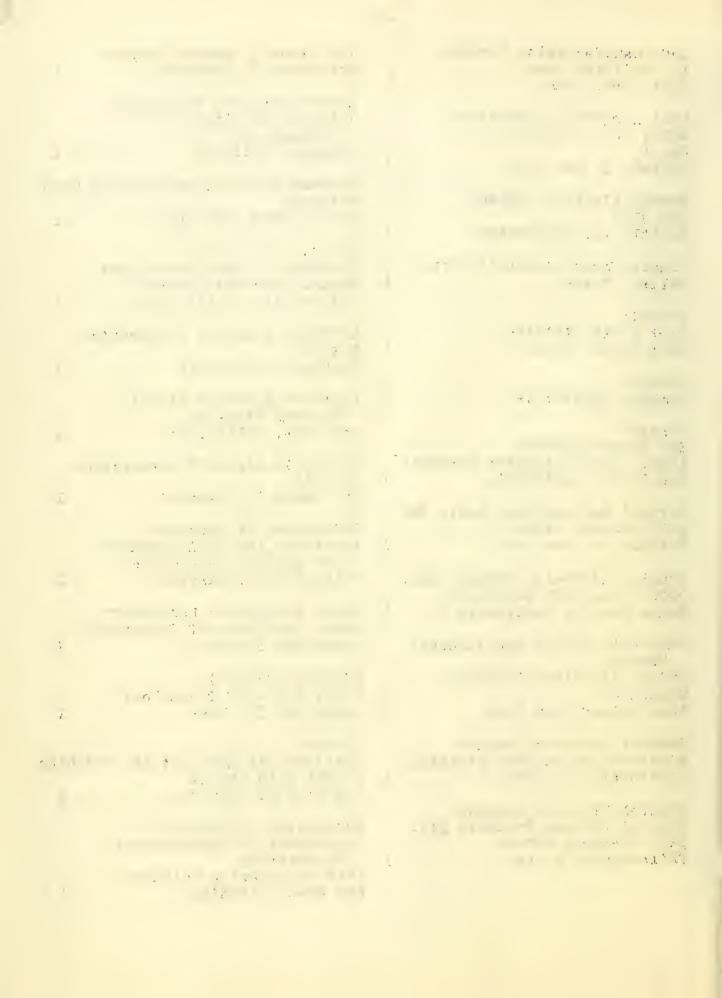
Commander Air Materiel Command Wright-Patterson Air Force Bas Ohio	e 1	Chief, Bureau of Aeronautics Department of the Navy ATTN: Research Division Washington 25, D. C. 1
Office of the Chief of Ordnanc Department of the Army ATTN: ORDTB Washington 25, D. C.	e l	Chief, Bureau of Ordnance Department of the Navy ATTN: Research + Development Div. Washington 25, D. C. 1
Director Ballistics Research Lab. Aberdeen Proving Ground Aberdeen, Maryland	1	Commander Naval Ordnance Laboratory White Oak Silver Spring, Maryland
Commanding Officer Office of Ordnance Research Duke Station Durham, North Carolina	1	Commander Naval Ordnance Test Station Inyokern China Lake, California
Commanding Officer Redstone Arsenal Huntsville, Alabama	1	Superintendent Naval Postgraduate School Monterey, California 1
Commanding Officer White Sands Proving Ground Las Cruces, New Mexico	1	Commander David Taylor Model Basin Washington 7, D. C. 1
Commander Air Force Office of Scientific Research Air Research + Development Command ATTN: Directorate of Research Information Washington 25, D. C. Chief of Naval Research Department of the Navy	2	Director Aeronautical Research National Advisory Committee for Aeronautics 1512 H Street, N.W. Washington 25, D. C. 1  Director High Speed Flight Research Station National Advisory Committee for
ATTN: Mechanics Branch (1) Air Branch (1) Washington 25, D. C.		Seronautics Edwards Air Force Base, California
Director U.S. Naval Research Laboratory ATTN: Mechanics Division Washington 25, D. C.	1	Commander European Office Air Research + Development Command 47 rue Cantersteen Shell Building
		Brussels, Belgium 1

· . . 1. 1 h . -5 (I (A) e 1 % 2 7 1 3 1 1 1 11 the second of the second of . - A 1 - 1 - 1 - 1 - 1 The state of the s

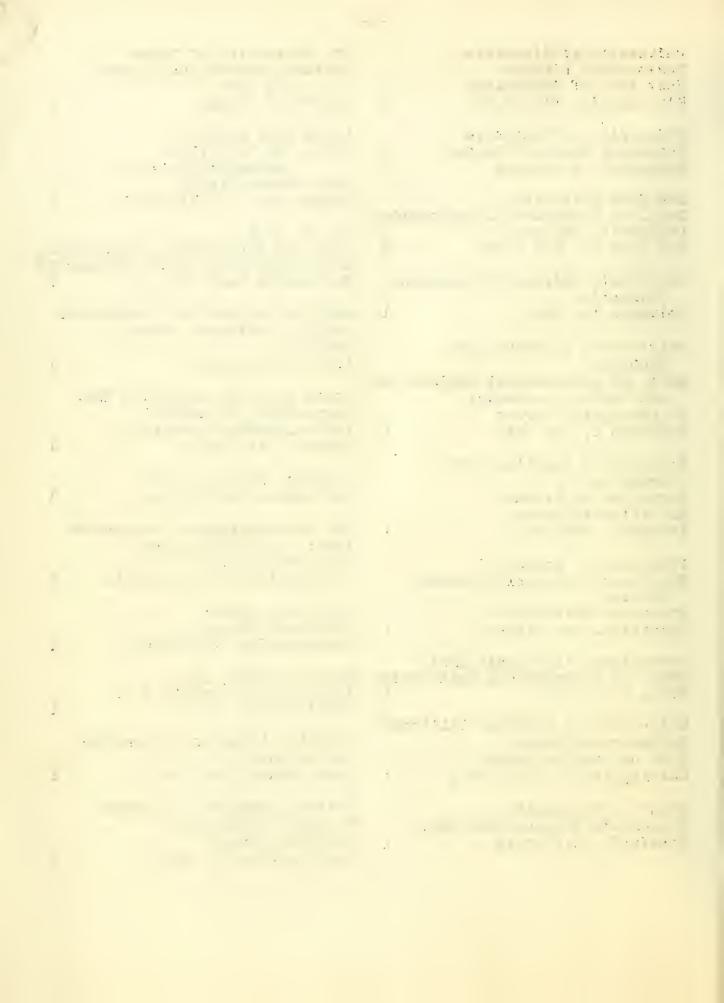
Director		Brown University	
Ames Aeronautical Laboratory		Div. of Engineering	
National Advisory Committee		Providence 12, Rhode Island	1
for Aeronautics		Trovidence 12, mode island	1
Langley Field, Virginia	1	University of Colifornia	
	_	University of California	
Director		Low Pressures Research Project	
Lewis Flight Propulsion Laborato	7777	Inst. of Engineering Research	
	ry	Berkeley 4, California	1
National Advisory Committee for			
Aeronautics		California Inst. of Technology	
Cleveland Airport	_	Jet Propulsion Laboratory	
Cleveland, Ohio	1	4800 Oak Grove Drive	
		Pasadena 3, California	1
Director		,	_
Langley Aeronautical Laboratory		California Inst. of Technology	
National Advisory Committee		Guggenheim Aeronautical Lab.	
far Aeronautics		Pasadena 4, California	1
Langley Field, Virginia	1	rabadena 4, Valifornia	
		Cornell University	
Director		Graduate School of Aeronautical	
National Bureau of Standards		Engineering	
Washington 25, D. C.	1		٦
Washington 29, D. O.	1	Ithaca, New York	1
U.S. Atomic Energy Commission		John Groven Tilhness	
		John Crerar Library	7
Technical Information Service		Chicago 1, Illinois	1
1901 Constitution Ave.	_		
Washington 25, D. C.	1	Harvard University	
		Dept. of Engineering Sciences	
Office of Technical Services		Cambridge 38, Mass.	1
Department of Commerce			
Washington 25, D. C.	1	Johns Hopkins University	
		Applied Physics Laboratory	
Applied Mechanics Reviews		8621 Georgia Ave.	
Southwest Research Institute		Silver Spring, Maryland	1
8500 Culebra Road		, 0,,	
San Antonio 6, Texas	2	Dr. Francis H. Clauser	
dan Antonia of Toxas	-	Dept. of Aeronautics	
Acronautical Engineering Paview		Johns Hopkins University	
Aeronautical Engineering Review		Baltimore 18, Maryland	1
2 East 64th Street	2	bar owner o roy har yrand	
New York 21, New York	2	Marquardt Aircraft Corporation	
D		7801 Havenhurst	
Professor George Herrmann			,
Dept. of Civil Engineering and		Van Nuys, California	1
Engineering Mechanics		AMOO Door and Told	
Columbia University		AVCO Research Laboratory	
New York 27, N. Y.		ATTN: Chief, Technical Library	
		2385 Revere Beach Parkway	
		Everett 49, Mass.	1



AVCO Manufacturing Company 155 Sniffins Lane Stratford, Conn.	1	The Glenn L. Martin Company Baltimore 3, Maryland	1
Bell Aircraft Corporation ATTN: J.S. Isenberg		Armour Research Foundation Illinois Institute of Technology	
Box 1 Buffalo 5, New York	ı	Chicago, Illinois	1
Boeing Airplane Company Box 3107		Grumman Aircraft Engineering ( Bethpage Long Island, New York	•
Seattle 14, Washington	1	Dr. A. E. Puckett	1
Chance-Vought Aircraft Corp. Dallas, Texas	1	Research + Development Labs. Hughes Aircraft Company Culver City, California	1
Convair Fort Worth Division		Lockheed Aircraft Corporation	
Fort Worth, Texas	1	Box 551 Burbank, California	1
Convair Pomona, California	1	Lookarood Arroraro Missire	
Convair San Diego Division		Systems Division Van Nuys, California	1
ATTN: Chief, Applied Research San Diego, California	1	McDonnell Aircraft Corporation Box 516	L
Cornell Aeronautical Lab., Inc	•	St. Louis 3, Missouri	1
4455 Genesee Street Buffalo 21, New York	1	University of Maryland Institute for Fluid Dynamics and Applied Mathematics	
Douglas Aircraft Company, Inc. 3000 Ocean Park Boulevard		College Park, Maryland	1
Santa Monica, California	1	Naval Supersonic Laboratory Mass. Institute of Technology	
Fairchild Engine and Aircraft Company		Cambridge 39, Mass.	1
Guided Missiles Division Wyandanch		Library (33-304) Mass. Inst. of Technology	
Long Island, New York	1	3,,	1
General Electric Company Aircraft Gas Turbine Division Cincinnati 15, Ohio	1	Library Institute of Aeronautical Scie 2 East 64th Street New York 21, New York	
General Electric Company Special Defense Products Div.		University of Michigan	1
3198 Chestnut Street Philadelphia 4, Pa.	1	Department of Aeronautical Engineering	
		East Engineering Building Ann Arbor, Michigan	1



University of Minnesota Engineering Library Institute of Technology Minneapolis, Minnesota	1	The University of Texas Defense Research Laboratory P. O. Box 8029 Austin 12, Texas	1
University of Minnesota Rosemount Research Center Rosemount, Minnesota	1	Linda Hall Library ATTN: Mr T. Gillies Document Division 5109 Cherry Street	
New York University Dept. of Aeronautical Engineer	ing	Kansas City 10, Missouri	1
University Heights New York 53, New York	1	Dr. H. G. Lew Dept. of Aeronautical Engineer The Pennsylvania State Univers	ing
Ohio State University Research Foundation		University Park, Pa.	1
Columbus 10, Ohio	1	Aerojet Engineering Corporatio 6352 N. Irwindale Avenue	n
Polytechnic Institute of Brooklyn		Box 296 Azusa, California	1
Dept. of Aeronautical Engineer and Applied Mechanics	ing	North American Aviation, Inc.	
99 Livingston Street Brooklyn 1, New York	1	Aerophysics Department 12214 Lakewood Boulevard Downey, California	1
Polytechnic Institute of Brooklyn		Northrop Aircraft, Inc.	_
Aerodynamics Library 527 Atlantic Avenue		Hawthorne, California	1
Freeport, New York	1	The Ramo-Woolridge Corporation ATTN: Chief Librarian	•
Professor S. Bogdonoff The James Forrestal Research Center		5730 Arbor Vitae Los Angeles 45, California	1
Princeton University Princeton, New Jersey	1	Rand Corporation 1700 Main Street	
Rensselaer Polytechnic Inst.	-	Santa Monica, California	1
Dept. of Aeronautical Engineer: Troy, New York	ing 1	Reed Research, Inc. 1048 Potomac Street, N.W. Washington 7, D. C.	1
University of Southern Californ Engineering Center	nia	Republic Aviation Corporation	1
3518 University Avenue Los Angeles 7, California	1	Farmingdale Long Island, New York	1
Stanford University Guggenheim Aeronautical Lab.		United Aircraft Corporation Research Department	
Stanford, California	1	362 Main Street East Hartford 8, Conn.	1



1

Westinghouse Electric Corporation Aviation Gas Turbine Division Lester Branch Post Office Philadelphia 13, Pa. 1

University of Washington
Dept. of Aeronautical Engineering
Seattle, Washington

Stanford University
Dept. of Aeronautical Engineering
Stanford, California

University of Toronto
Institute of Aerophysics
Toronto 5, Canada

Dr. Leon Trilling
Mass. Inst. of Technology
Fluid Dynamics Research Group
Cambridge 39, Mass.

. . . 5 .401

c.1 NYU IMM-250 John. On finite deformations of an elastic isotropic material. NYU c. 1 IMM-250 John AUTHOR On finite deformations of TITLE an elastic isotropic material DATE DUE 1,1

> N. Y. U. Institute of Mathematical Sciences 25 Waverly Place New York 3, N. Y.

